

A.M. Habraken, L. Duchêne, C. Bouffioux, C. Canalès



Characterization of Fatigue Behaviour, from Material Science to Civil Engineering Applications

OptiBri-Workshop

"Design Guidelines for Optimal Use of
HSS in Bridges"

3rd May 2017

Chantal
Bouffioux



Outline

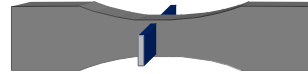


Small samples

Static & fatigue tests

Material behaviour

- Static
- Fatigue (small size)



Large welded plates

Fatigue tests
Residual stress

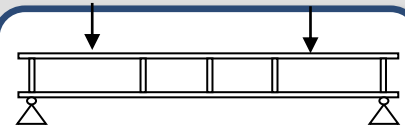
Effect of:

- size & machining
- welding
- post-treatments

Fatigue behaviour

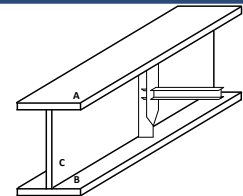


Future applications



Beams

- Size effect
- Laws validation



Critical bridge detail

- interest of HSS

Outline

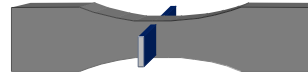


Small samples

Static & fatigue tests

Material behaviour

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Large welded plates

**Fatigue tests
Residual stress**

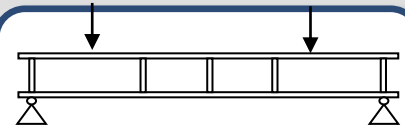
Effect of:

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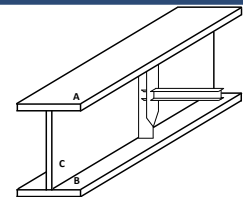


Future applications



Beams

- Size effect
- Laws validation



Critical bridge detail

- interest of HSS

Small samples

- **4 materials:**

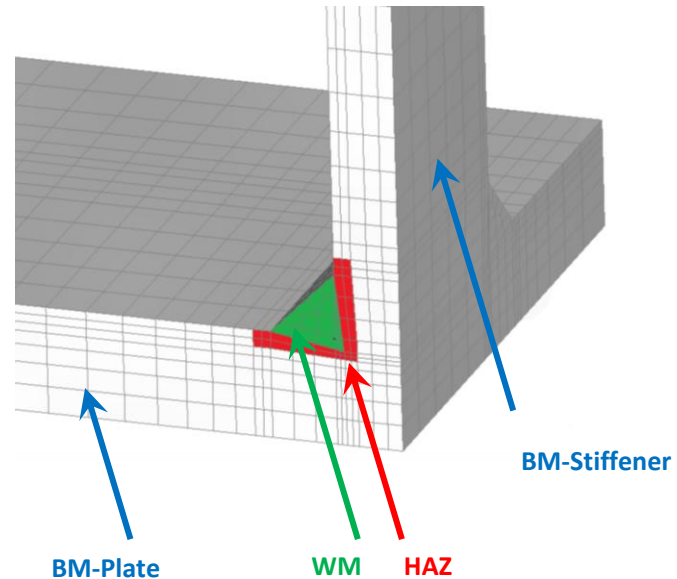
- **Base Material: BM** (HSS - S690QL)

- **Heat Affected Zone:**

- **HAZ1** (25 mm thick)

- **HAZ2** (40 mm thick)

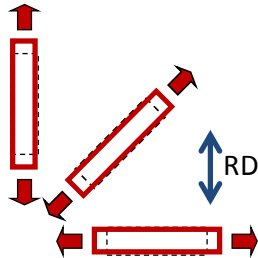
- **Welded Metal: WM**



Small samples

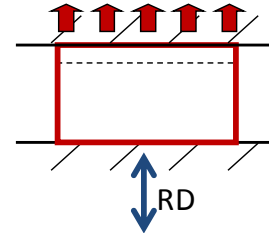
- **Static tests:**

Tensile test



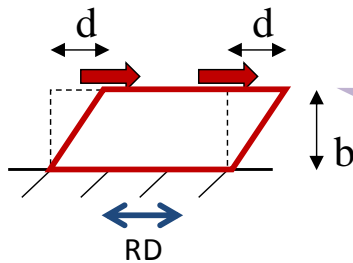
or

Large tensile test



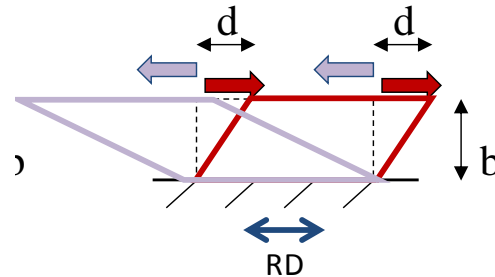
+

Shear test



+

Bauschinger shear test



Small samples

- **Static behavior – material laws & parameters :**

Elastic part: Hooke's law: E, ν

Plastic part: Hill's law (Hill48):

$$F_{HILL}(\underline{\sigma}) = \frac{1}{2} \left[H(\sigma_{xx} - \sigma_{yy})^2 + G(\sigma_{xx} - \sigma_{zz})^2 + F(\sigma_{yy} - \sigma_{zz})^2 + 2N(\sigma_{xy}^2 + \sigma_{xz}^2 + \sigma_{yz}^2) \right] - \sigma_F^2 = 0$$

Isotropic hardening: Voce formulation:

$$\sigma_F = \sigma_0 + K(1 - \exp(-n \cdot \varepsilon^{pl}))$$

Back-stress (kinematic hardening): Armstrong-Frederick's equation:

$$\dot{\underline{X}} = C_X (X_{sat} \underline{\dot{\varepsilon}}^{pl} - \frac{\underline{\varepsilon}^{pl}}{\tau} \cdot \underline{X})$$

E & ν : defined by tensile tests

F, G, H: defined by tensile tests in 3 directions (RD, TD, 45°)

N, σ_0 , K, n, C_X , X_{sat} : defined by Optim

Small samples

- **Static behavior – material data (inverse method):**

- **BM:** hardening fully kinematic
- **HAZ1** \approx **HAZ2:** same static behaviour
- **WM**

For fatigue tests:

