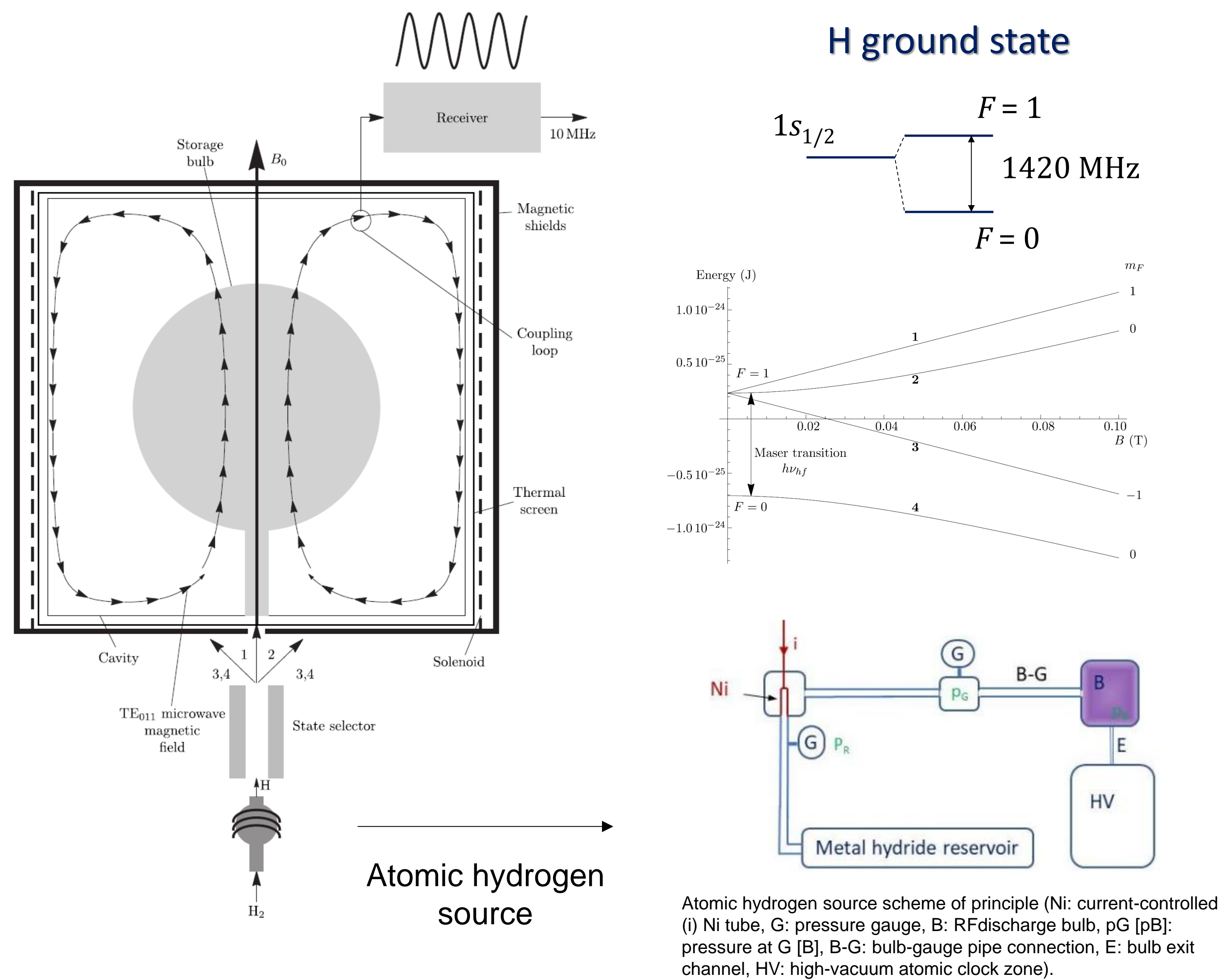


We report an experimental procedure that allows full characterisation of a metal hydride storage system and, using this procedure, demonstrate that the selected storage is suitable for 15 years of compact hydrogen maser operation dedicated to space applications.

