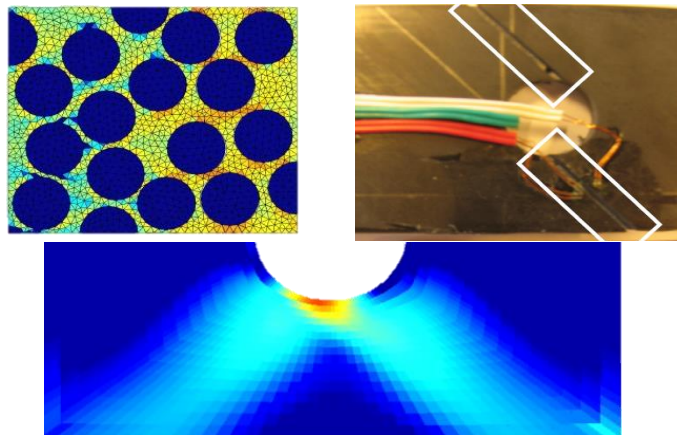


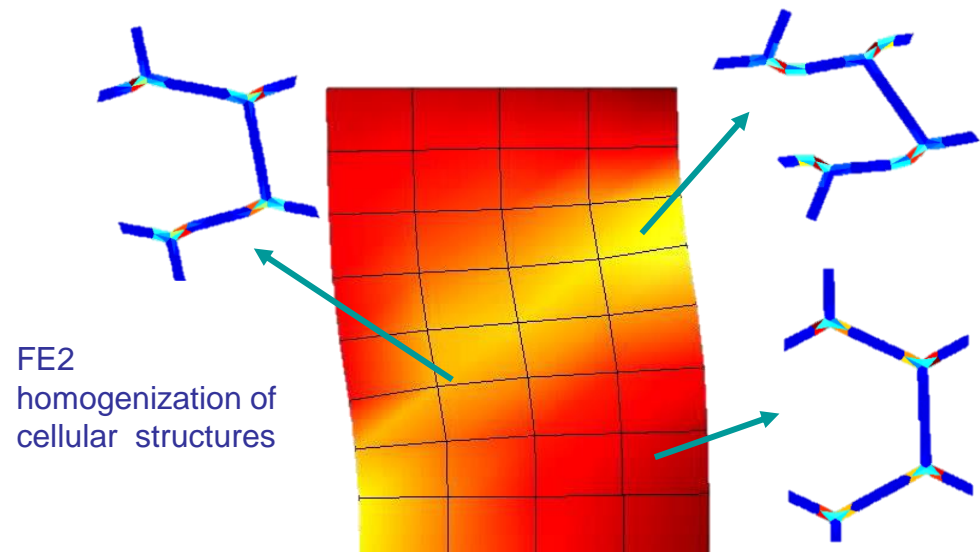
Muti-scale methods with strain-softening: damage-enhanced MFH for composite materials and computational homogenization for cellular materials with micro-buckling

L. Wu, V.-D. Nguyen, L. Adam (e-Xstream)

I. Doghri (UCL), L. Noels



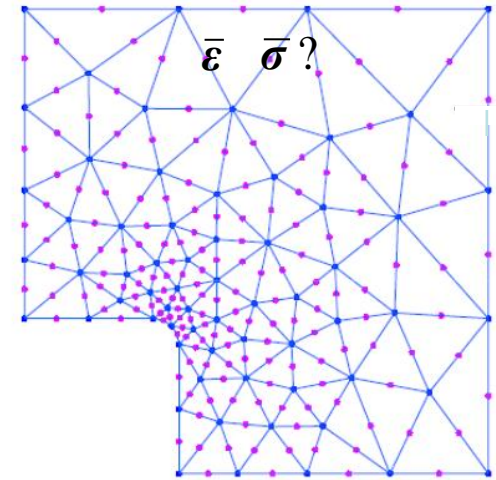
Non-local damage mean-field-homogenization