For FEM:

$$\sigma_{u,eng} = F_i / A_0$$

$$\sigma_{u,true} = F_i / A_i$$

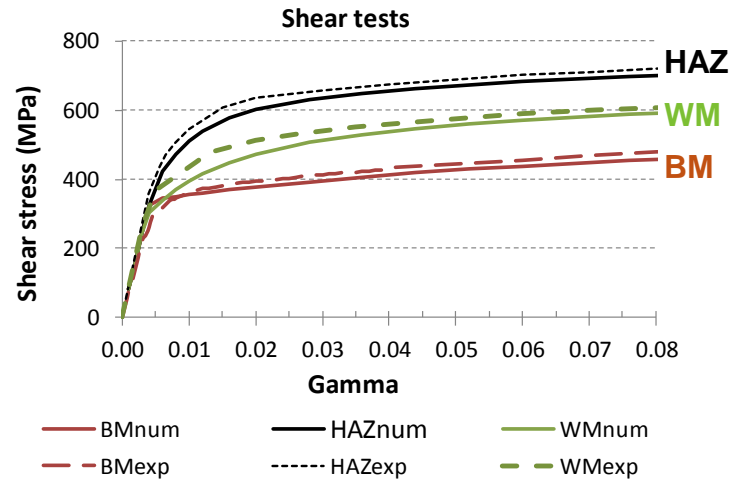
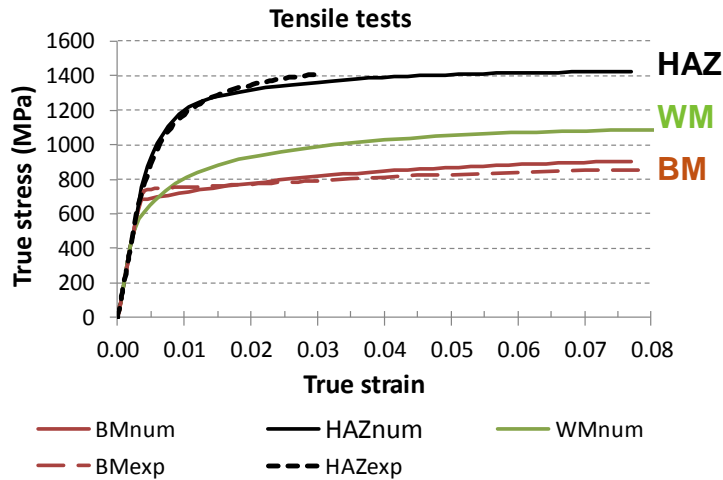
	Ultimate tensile strength (Mpa)	
	$\sigma_{u,eng}$	$\sigma_{u,true}$
BM (S690QL)	838	905
HAZ1, HAZ2	1338	1424
WM	1008	1101

Data for Hooke, Hill, Voce and Armstrong-Frederick laws (units: MPa, s)

Material	Elast. data		Yield locus				Isotropic hardening			Kinematic hardening	
	E	ν	F	G	H	$N=L=M$	K	σ_0	n	C_x	X_{sat}
BM (S690QL)	210 116	0.3	1	1	1	3.9	0	674	0	31.9	167
HAZ1, HAZ2	210 000	0.3	1	1	1	4.45	371	827	511	52.5	152
WM	210 000	0.3	1	1	1	3.2	241	531	285	42.6	218

Small samples

- **Static behavior – comparison of material behavior:**



Small samples

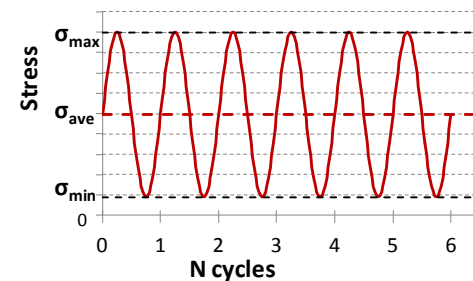
- **Fatigue tests:**

- On vibrophore
- Axial loading
- Frequency: 100 – 150 Hz
(→ correction factor)

$$R = \sigma_{\min} / \sigma_{\max} = 0.1 \text{ or } 0.2 \text{ or } \dots$$

Material	Smooth	Notch
BM (S690QL)	4 R	3 geom.
HAZ1	1 R	1 geom.
HAZ2	2 R	1 geom.
WM	2 R	1 geom.

Vibrophore



Small samples

- Fatigue behavior – material laws & parameters :**

Multiaxial Lemaître Chaboche fatigue model

$$\frac{\partial D}{\partial N} \begin{cases} = 0 & \text{if } f_D < 0 \\ = [1 - (1 - D)^{\beta+1}]^\alpha \left(\frac{\tilde{A}_{II}}{M} \right)^\beta & \text{if } f_D \geq 0 \end{cases}$$

$$f_D = A_{II} - A_{II}^*$$

$$A_{II} = \frac{1}{2} \sqrt{\frac{3}{2} (\hat{\sigma}_{ij\max} - \hat{\sigma}_{ij\min}) (\hat{\sigma}_{ij\max} + \hat{\sigma}_{ij\min})} \quad \text{with } \hat{\sigma}_{ij} = \sigma_{ij} - \sum_k \frac{1}{3} \sigma_{kk}$$

$$A_{II}^* = \sigma_{I0} (1 - 3 \cdot b \cdot \sigma_{Hm}) \quad (\text{Sines' criterion})$$

$$\tilde{A}_{II} = \frac{A_{II}}{1 - D}$$

$$M = M_0 (1 - 3 \cdot b \cdot \sigma_{Hm})$$

$$\alpha = 1 - \left(a \left(\frac{A_{II} - A_{II}^*}{\sigma_u - \sigma_{eq\max}} \right) \right)$$

$$\sigma_{Hm} = \frac{1}{3} \left[\frac{1}{T} \int_T \text{Tr} (\underline{\underline{\sigma}}(t)) dt \right]$$