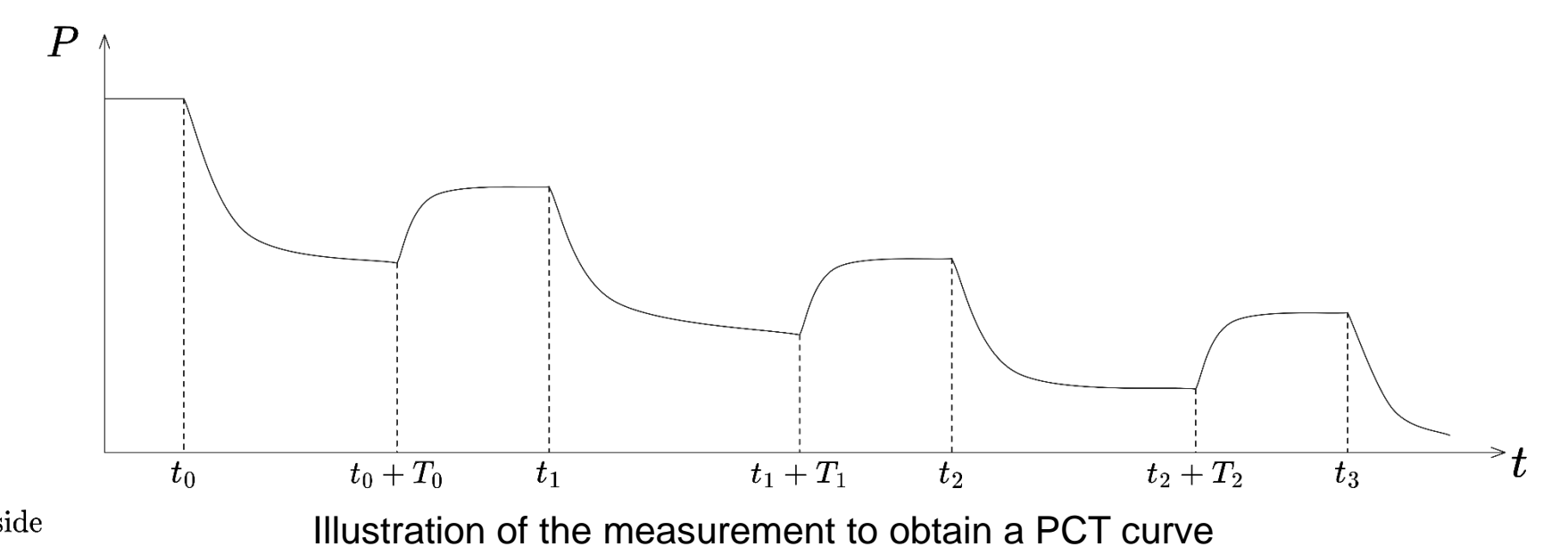
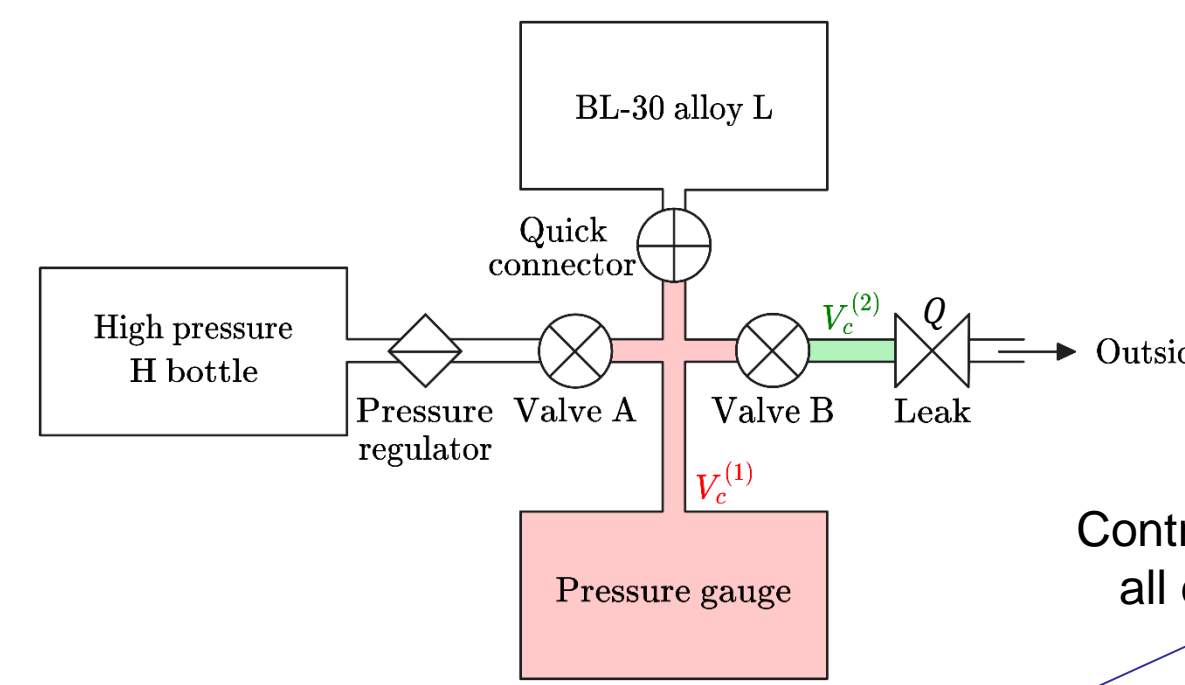
HYDROGEN MASER: SCHEME OF PRINCIPLE



EXPERIMENTAL PROCEDURE TO OBTAIN PCT CURVES

Metal hydrides exhibit low desorption rates

Discrete measurements



$$n(P(t_{i+1})) = n_0 - \sum_{k=1}^i \left(\int_{t_k}^{t_k+T_k} Q(P(t)) dt + \frac{P(t_k + T_k) V_c^{(2)}}{RT} \right) - \frac{P(t_{i+1}) V_c^{(1)}}{RT}$$

Contribution of all decays.

