

# Reduced Order Modeling of Cyclically Symmetric Bladed Disks with Geometric and Contact Nonlinearities

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# Outline

## Context

Single blade

Full bladed disk

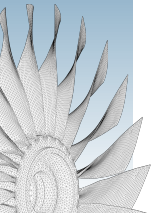
Conclusion

### 1 Context

### 2 Single blade

### 3 Full bladed disk

### 4 Conclusion





## Context

Single blade

Full bladed disk

Conclusion

## Environmental constraints



### Flightpath 2050 Europe's Vision for Aviation

Report of the High Level Group  
on Aviation Research

### By 2050...

- ▶ 75% reduction in CO<sub>2</sub>
- ▶ 90% reduction in NO<sub>x</sub>
- ▶ 65% reduction of noise



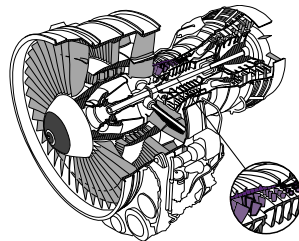
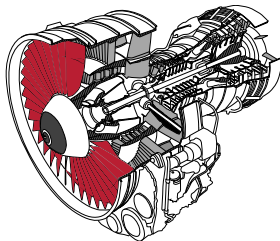
## Consequences on bladed disks design

### Context

#### Single blade

#### Full bladed disk

#### Conclusion



- ▶ Designing lighter and more flexible blades
- ▶ Geometric nonlinearities
- ▶ Reducing clearances between the rotating blades and the casing
- ▶ Contact nonlinearities

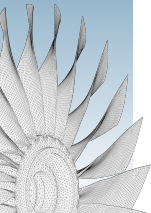
Bladed disks dynamics fundamentally nonlinear



# Numerical modeling

## Full order model

$$\mathbf{M}\ddot{\mathbf{u}} + \mathbf{C}\dot{\mathbf{u}} + \mathbf{K}\mathbf{u} + \mathbf{g}_{nl}(\mathbf{u}) = \mathbf{f}_e(t) + \mathbf{f}_c(\mathbf{u}, \dot{\mathbf{u}})$$





# Numerical modeling

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## Reduced order model

$$\tilde{\mathbf{M}}\ddot{\mathbf{q}} + \tilde{\mathbf{C}}\dot{\mathbf{q}} + \tilde{\mathbf{K}}\mathbf{q} + \tilde{\mathbf{g}}_{nl}(\mathbf{q}) = \tilde{\mathbf{f}}_e(t) + \tilde{\mathbf{f}}_c(\mathbf{q}, \dot{\mathbf{q}})$$

$$\mathbf{u} = \Phi \mathbf{q}$$

- ▶ Projection basis  $\Phi$ ?
- ▶ Reduced nonlinear internal forces  $\tilde{\mathbf{g}}_{nl}$ ?
- ▶ Treatment of contact in the reduced space  $\tilde{\mathbf{f}}_c(\mathbf{q}, \dot{\mathbf{q}})$ ?



# Objectives

## Previous work<sup>1</sup>

- ▶ Development of a methodology to study the contact interactions of a single rotating blade with geometric nonlinearities
- ▶ Validation on an industrial compressor blade model

<sup>1</sup>E. Delhez et al. *Journal of Sound and Vibration* (2021). doi: 10.1016/j.jsv.2021.116037.



# Objectives

Context

Single blade

Full bladed disk

Conclusion

## Previous work<sup>1</sup>

- ▶ Development of a methodology to study the contact interactions of a single rotating blade with geometric nonlinearities
- ▶ Validation on an industrial compressor blade model

## This presentation

- ▶ In-depth contact analyses to characterize the influence of geometric nonlinearities
- ▶ Generalization of the methodology to full bladed disks

<sup>1</sup>E. Delhez et al. *Journal of Sound and Vibration* (2021). doi: 10.1016/j.jsv.2021.116037.