D: damage val., 0: sound material, 1: rupture

N: number of cycle

A_{II} : 2nd invar. of amplit. deviator of σ tensor

A_{II}^* : fatigue limit

f_D : damage yield locus

σ_{Hm} : mean hydrostatic stress

$\sigma_{eq\max}$: maximum Von Mises stress per cycle

$\langle x \rangle$: = x if x > 0 else = 0

b = $1/\sigma_u$

σ_u : ultimate tensile stress

σ_{I0} : endurance limit= fatigue limit at null σ_{mean}

a, M_0 , β : other material data to define

Small samples

- Fatigue behavior – material laws & parameters :**

Volume averaged stress gradient method

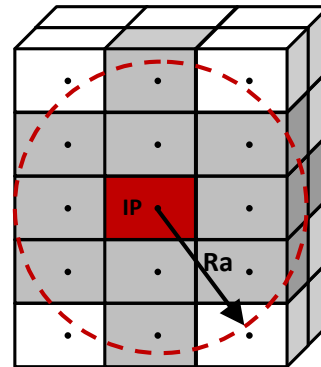
For each element, variables χ_{ip} : replaced by an average value of all the elements with their integration point inside the circle with a radius **Ra**

$$\overline{\chi_{ip}} = \frac{1}{V} \cdot \sum_{i=1}^{Nelem} \chi_{ip,i} \cdot V_i$$

$$\chi_{ip} = \{A_{II}, \sigma_{eqmax}, \sigma_{Hm}\}$$

$$V = \sum_{i=1}^{Nelem} V_i$$

- A_{II} : 2nd invar. of amplit. deviator of σ tensor
- σ_{eqmax} : maximum Von Mises stress per cycle
- σ_{Hm} : mean hydrostatic stress
- Ra**: material data to define



Small samples

- **Fatigue behavior – material data (inverse modelling):**

HAZ1 ≈ **HAZ2**: same fatigue behavior

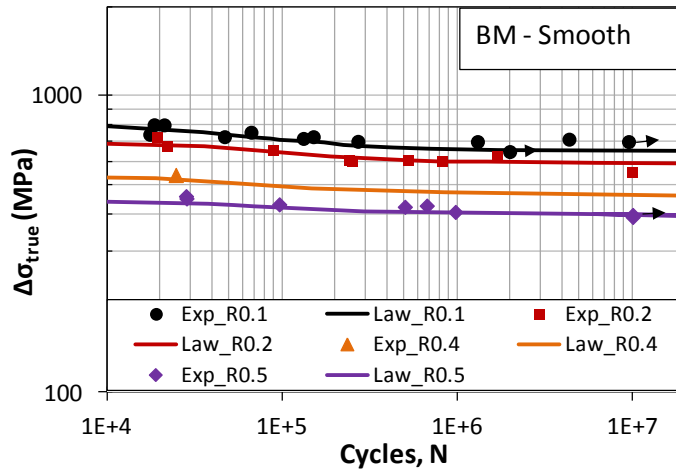
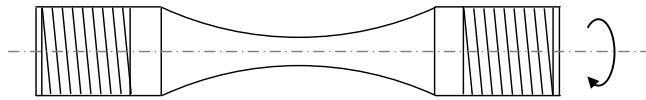
Material	σ_U (Mpa)	σ_{l0} (Mpa)	b	β	a	M0	Ra (mm)	$a^*(M0^{-\beta})$
BM	905.0	580.0	1.10 E-03	0.17	1	5.385 E+30	0.06	5.966E-06
HAZ1, HAZ2	1424.0	428.4	7.02E-04	2.094	1	4.410E+05	0.00	1.516E-12
WM	1101.0	319.4	9.08E-04	0.161	1	7.245E+32	0.00	5.182E-06

Small samples

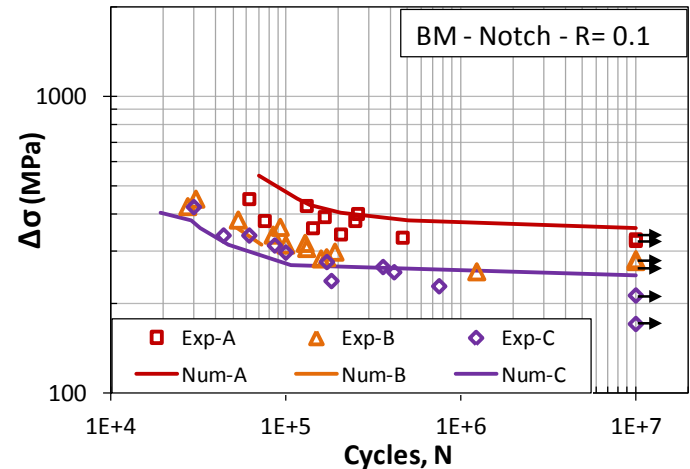
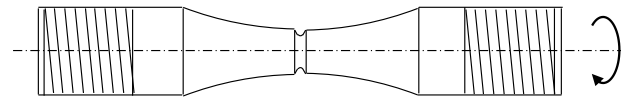
- Fatigue behavior – comparison experiments & fatigue law:

Base material (BM)

Smooth samples, 4 R



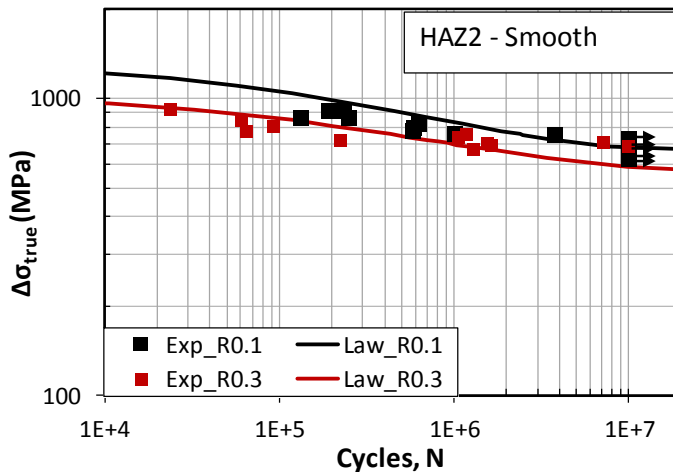
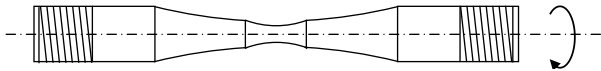
Notched samples, 3 geom.