Amount of stored hydrogen corresponding to the equilibrium pressure $P(t_{i+1})$

Initial amount in the apparatus

Amount that has exited the apparatus during the decay

Amount present between valve B and the leak that can exit after valve B closing

Amount in the connecting pipes

LEAK RATE CALIBRATION

The manufacturer-stated accuracy of the leak rate is 20%

Need to calibrate the leak

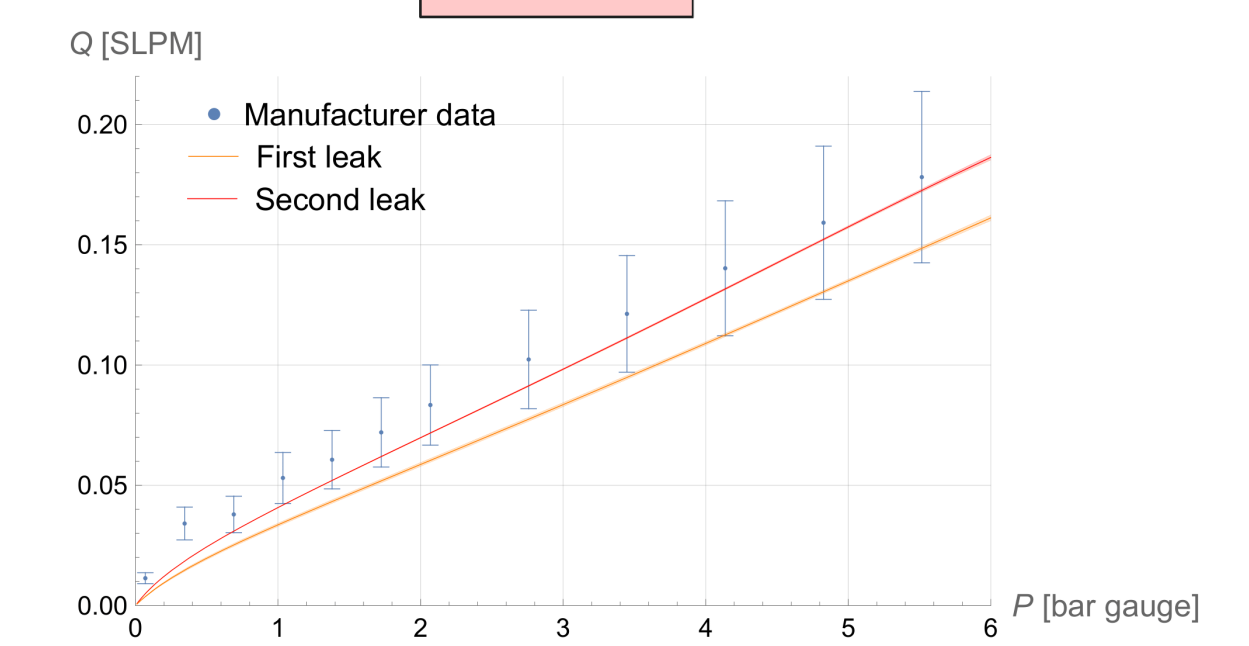
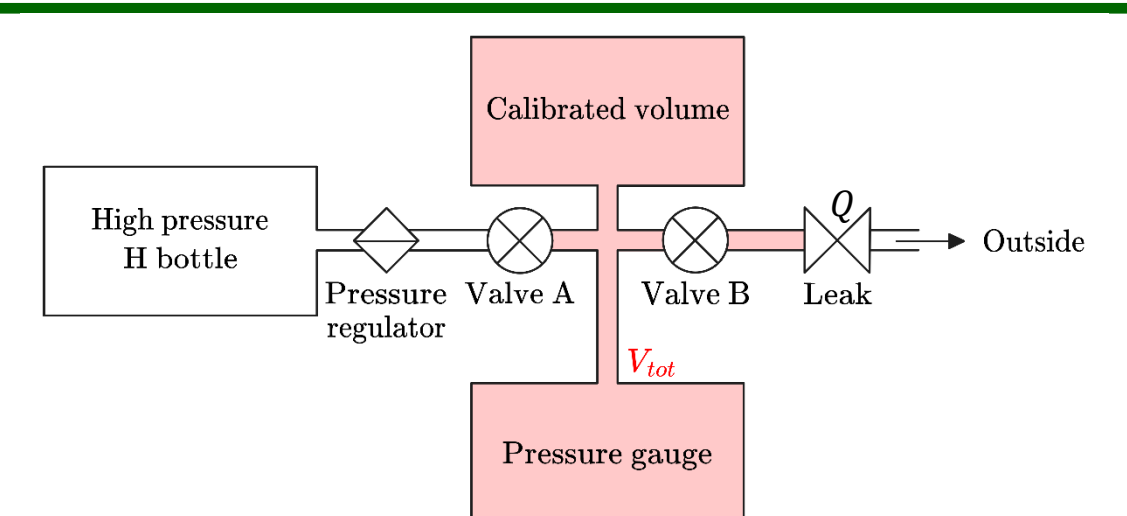
Main issue: connecting pipe volume is unknown