FE2
homogenization of
cellular structures

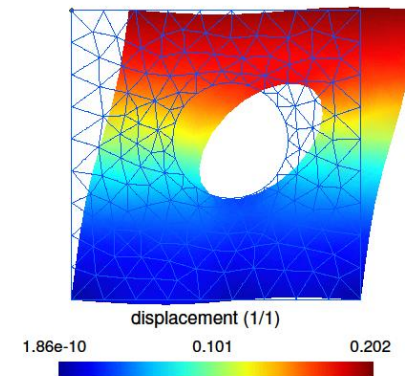
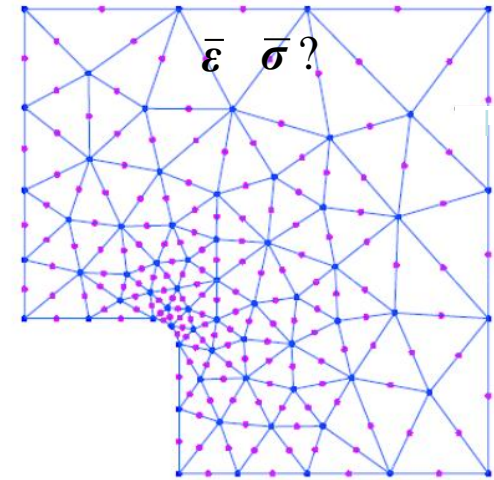
- Introduction
 - Finite element multi-scale modelling
 - Strain softening issues
- Non-local damage-enhanced mean-field-homogenization
- Computational homogenization for cellular materials
- Conclusions

- Computational technique: FE^2
 - Macro-scale
 - FE model
 - At one integration point $\bar{\epsilon}$ is known, $\bar{\sigma}$ is sought



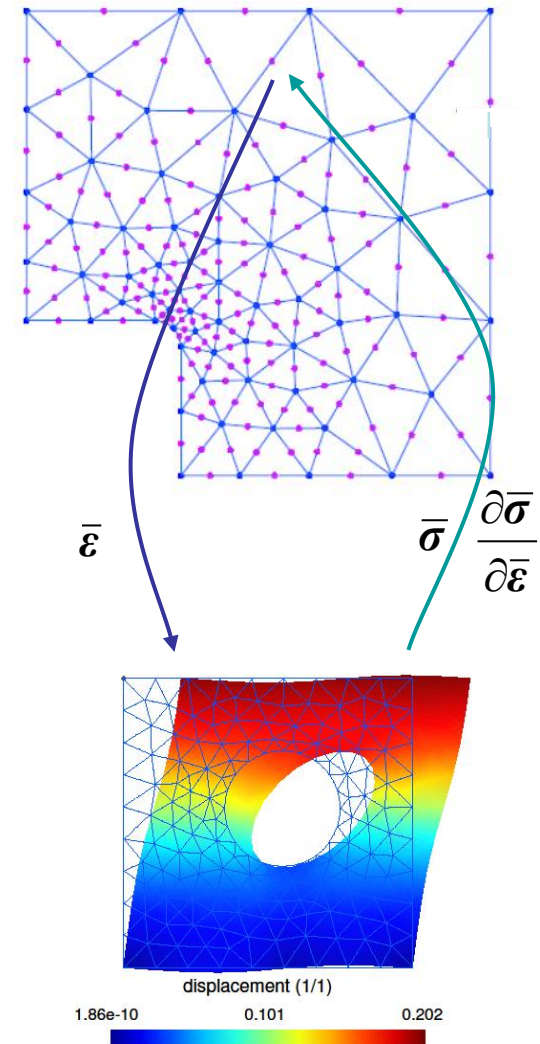
Multi-scale modelling: How?