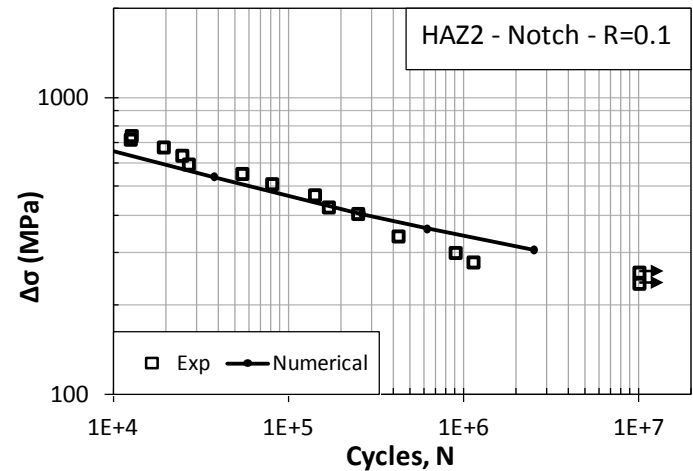
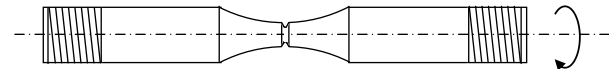
Small samples

- Fatigue behavior – comparison experiments & fatigue law:**
Heat affected zone (HAZ) with HAZ1 \approx HAZ2

Smooth samples, 2 R



Notched samples, 1 geom.

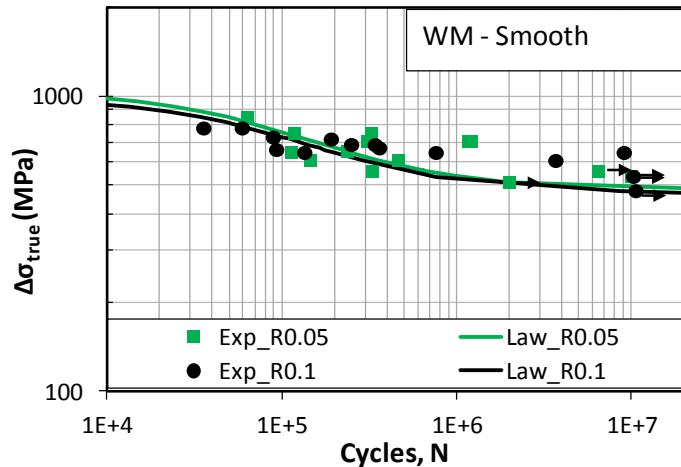
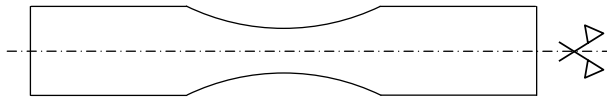


Small samples

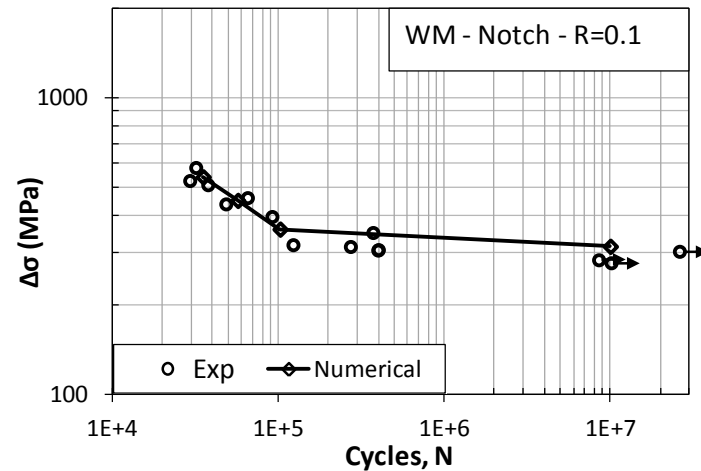
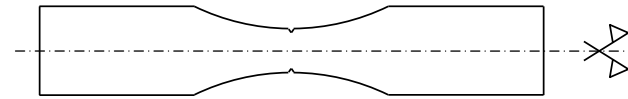
- Fatigue behavior – comparison experiments & fatigue law:

Weld metal (WM)

Smooth samples, 2 R



Notched samples, 1 geom.



Outline

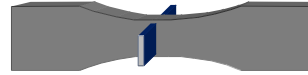


Small samples

Static & fatigue tests

Material behaviour

- Static
- Fatigue (small size)



Large welded plates

Fatigue tests
Residual stress

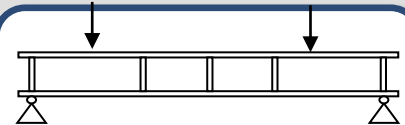
Effect of:

- size & machining
- welding
- post-treatments

Fatigue behaviour

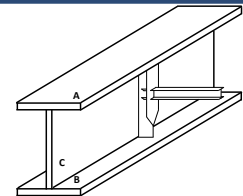


Future applications



Beams

- Size effect
- Laws validation

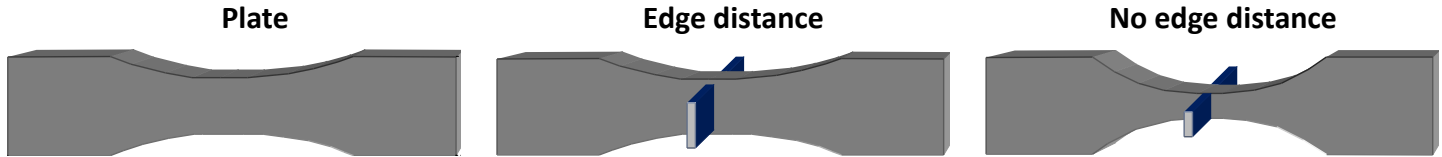


Critical bridge detail

- interest of HSS

Large welded plates

- Fatigue tests on plates (length= 1070 mm):



Case	Post-treatment	plate thickness (mm)	Stiffener thickness (mm)	Stiffener length (mm)	distance to edge	stress ratio R
Plate	No weld	25	-	-	-	0.1
A (ref case)	PIT	25	15	60	✓	0.1, 0.3, 0.5
B	PIT	15	15	60	✓	0.1
E	PIT	25	15	60	no	0.1
H	PIT	40	15	60	no	0.1
C	TIG remelting	15	15	60	✓	0.1
D	TIG remelting	25	15	60	✓	0.1
F	TIG remelting	25	15	40	✓	0.1
G	TIG remelting	15	6	60	✓	0.1
I	No post-treatment	15	15	60	✓	0.1