Use of different calibrated volumes V_1 , V_2 and V_3

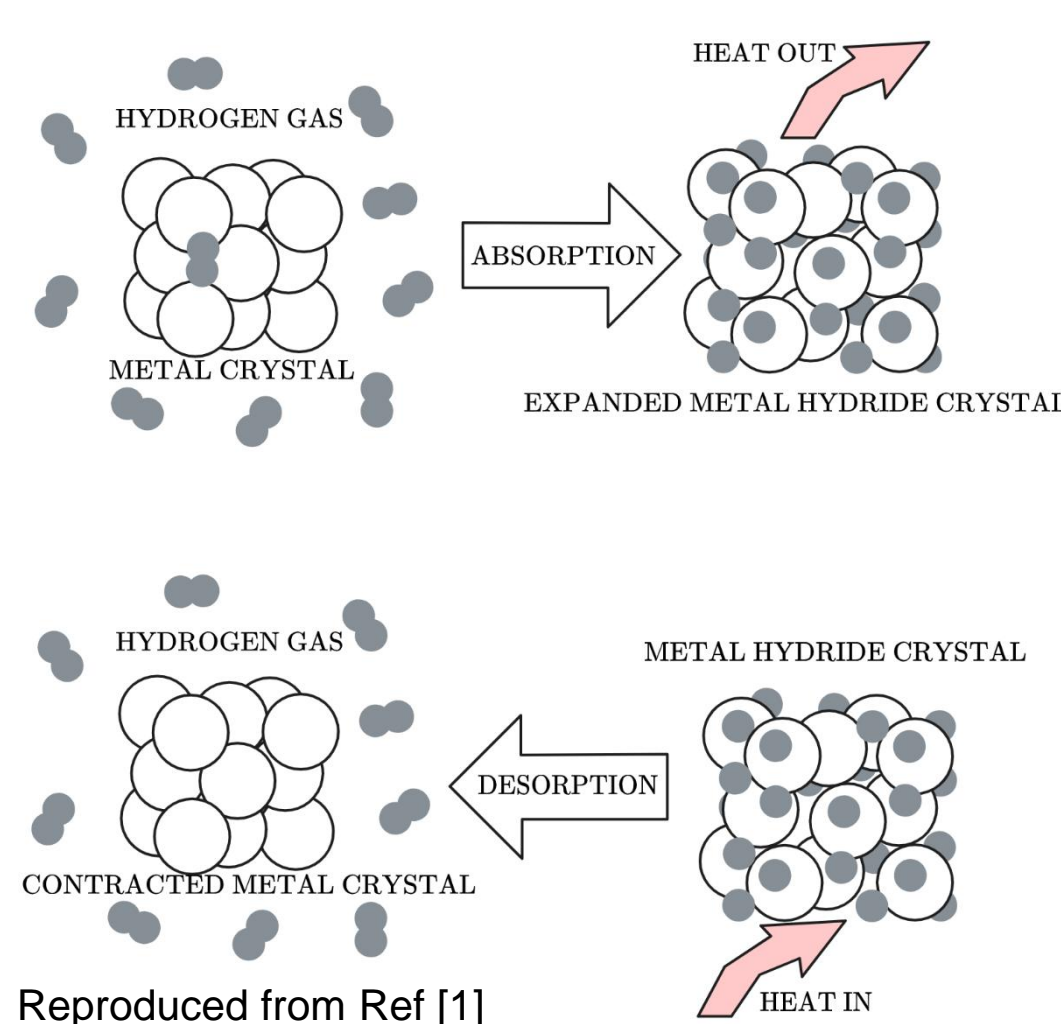
$$\begin{cases} \dot{P}_{V_1}(P) = \dot{n}(P) \frac{RT}{V_{tot,1}} \\ \dot{P}_{V_2}(P) = \dot{n}(P) \frac{RT}{V_{tot,2}} \end{cases} \Rightarrow \dot{n}(P) = \frac{1}{\frac{1}{\dot{P}_{V_1}(P)} - \frac{1}{\dot{P}_{V_2}(P)}} \frac{V_{tot,1} - V_{tot,2}}{RT} = -Q(P)$$