- Computational technique: FE²
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 - Micro-scale
 - Usual 3D finite elements
 - Periodic boundary conditions



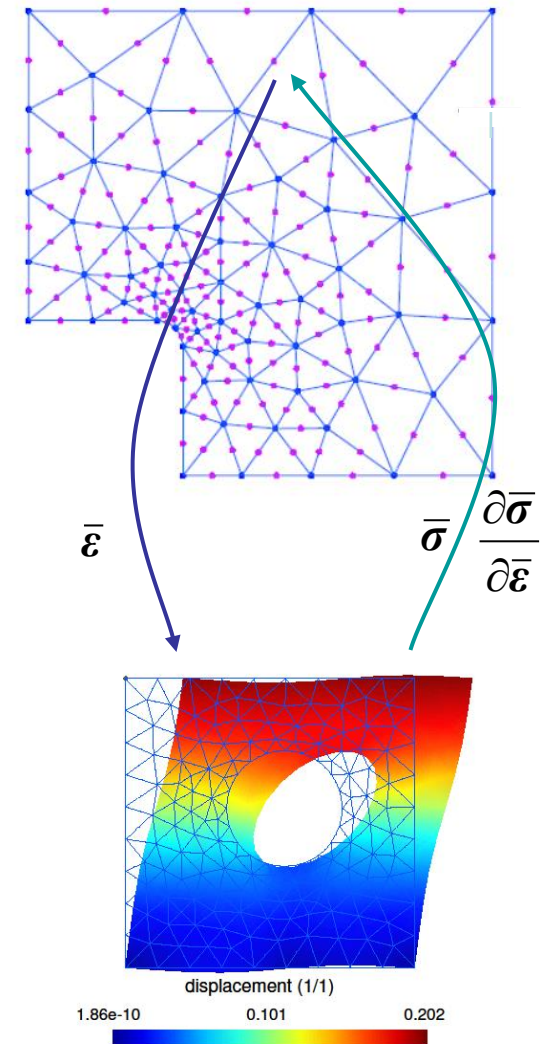
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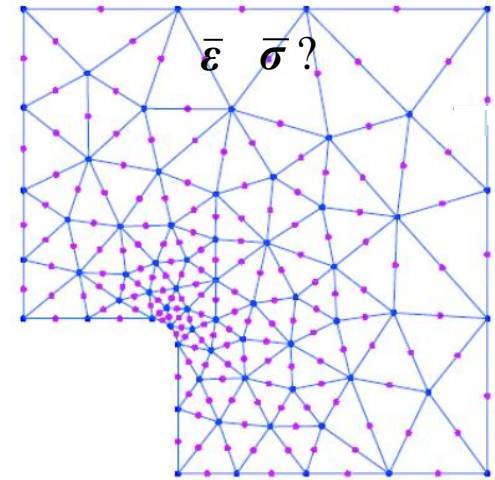
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 - Periodic boundary conditions
 - Advantages
 - Accuracy
 - Generality
 - Drawback
 - Computational time



Ghosh S et al. 95, Kouznetsova et al. 2002, Geers et al. 2010, ...

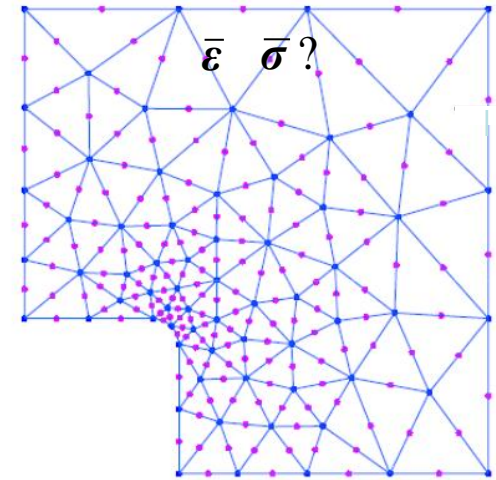
- Mean-Field Homogenization
 - Macro-scale
 - FE model
 - At one integration point $\bar{\epsilon}$ is known, $\bar{\sigma}$ is sought