Large welded plates

- Fatigue behavior – material data (inverse modelling):**

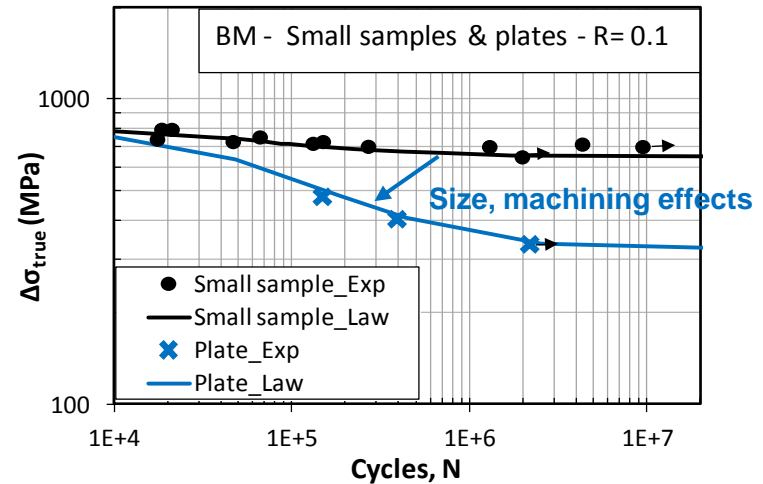
Material	σ_U (Mpa)	σ_{10} (Mpa)	b	β	a	M0	Ra (mm)	$a^*(M0^{-\beta})$
BM-SS	905.0	580.0	1.10 E-03	0.17	1	5.385 E+30	0.06	5.966E-06
BM-plate	905.0	203.0	1.10 E-03	0.17	1	5.385 E+30	0.06	5.966E-06

σ_{10} = endurance limit= fatigue limit at null σ_{mean}

Small samples, Ra= 0.8 μ m,
length: 96 mm



Plates, as produced,
length: 1070 mm



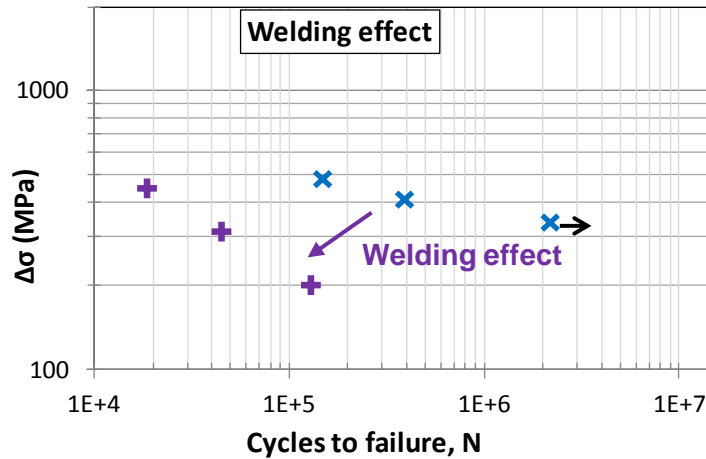
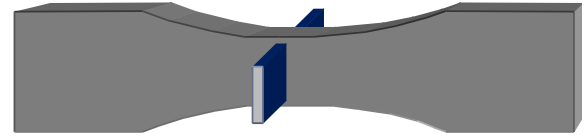
Large welded plates

- Fatigue tests: **welding effect**

Plates



Welded plates



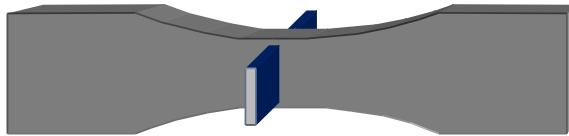
+ |

x | BM-Plate

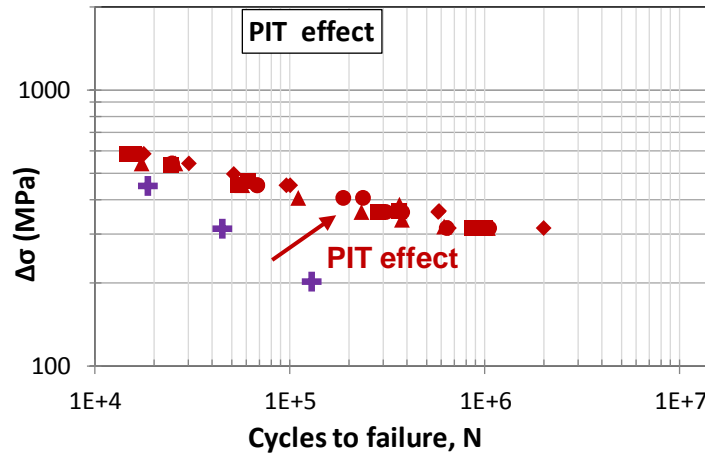
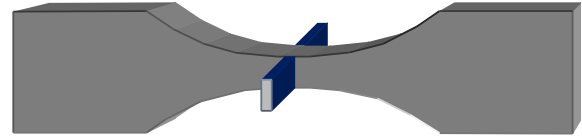
Large welded plates

- Fatigue tests: **PIT post-treatment effect**

Welded plates



Welded plates + PIT, ≠ geom.



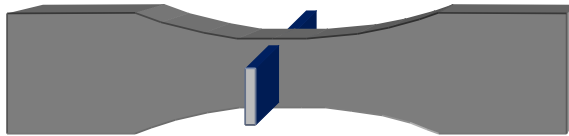
PIT: 4 cases
low effect of geometry

■ A ● B ◆ E ▲ H + I

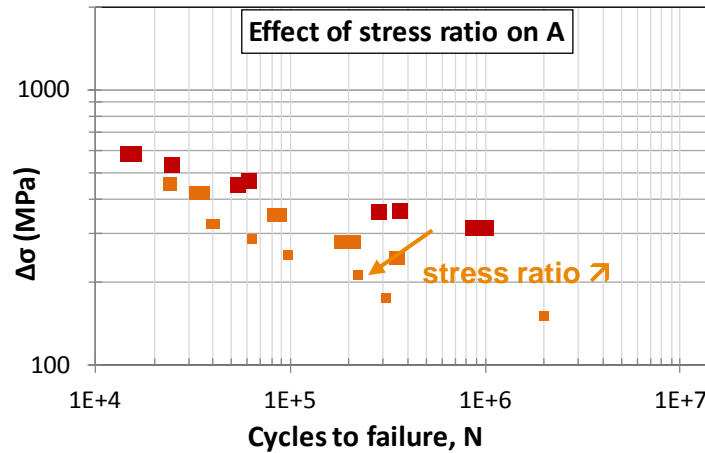
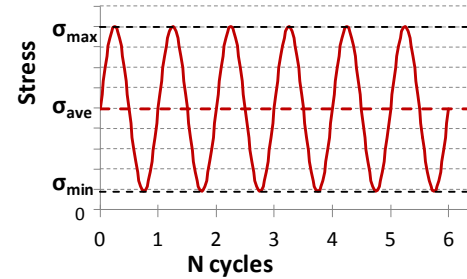
Large welded plates