$V_1 - V_2$

The average of the 3 independent estimations for the pairs (V_1 , V_2), (V_2 , V_3) and (V_1 , V_3) is then taken



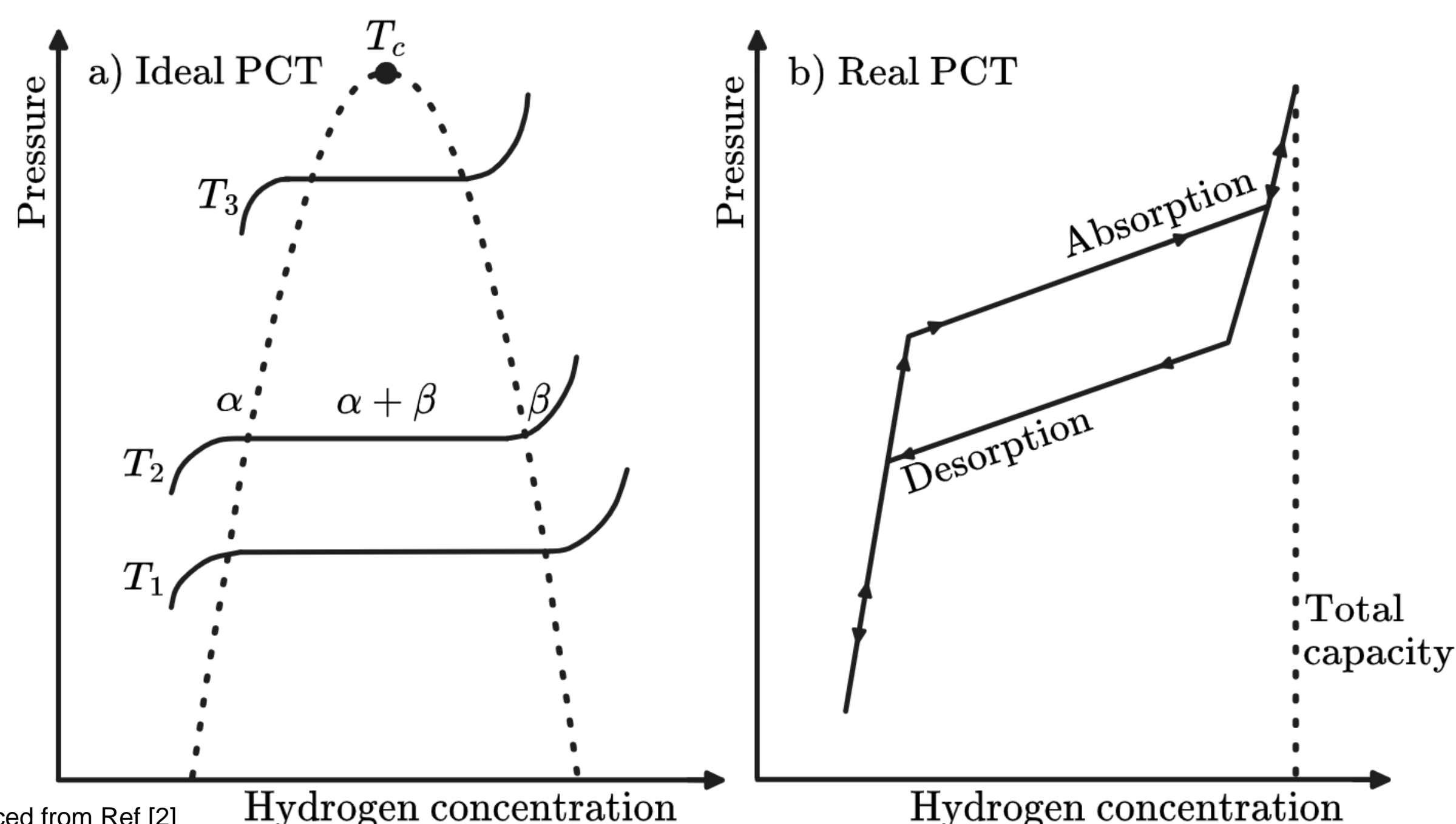
METAL HYDRIDES



Metal hydride storage → Solid state storage

Equilibrium between gas pressure and amount of hydrogen stored in the metal