- Mean-Field Homogenization

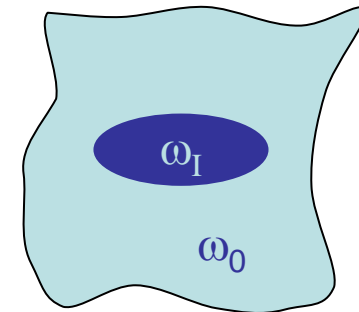
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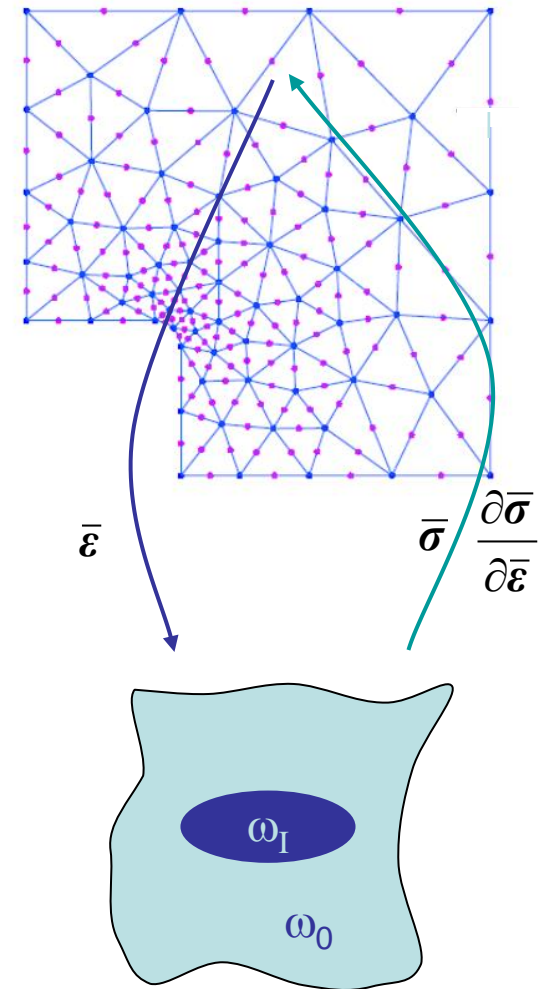
- Micro-scale

- Semi-analytical model
 - Predict composite meso-scale response
 - From components material models



- Mean-Field Homogenization

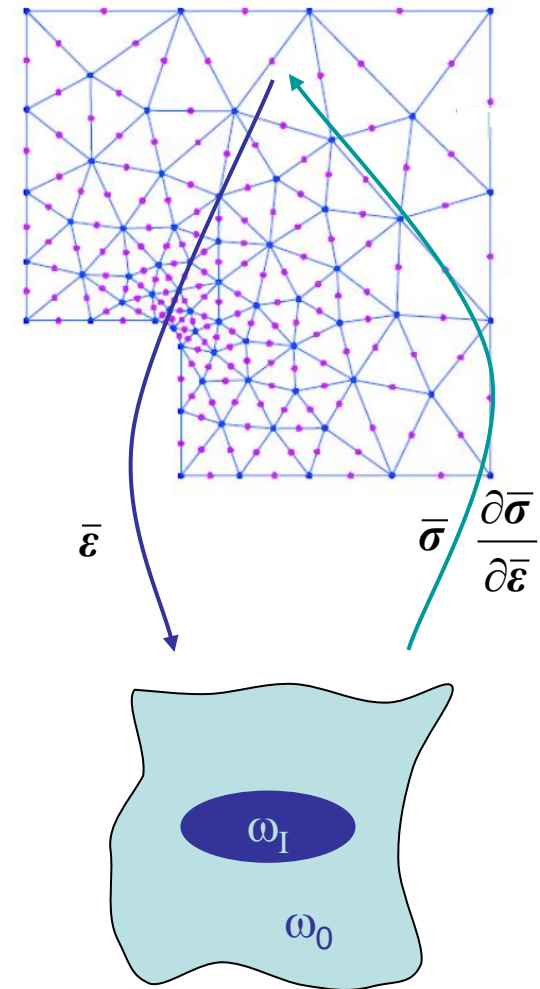
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Mori and Tanaka 73, Hill 65, Ponte Castañeda 91, Suquet 95, Doghri et al 03, Lahellec et al. 11, Brassart et al. 12, ...