# Outline

Context

**Single blade**

Methodology

Contact simulations

Full bladed disk

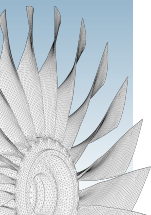
Conclusion

1 Context

2 **Single blade**

3 Full bladed disk

4 Conclusion





# Methodology

## Full order model

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$$\mathbf{u} = \Phi \mathbf{q}$$

- ▶ **Projection basis:** Craig-Bampton modes and a selection of their modal derivatives<sup>2</sup>
- ▶ **Reduced nonlinear internal forces:** evaluation with the stiffness evaluation procedure (STEP)<sup>3</sup>
- ▶ **Contact:** explicit central finite difference time integration scheme with Lagrange multipliers<sup>4</sup>

<sup>2</sup>L. Wu et al. *Proceedings of the 27th International Conference on Noise and Vibration Engineering*. Leuven (Belgium), 2016.

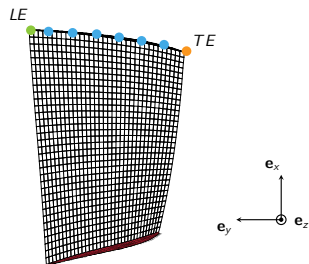
<sup>3</sup>A. Muravyov et al. *Computers & Structures* (2003). doi: 10.1016/s0045-7949(03)00145-7.

<sup>4</sup>N. J. Carpenter et al. *International Journal for Numerical Methods in Engineering* (1991). doi: 10.1002/nme.1620320107.



## Test case

- ▶ NASA rotor 37 blade (transonic compressor blade) clamped at its root<sup>5</sup>
- ▶ Open and industrial test case
- ▶ 8 boundary nodes distributed between *LE* and *TE* (contact interface)
- ▶ Reduction basis: 189 modes = 24 static modes + 15 fixed interface linear normal modes + 150 modal derivatives

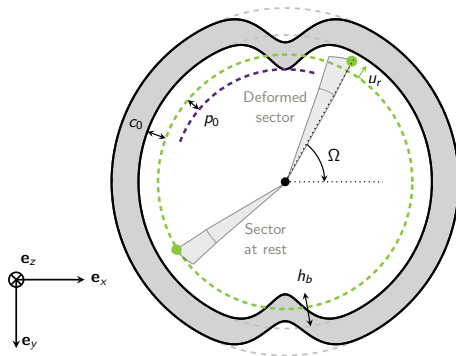


<sup>5</sup>L. Reid et al. Technical report. NASA TP 1337, 1978. doi: 10.1049/iet-gtd.2015.0403.



## Contact scenario

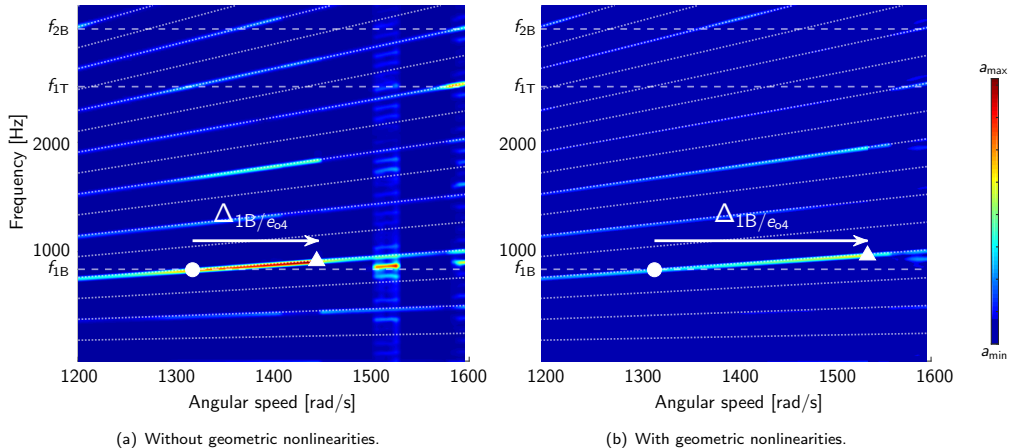
- ▶ Blade rotating at a constant speed  $\Omega$  around  $\mathbf{e}_z$
- ▶ Direct contact with rigid casing – sliding friction
- ▶ Contact initiated by deformation of the casing with two lobes
- ▶ No aerodynamic loading, no gyroscopic or centrifugal effects, no thermal effects