Characterization: Pressure Composition Temperature (PCT) curves



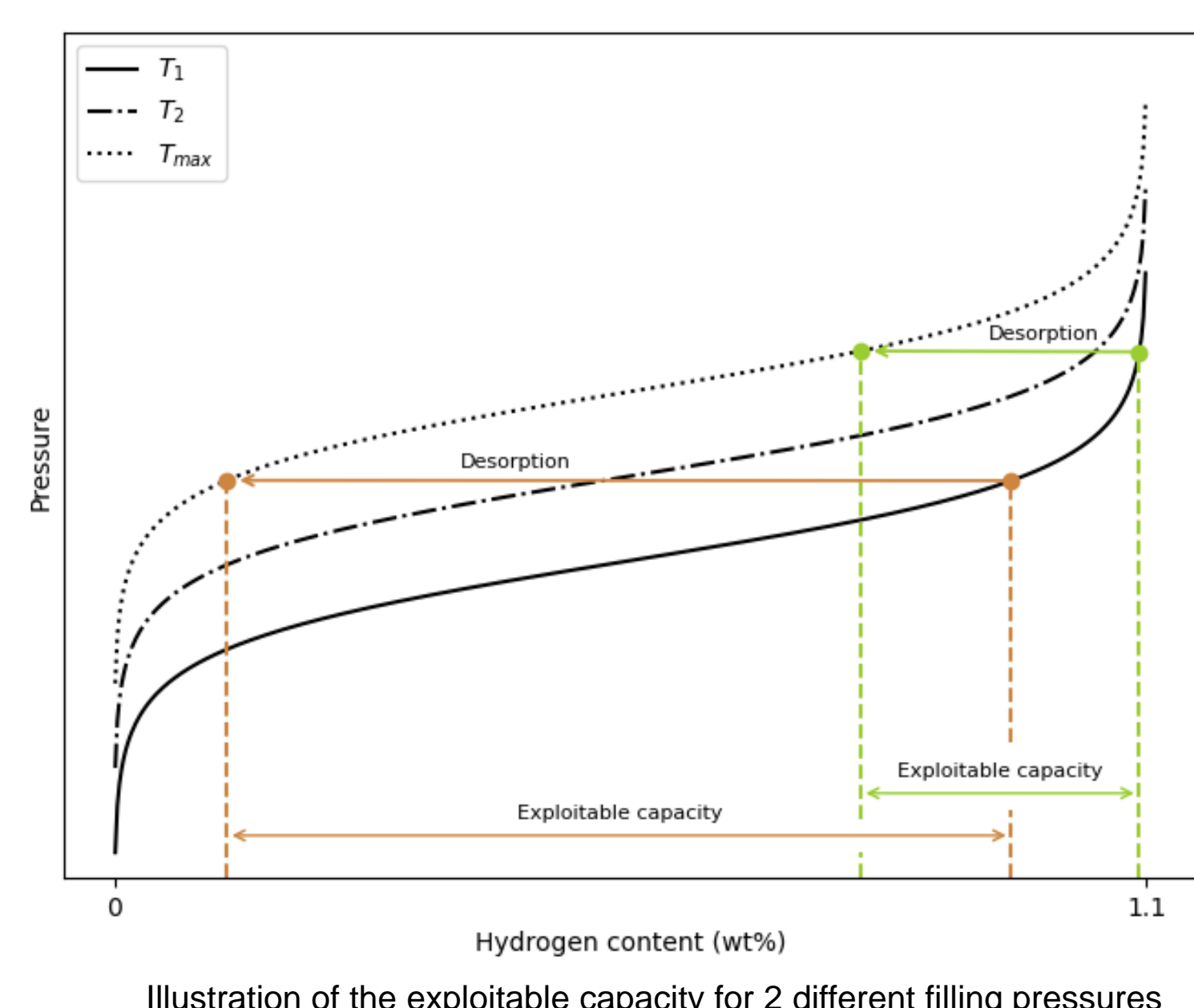
TOTAL CAPACITY VS EXPLOITABLE CAPACITY

Stability requirements:

- Need of a constant output pressure
- Heating of the container during desorption. The container cannot withstand temperatures above 50°C
- Exploitable capacity < Total capacity