- Mean-Field Homogenization

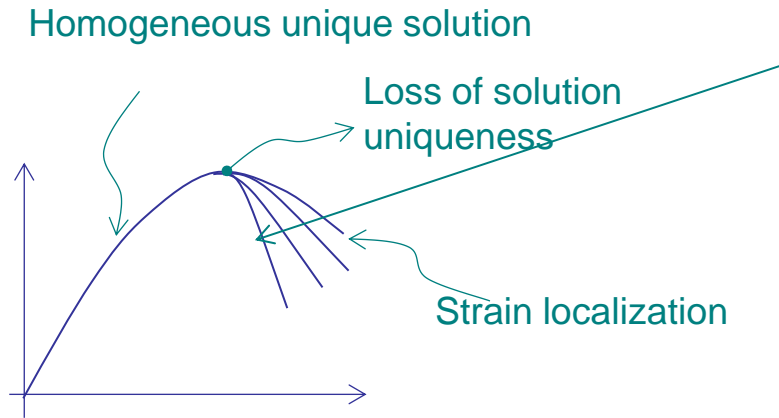
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 - Semi-analytical model
 - Predict composite meso-scale response
 - From components material models
- Advantages
 - Computationally efficient
 - Easy to integrate in a FE code (material model)
- Drawbacks
 - Difficult to formulate in an accurate way
 - Geometry complexity
 - Material behaviours complexity



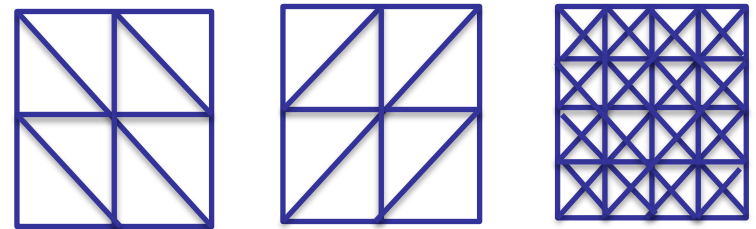
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Strain softening of the microscopic response

- Finite element solutions for strain softening problems suffer from:
 - Loss of solution uniqueness and strain localization
 - Mesh dependence



The numerical results change with the size of mesh and direction of mesh

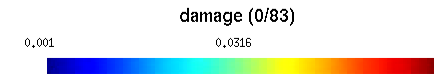
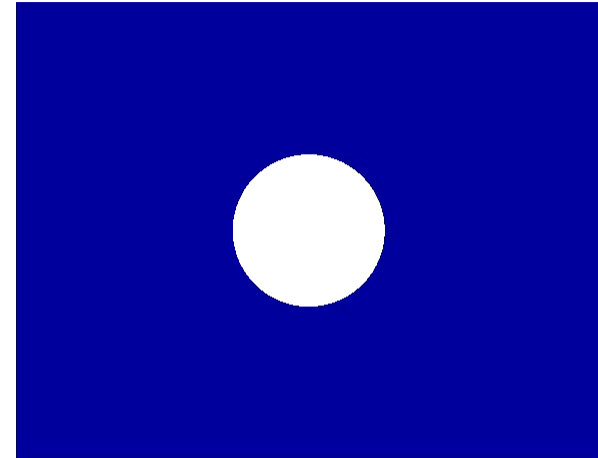