# Interaction maps of the radial displacement at $LE$

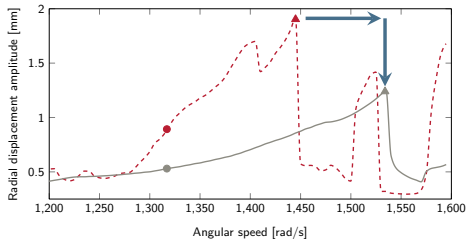
## Interaction between the first bending mode (1B) and the fourth engine order ( $e_{o4}$ )



► Interaction maps, predicted linear (●) and nonlinear (▲) resonances.



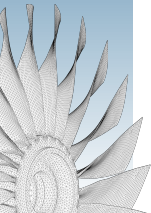
## Nonlinear frequency response curve



- NFRC without (—) and with (—) geometric nonlinearities, predicted linear (●) and nonlinear (▲) resonances.

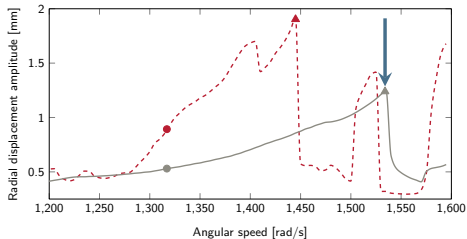
### Observations

- Contact stiffening
- Amplitude jumps
- Influence of geometric nonlinearities
  - Smoother interactions
  - Additional contact stiffening





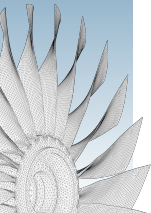
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## Observations

- Contact stiffening
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- Influence of geometric nonlinearities
  - Smoother interactions?
  - Additional contact stiffening





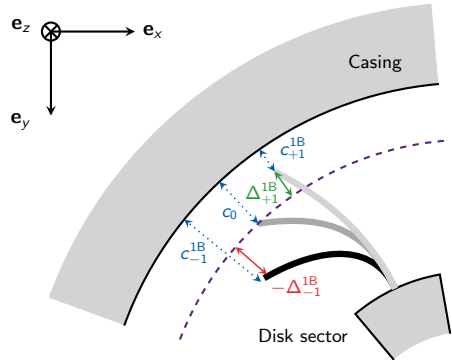
## Clearance consumption

### Definition

- Evolution of the clearance between the blade and the casing when the blade vibrates along 1 mode

$$\Delta(\delta) = c_0 - c(\delta)$$

- Possible key parameter for the design of blades subjected to contact interactions<sup>6</sup>



<sup>6</sup>A. Batailly et al. *Proceedings of the ASME Turbo Expo* (2016). doi: 10.1115/gt2016-56721.

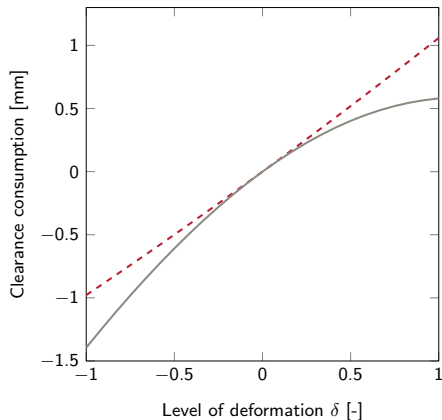




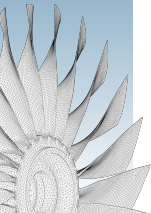
# Clearance consumption

## Observations

- ▶ Reduced clearance consumption with geometric nonlinearities
- ▶ Justify that the blade with geometric nonlinearities features lower vibration response to contact
- ▶ Linear model valid for  $\delta \in [-0.25, 0.2]$

