


# Fatigue Assessment Diagram in Updating Failure Probability

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## 1. Introduction



The support structure contributes to a significant part of the Levelized Cost Of Energy

- Reduce O&M costs is the key!
- How?
  - Reliability- and risk-based maintenance strategies
  - Updating of the reliability based on e.g. inspections performed during the design lifetime.

**The aim of this research:**

**Update failure probability using the Fatigue Assessment Diagram as a Limit State Function**

Jacket foundation in the "alpha-ventus" wind farm (photograph taken by Matthias Ibeler)

## 2. Fatigue Assessment Diagram

- A method for assessing the acceptability of cracks in metallic structures, following BS 7910
- The acceptable region area decreases when more information about stress/strain data available

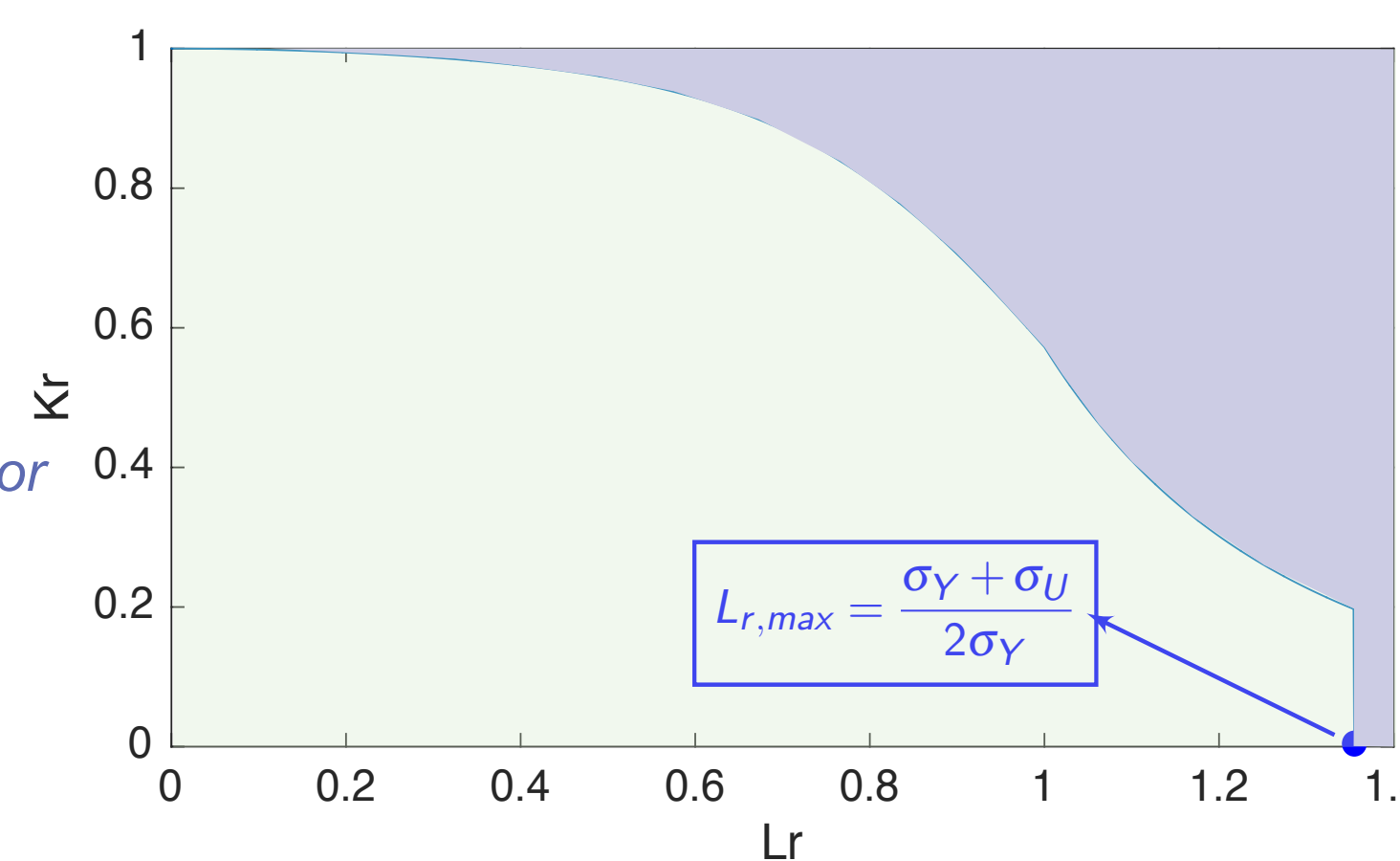
The coordinate of an assessment point:

Reference stress,  $f(\text{joint type, crack size})$

Stress Intensity Factor

Yield strength

Fracture Toughness

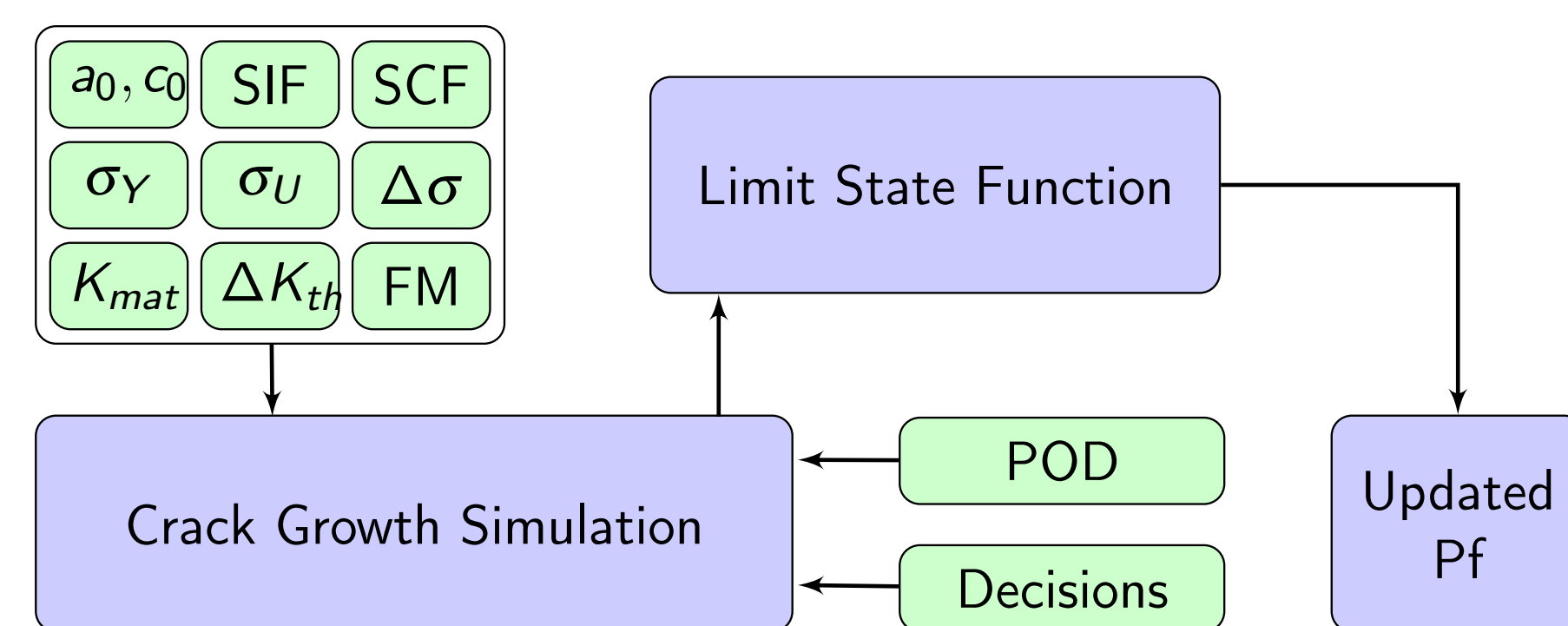
$$\left( L_r = \frac{\sigma_{ref}}{\sigma_Y}; K_r = \frac{K_I}{K_{mat}} \right)$$