- Fatigue tests: **Stress ratio effect**

Welded plates + PIT



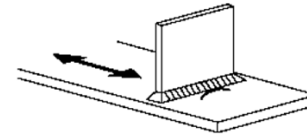
$$R = \sigma_{\min} / \sigma_{\max} = 0.1 \text{ or } 0.3 \text{ or } 0.5$$



■ A_0.1 ■ A_0.3 ■ A_0.5

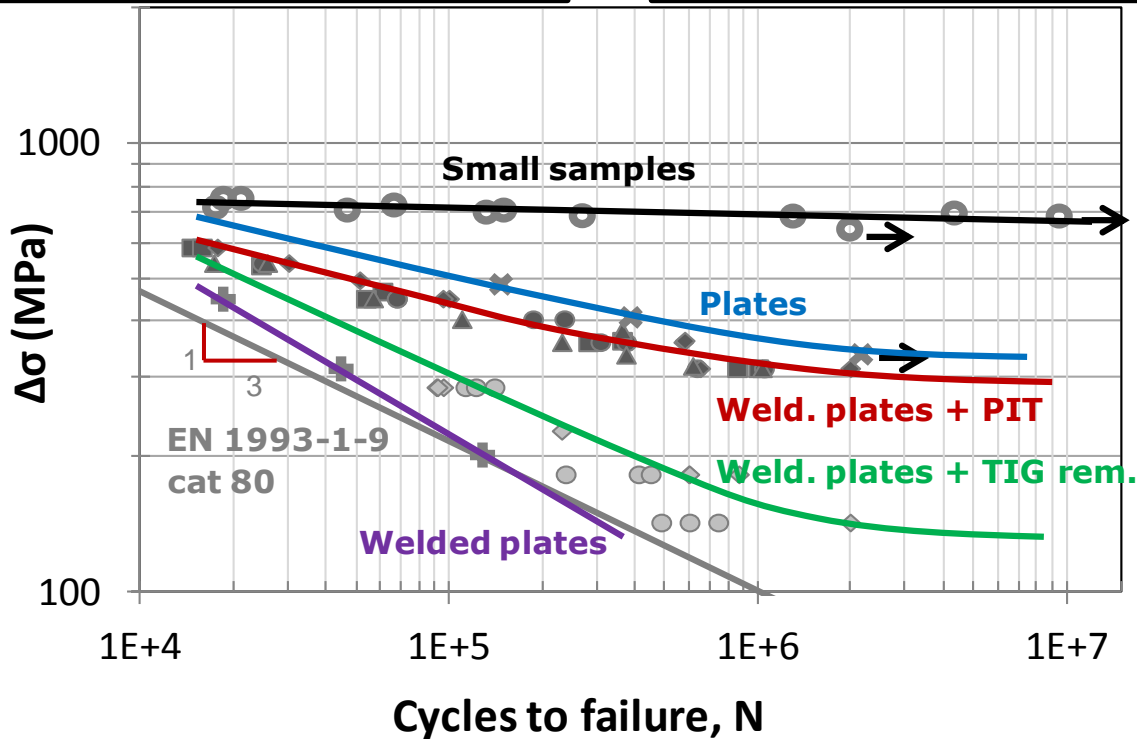
Large welded plates

- Fatigue tests



Summary, all cases, R= 0.1

EN 1993-1-9, $\Delta\sigma_c = 80$ MPa



Large welded plates

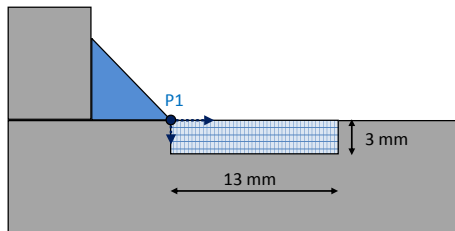
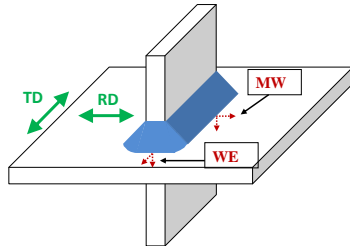
Residual stress measurement

Several cases:

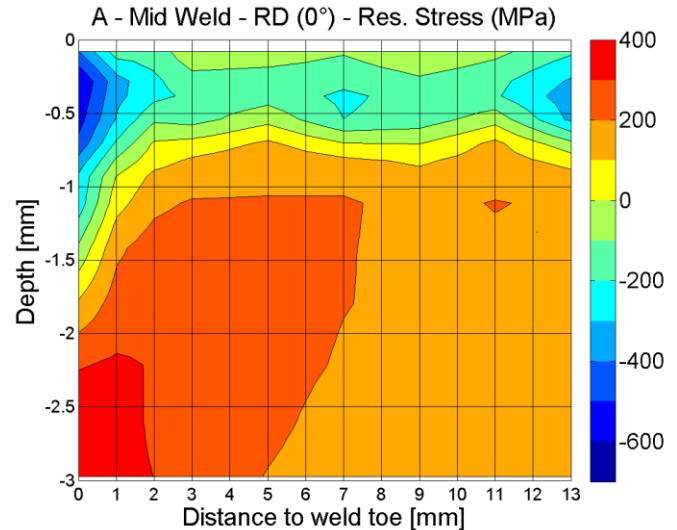
- 3 geometries
- 3 cases: PIT, TIG remelting, no post-treat.
- Mid-weld (MW), weld edge (WE), RD, TD
- X-ray, neutron diffraction

Welding & post-treatment effects:

- up to depth \approx 3-4 mm
- up to weld toe distance \approx 6-7 mm



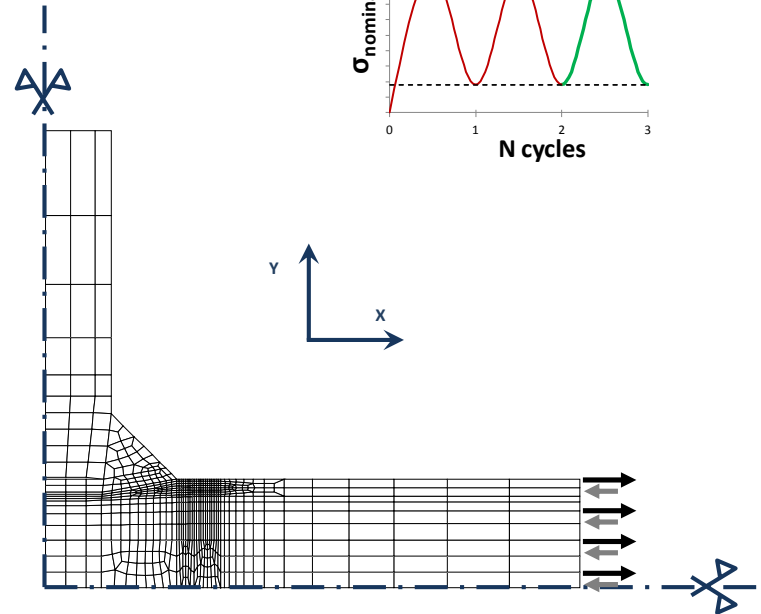
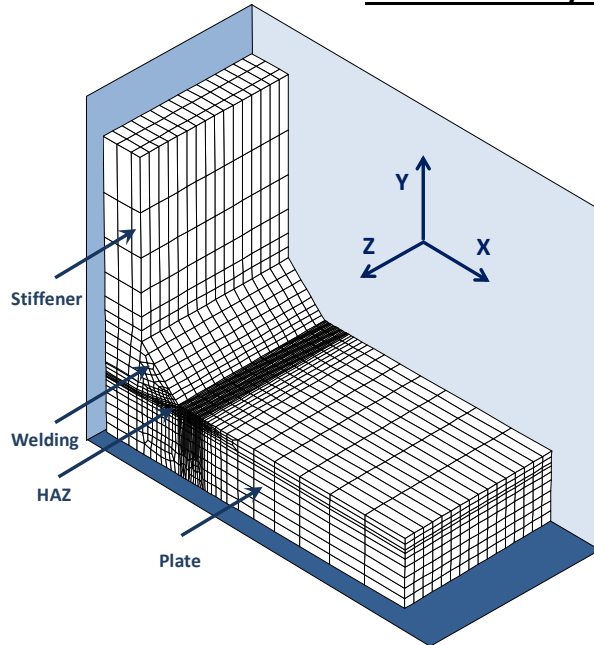
Ex: ref. case: A, with post-treatment (X-ray)



Large welded plates

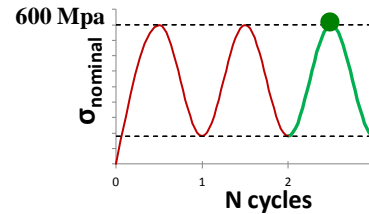
- Fatigue - numerical analysis

FEM model & symmetry conditions



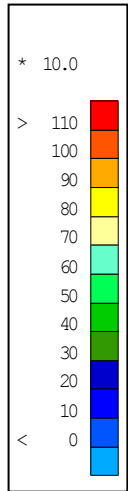
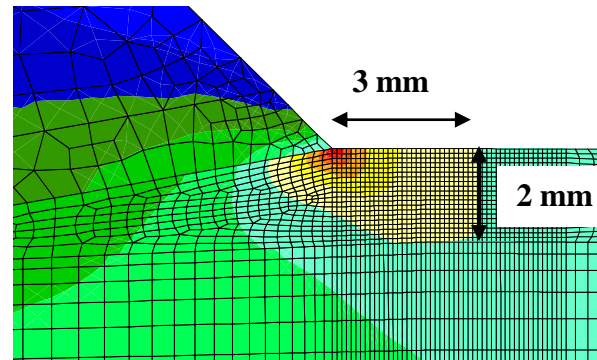
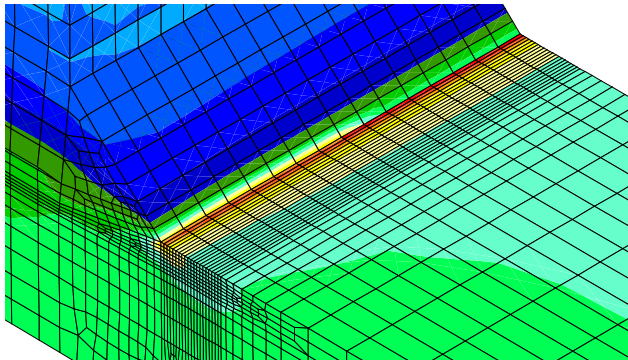
Large welded plates

- Fatigue - numerical analysis



Example of results without post-treatment, no σ_{res}

σ_{xx} (Mpa) for nominal stress = 600 Mpa ($\sigma_{nom} = F/A_0$ and $A_0 =$ section of web)



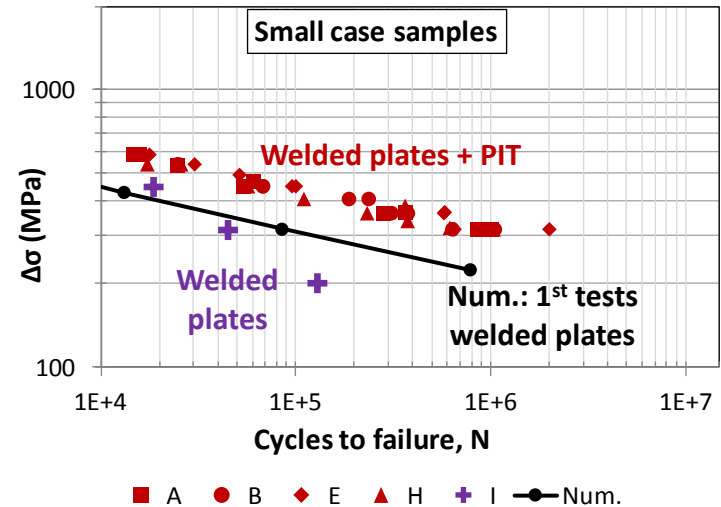
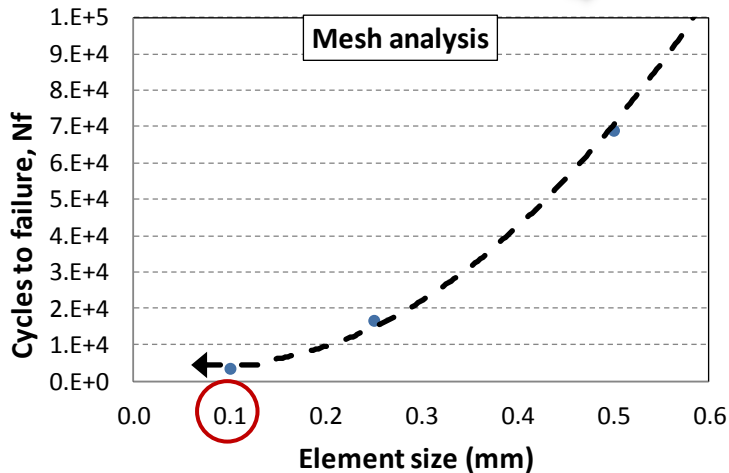
Stress concentration at weld toe:

- up to depth \approx 2 mm (in HAZ)
- up to weld toe distance \approx 3 mm

Large welded plates

Fatigue - numerical analysis

1. Mesh analysis → element size at weld toe: 0.1 mm (results not mesh dependent)
2. For several stress ranges and a specified stress ratio (here: 0.1):
 - Numerical analysis → Stress distribution → number of cycle at rupture



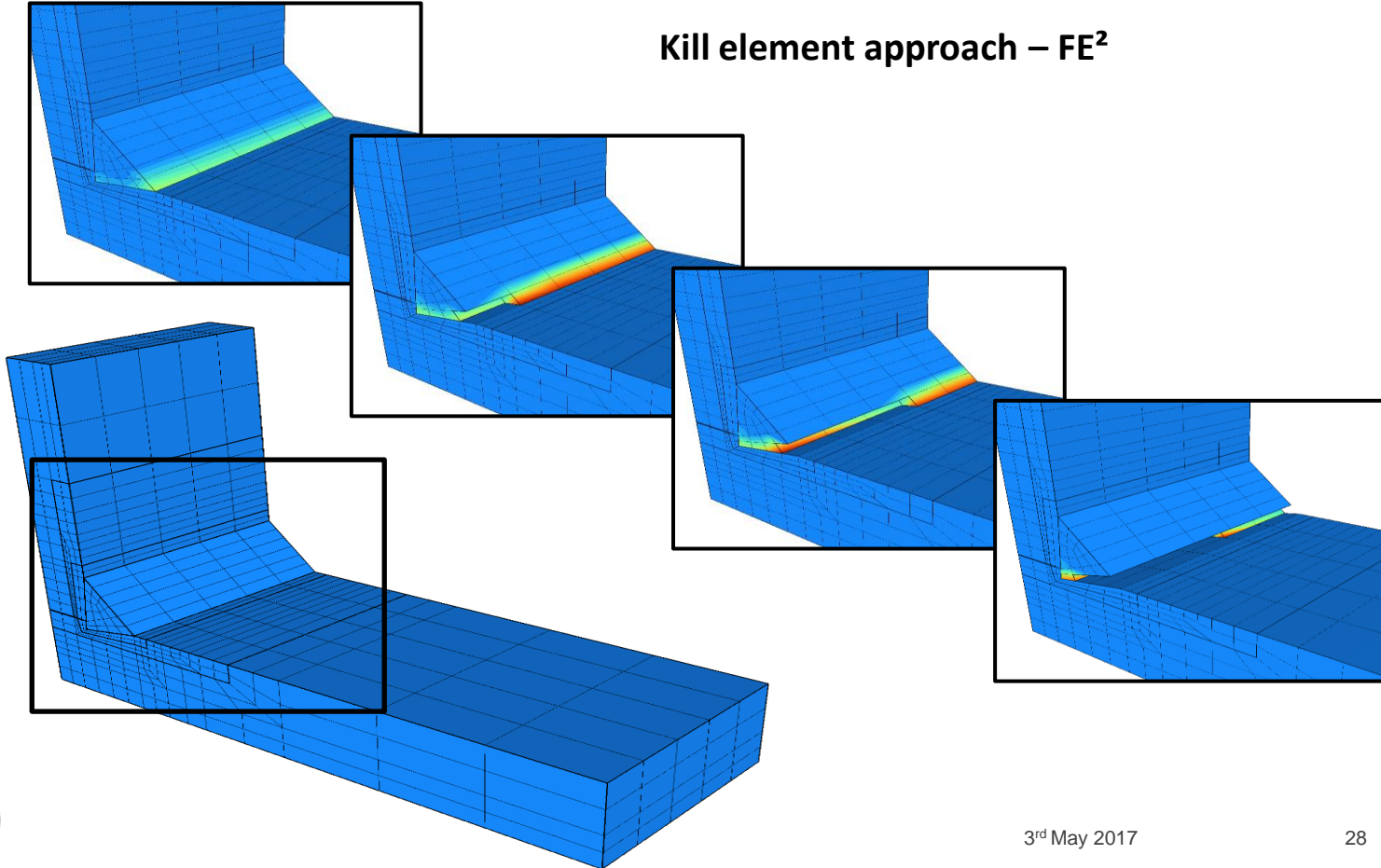
Next steps:

- to add σ_{res} to model (welding + post-treatment)
- to improve fatigue mat. data of HAZ (σ_{10})
- to study beams and critical bridge detail

Large welded plates

- Fatigue - numerical analysis, crack propagation

Kill element approach – FE²



Summary & conclusions

Important test campaign has been done to prepare numerical fatigue study of bridge details:

- **Static tests** on small samples → **Static behaviour** BM, HAZ, WM
- **Fatigue tests** on large welded plates → **Fatigue behaviour (small size)**
- **Fatigue tests on large welded plates** → **Effects of size, surface roughness, welding, geometry, post-treatments**
- **Residual stresses measurements** → **Effect of post-treatments for num. analysis**
- **Mesh analysis** → **welded plates: elem. size at weld toe: 0.1 mm**
- **Crack propagation** → **deep analysis of fatigue study**

→ **Positive effect in fatigue life is shown on welded plates**
Fatigue characterisation almost ready for analysis on critical bridge detail

Thank you for your attention!



E-Mail chantal.bouffioux@ulg.ac.be

Telefon +32 (0) 4 366 92 19

Fax +32 (0) 4 366 95 34

University of Liège

ARGENCO department

Quartier Polytech 1,

Allée de la découverte, 9 Bât B52/3

B-4000 Liège BELGIUM