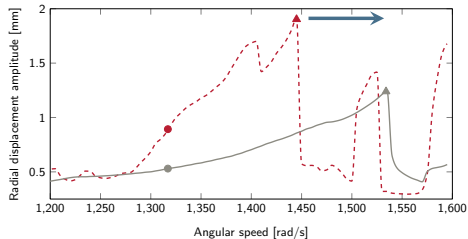


- ▶ Clearance consumption at *LE* without (—) and with (—) geometric nonlinearities.





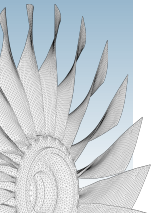
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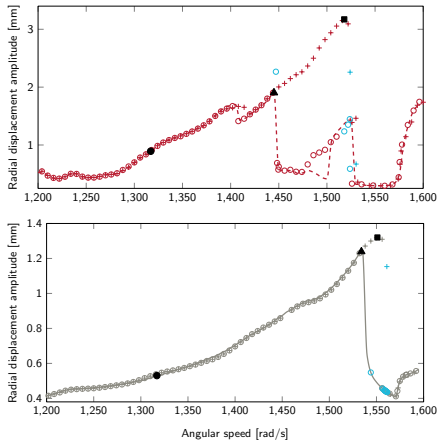
### Observations

- Contact stiffening
- Amplitude jumps
- Influence of geometric nonlinearities
  - Smoother interactions
  - **Additional contact stiffening?**





# Nonlinear frequency response curve with continuation



- NFRC without (— / - -) and with continuation (+ / O), without geometric nonlinearities (above) and with geometric nonlinearities (below).

## Numerical procedure

- NFRC built with a sequential continuation procedure
- Upward (+) and downward (O) angular speed sweeps

## Observations

- Without continuation, nonlinear resonance (■) not correctly captured
- Contact stiffening similar with and without geometric nonlinearities



# Outline

Context

Single blade

**Full bladed disk**

Methodology

Verification without  
contact

Contact simulations

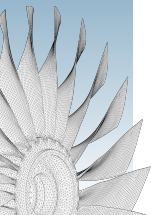
Conclusion

1 Context

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# Generalization of the methodology with CMS techniques

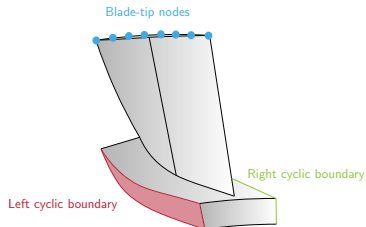
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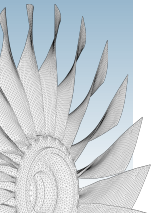
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- **Projection basis:** for each sector, Craig-Bampton modes and a selection of their modal derivatives + second reduction of the cyclic boundary