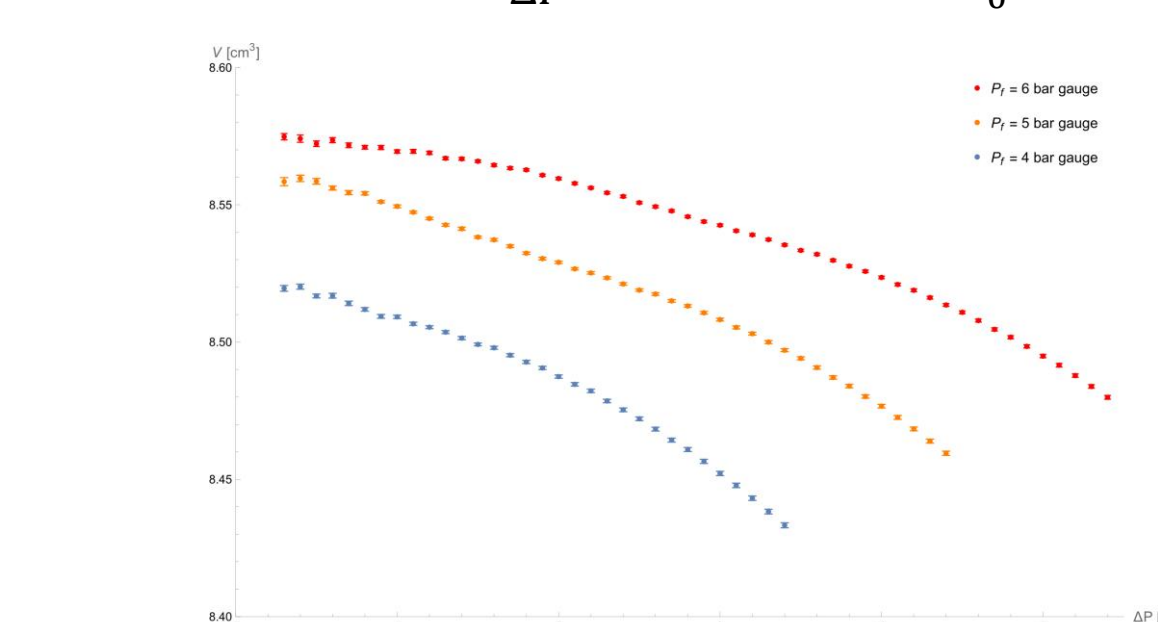
Metal hydride storage selected based on theoretical and numerical arguments [1]: BL-30 with alloy L from Fuel Cell Store

Objective: Verify that the exploitable capacity of the selected container will suffice to sustain hydrogen maser operation in a satellite (typical duration of 15 years)

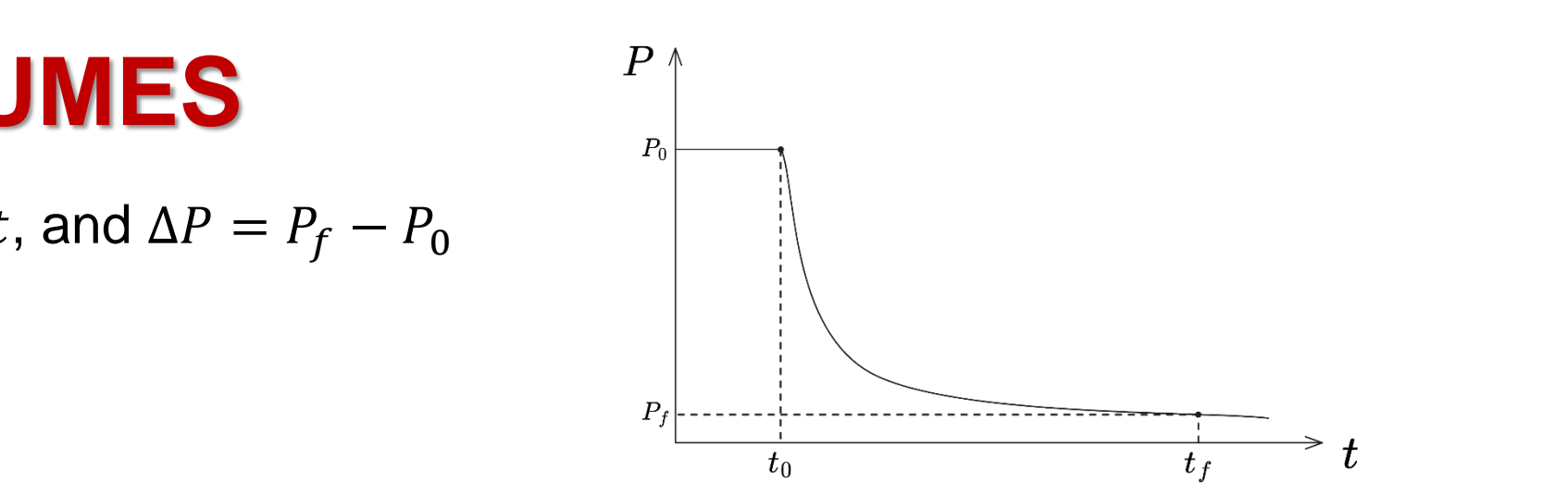


CONNECTING PIPE VOLUMES

$$V = V_c^{(1)} + V_c^{(2)} = \frac{\Delta n}{\Delta P} RT, \text{ with } \Delta n = \int_{t_0}^{t_f} Q(P(t)) dt, \text{ and } \Delta P = P_f - P_0$$