The numerical results change without convergence

- Requires a non-local formulation of the macro-scale problem

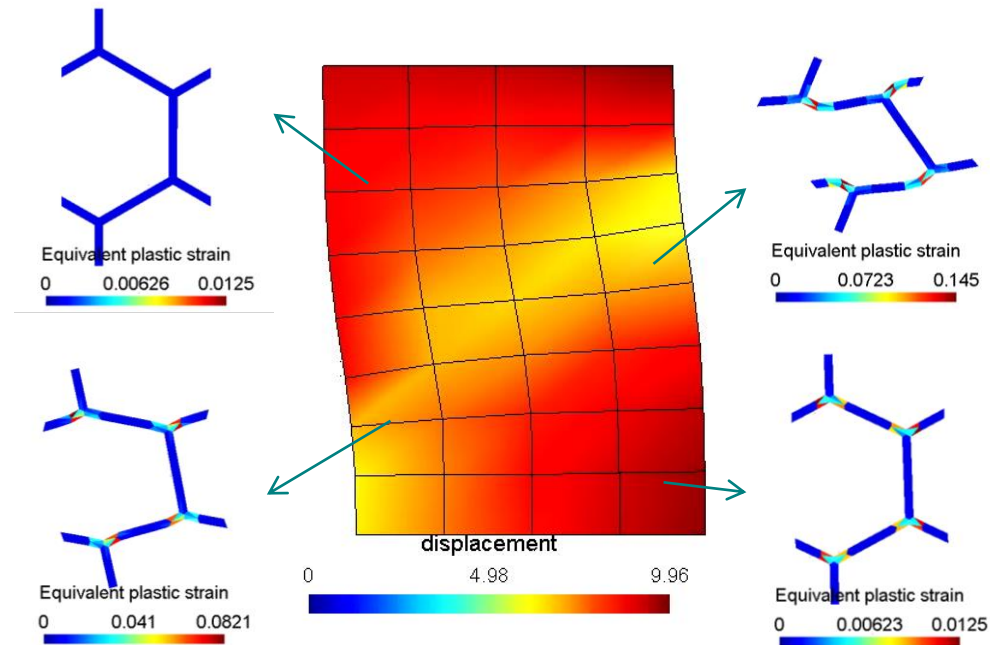
Multi-scale simulations with strain softening

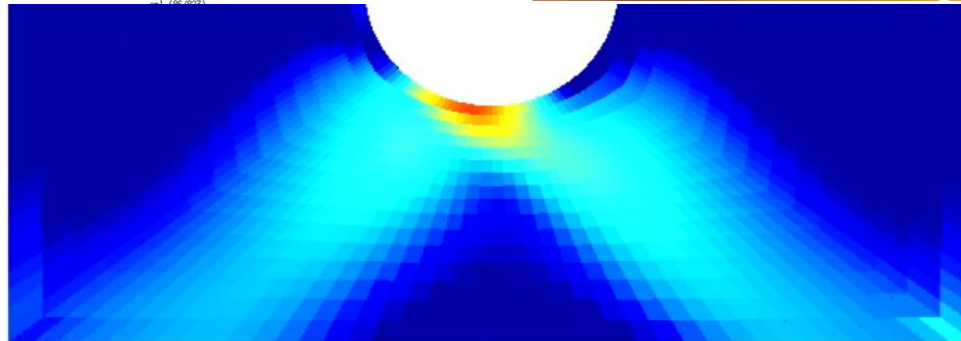
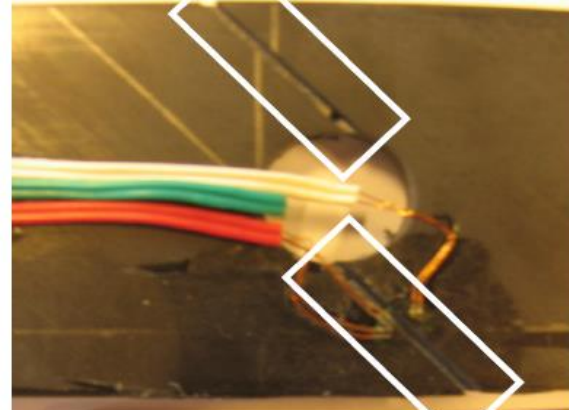
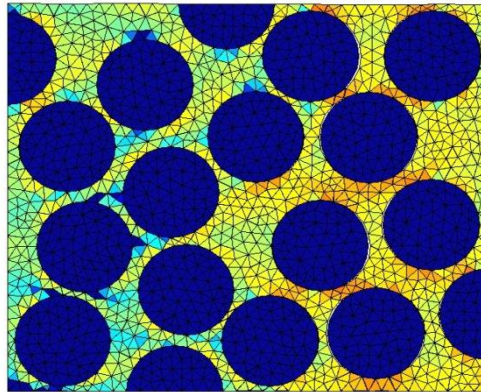
- Two cases considered
 - Composite materials
 - Mean-field homogenization
 - Non-local damage formulation