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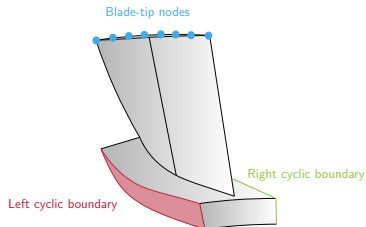
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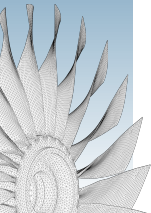
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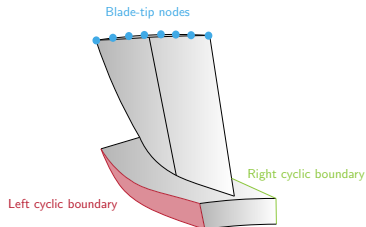
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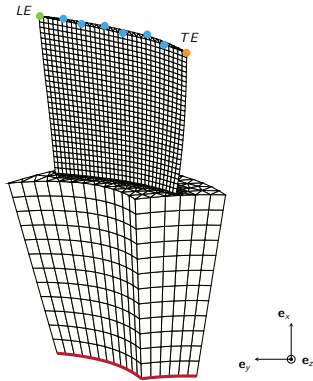
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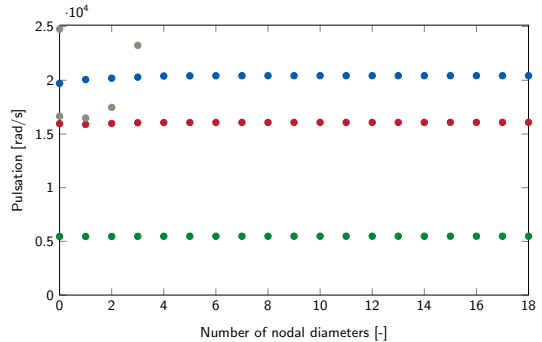
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- **Reduced nonlinear internal forces:** STEP, assumption of linear coupling between the sectors
- **Contact:** explicit central finite difference time integration scheme with Lagrange multipliers



## Test case



- ▶ NASA rotor 37 bladed disk with 36 sectors
- ▶ 133,605 degrees-of-freedom per sector
- ▶ Sectors clamped at disk lower surface



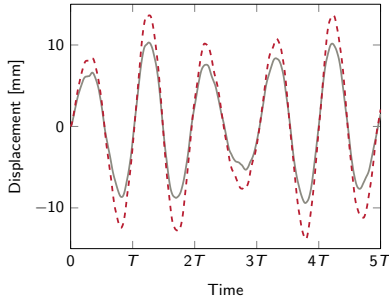
▶ SAFE diagram.



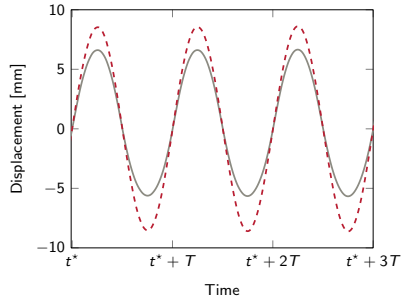


## Verification without contact

- Reduction basis: per sector, 3 static modes ( $LE$ ) + 10 fixed interface linear normal modes + 10 modal derivatives + 3 modes for the second projection (total: 972 modes)
- Blade excited by a harmonic excitation of amplitude  $A = 400$  N and pulsation  $\omega = 4,500$  rad/s in the  $\mathbf{e}_z$  direction



(a) Transient regime.



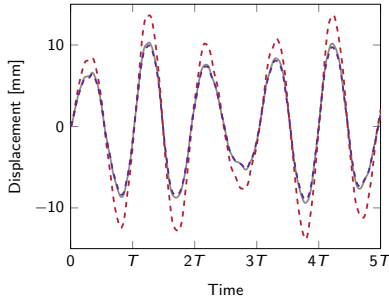
(b) Steady-state regime.

- Reference linear (---) and nonlinear (—) solutions, reduced order model nonlinear solution (— —).

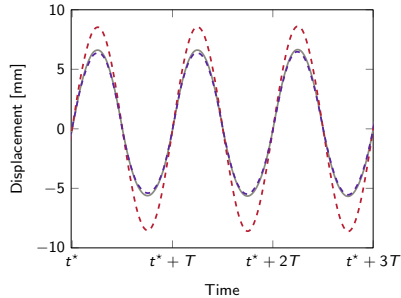


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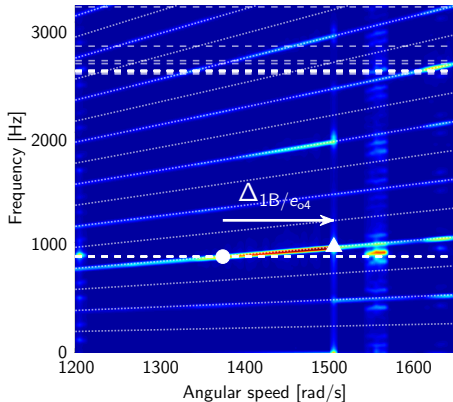