Fatigue Assessment Diagram  
acceptable; unacceptable

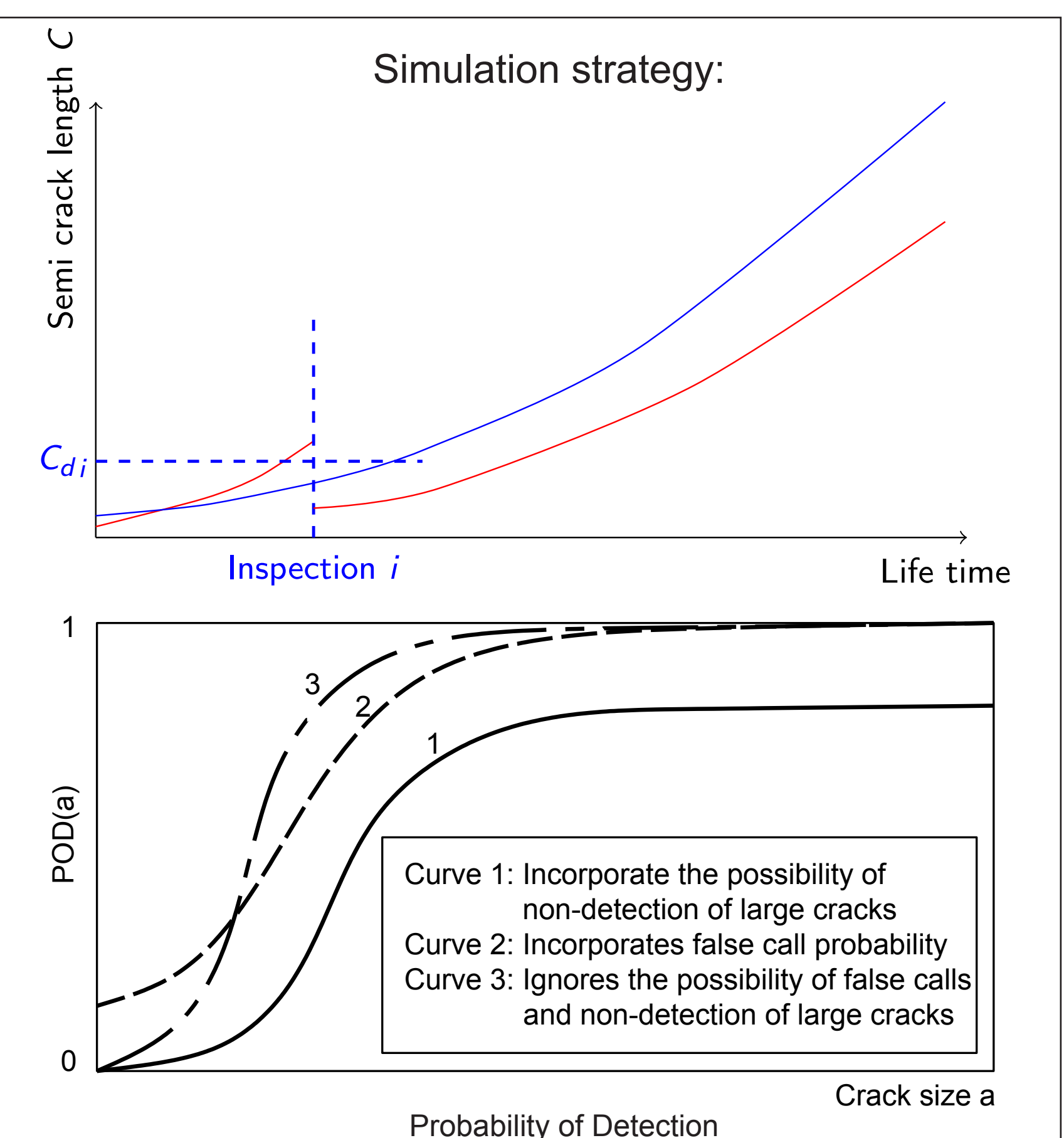
## 3. Methodology

Variable	Distr.	Mean	CoV
S Stress range [MPa]	W	k=0.8	$N(\mu, \sigma)$
$\sigma_Y$ Yield strength [MPa]	LN	368.75	0.07
$\sigma_U$ Ultimate strength [MPa]	LN	750	0.04
$\Delta K_{th}$ SIF range threshold	LN	160	0.4
$K_{mat}$ Fracture toughness	3p W	-	-
$C_1$ Paris law, 1 <sup>st</sup> line	LN	$4.8 \times 10^{-18}$	1.7
$C_2$ Paris law, 2 <sup>nd</sup> line	LN	$5.86 \times 10^{-13}$	0.6
$a_0$ Initial crack depth	LN	0.15	0.66
$a_0/c_0$ Initial aspect ratio	LN	0.6	0.40
$B_{scf}$ Uncertainty in SCF	LN	1	0.05
$B_{sif}$ Uncertainty in SIF	LN	1	0.05