Connecting pipe volume estimation as a function of the pressure interval length on which the estimation is performed, for three different starting pressures



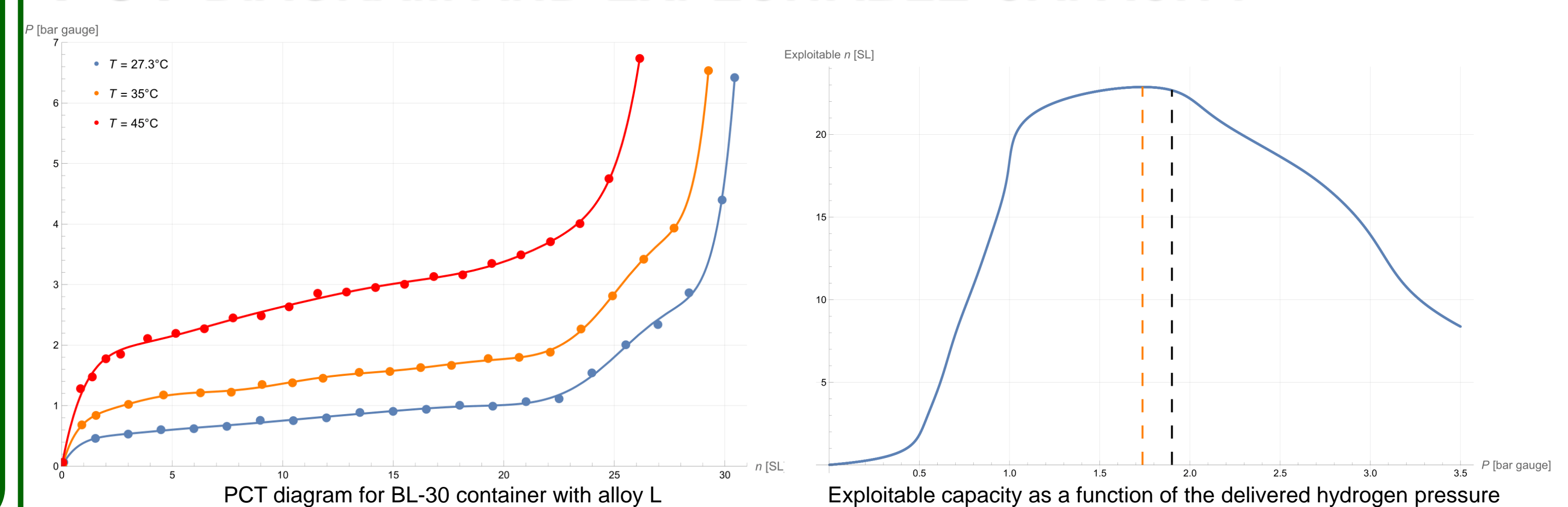
A systematic bias is observed, with an undetermined origin

$$V = V_c^{(1)} + V_c^{(2)} = 8.50(7) \text{ cm}^3$$

$V_c^{(2)}$ is geometrically determined: $V_c^{(2)} = 1.28 \text{ cm}^3$

$$\Rightarrow V_c^{(1)} = 7.22 \text{ cm}^3$$

PCT DIAGRAM AND EXPLOITABLE CAPACITY



Maximum exploitable capacity: 22.9 SL at 2.7 bar absolute

Ideal delivered pressure of 3 bar absolute

- Need a compromise between maximising the exploitable capacity and reaching a output pressure as close as possible to 3 bar absolute
- Chosen working pressure of 2.9 bar absolute corresponding to an exploitable capacity of 22.7 SL

The minimum capacity required to sustain onboard maser operation is 19.44 SL [1] → BL-30 with alloy L is suited

CONCLUSIONS

We have presented an experimental procedure that allows to completely characterize a metal hydride storage. Using this procedure, the exploitable capacity of the container as a function of the output pressure has been extracted. Regarding the compromise between maximising the exploitable capacity and obtaining a delivered pressure as close as possible to 3 bar, the working pressure has been chosen to 2.9 bar, corresponding to an exploitable capacity of 22.7 SL, thereby demonstrating that the selected container is able to provide molecular hydrogen to a compact spaceborne hydrogen maser for the entire duration of a space mission (~15 years).

→ The metal hydride storage BL-30 with alloy L from Fuel Cell Store is suited for a spaceborne hydrogen maser

REFERENCES

- [1] E. Van der Beken, "Numerical Modelling of Hydrogen Maser Physics in the Context of Space Applications", PhD thesis (University of Liège, Liège, Belgique, 2021).
- [2] N. Klopčič, I. Grimmer, F. Winkler, et al., "A Review on Metal Hydride Materials for Hydrogen Storage", Journal of Energy Storage **72**, 108456 (2023).