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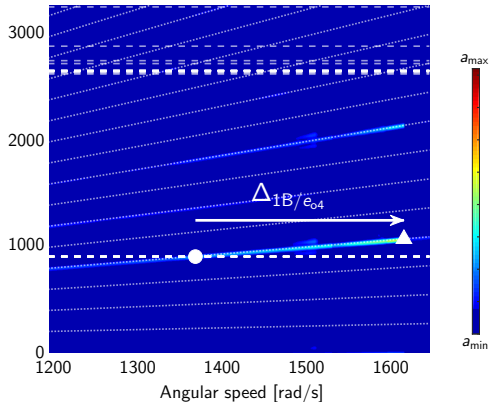
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# Interaction maps of the radial displacement at $LE$



(a) Without geometric nonlinearities.

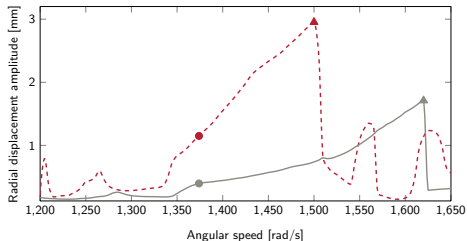


(b) With geometric nonlinearities.

► Interaction maps, predicted linear (●) and nonlinear (▲) resonances.



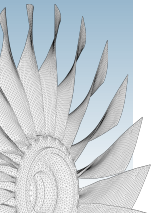
## Nonlinear frequency response curve



► NFRC without (—) and with (—) geometric nonlinearities, predicted linear (●) and nonlinear (▲) resonances.

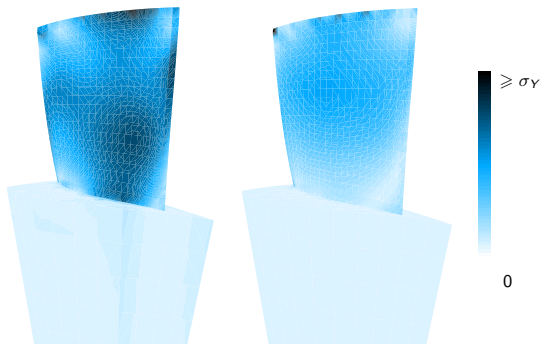
### Observations

- Contact stiffening
- Amplitude jumps
- Influence of geometric nonlinearities
  - Smoother interactions (see clearance consumption analysis)
  - 'Additional contact stiffening' (continuation procedure required for accurate quantification)





## Von Mises stress fields



(a) Without geometric nonlinearities. (b) With geometric nonlinearities.

► Von Mises stress fields at the resonance.

### Comparison

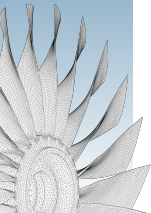
- Zones of maximal stresses not at the same locations
- Non-negligible stresses in the disk for the case without geometric nonlinearities
- Smaller stresses predicted with geometric nonlinearities (in line with predicted displacements)



Context  
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Full bladed disk  
**Conclusion**

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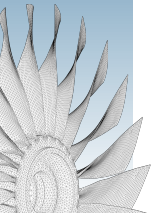
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- 3 Full bladed disk
- 4 Conclusion**





## Conclusion

- ▶ Methodology to study the rubbing interactions of full bladed disks with geometric nonlinearities
  - Projection basis: Craig-Bampton reduction basis and selection of their modal derivatives + second reduction of cyclic boundary
  - Geometric nonlinearities: STEP
  - Contact nonlinearities: Lagrange multipliers
- ▶ Reduced order models are an efficient alternative to full order models
- ▶ Influence of geometric nonlinearities not negligible
- ▶ Parametric reduced order models can be built to account for gyroscopic and centrifugal effects
- ▶ Methodology also compatible with the introduction of mistuning





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Thank you for your attention