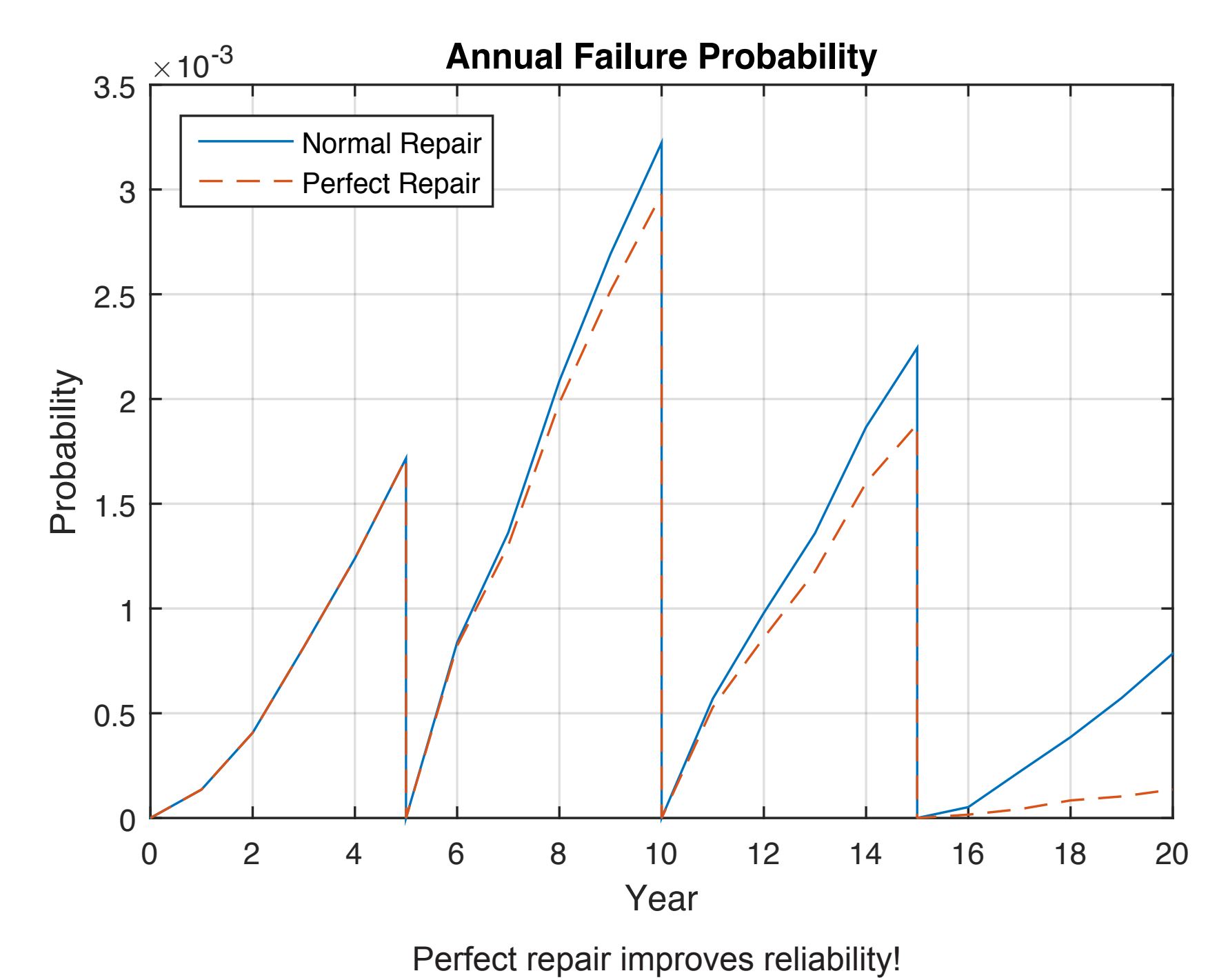
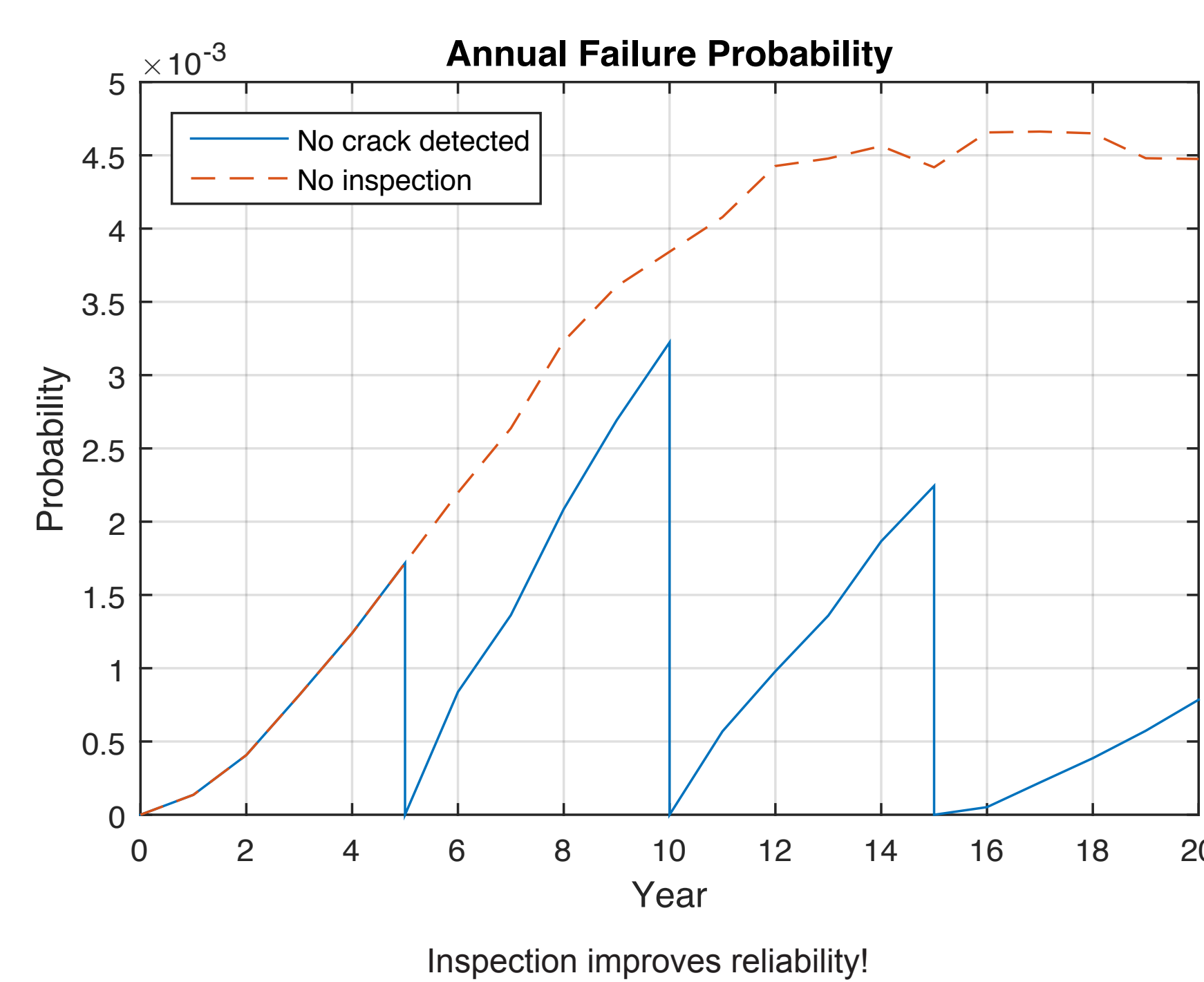
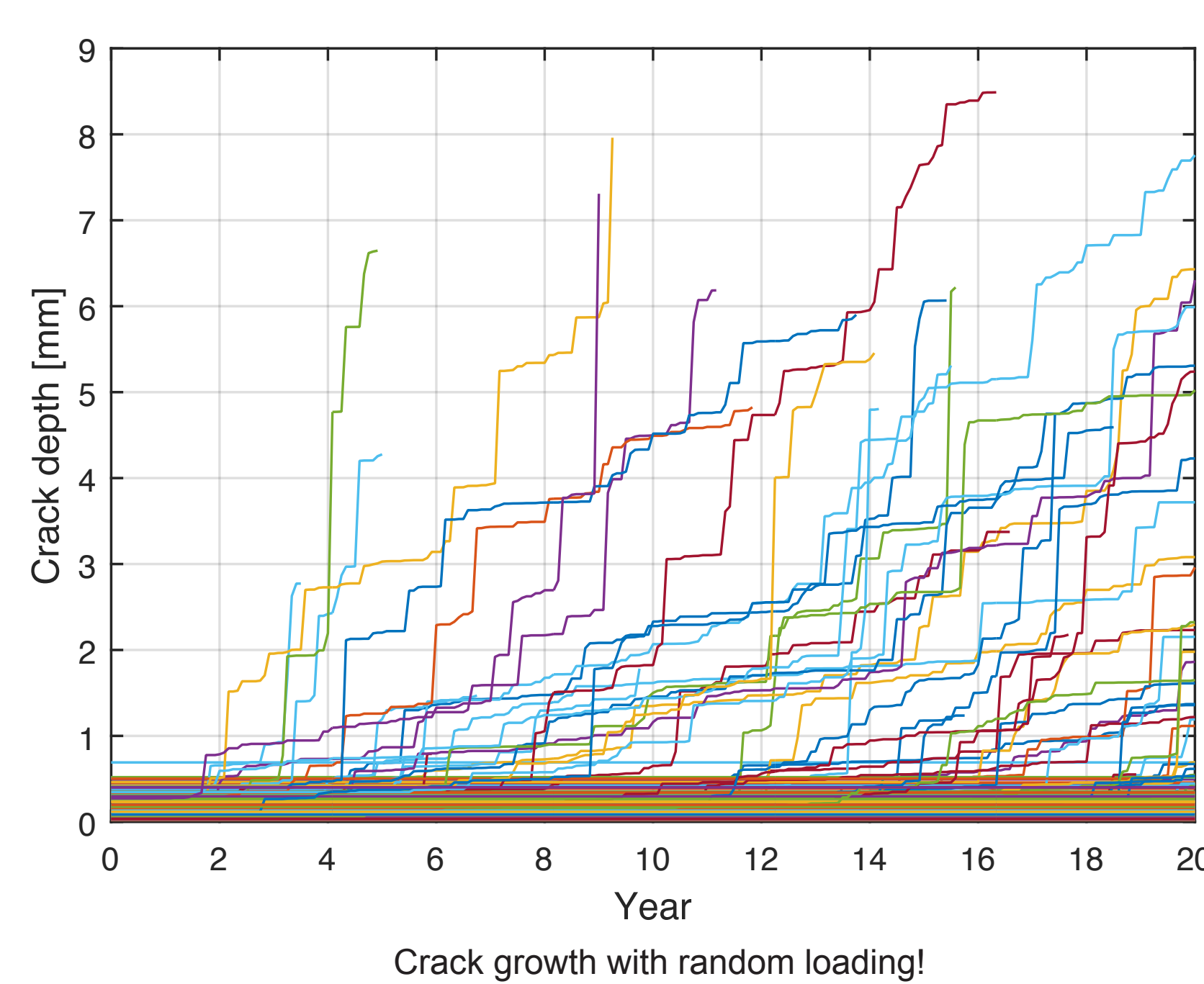
Parameters of the random variables



- Semi-elliptical surface crack
- Paris-Erdogan law
- Coupled depth and length
- Constant stress-range for short periods of time
- No crack detected
- Crack detected & repaired:
  - normal repair
  - perfect repair
- Crack detected, not repaired



## 4. Results



## 5. Conclusion and Perspective

- For surface cracks, FAD and the conventional LSFs give similar failure probabilities.
- FAD can be used to assess failure for crack depths larger than plate thickness.
- The FAD approach showed to work well in up- dating failure probability of a joint.

FAD approach can be used further in inspection planning for OWT support structures:

- to include systems effects,
- and for reducing the required safety factors at the design stage

## Acknowledgments

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**fnr's**  
FREEDOM TO RESEARCH