Y
Z

- Honeycomb structures
 - Computational homogenization
 - Second-order FE2
 - Micro-buckling





Non-local damage-enhanced mean-field-homogenization

L. Wu (ULg), L. Noels (ULg), L. Adam (e-Xstream), I. Doghri (UCL)

SIMUCOMP The research has been funded by the Walloon Region under the agreement no 1017232 (CT-EUC 2010-10-12) in the context of the ERA-NET +, Matera + framework.

- Semi analytical Mean-Field Homogenization

- Based on the averaging of the fields

$$\langle a \rangle = \frac{1}{V} \int_V a(\mathbf{X}) dV$$

- Meso-response

- From the volume ratios ($v_0 + v_I = 1$)

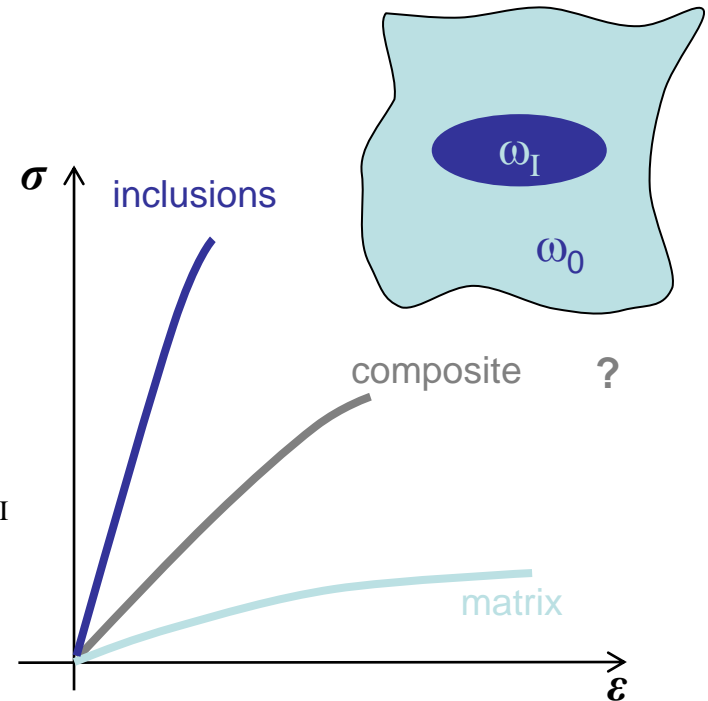
$$\begin{cases} \bar{\sigma} = \langle \sigma \rangle = v_0 \langle \sigma \rangle_{\omega_0} + v_I \langle \sigma \rangle_{\omega_I} = v_0 \sigma_0 + v_I \sigma_I \\ \bar{\varepsilon} = \langle \varepsilon \rangle = v_0 \langle \varepsilon \rangle_{\omega_0} + v_I \langle \varepsilon \rangle_{\omega_I} = v_0 \varepsilon_0 + v_I \varepsilon_I \end{cases}$$

- One more equation required

$$\varepsilon_I = \mathbf{B}^\varepsilon : \varepsilon_0$$

- Difficulty: find the adequate relations

$$\begin{cases} \sigma_I = f(\varepsilon_I) \\ \sigma_0 = f(\varepsilon_0) \\ \varepsilon_I = \mathbf{B}^\varepsilon : \varepsilon_0 \end{cases} \quad \mathbf{B}^\varepsilon ?$$



Non-local damage-enhanced mean-field-homogenization

• Mean-Field Homogenization for different materials

– Linear materials

• Materials behaviours

$$\begin{cases} \sigma_I = \bar{C}_I : \varepsilon_I \\ \sigma_0 = \bar{C}_0 : \varepsilon_0 \end{cases}$$

• Mori-Tanaka assumption $\varepsilon^\infty = \varepsilon_0$

• Use Eshelby tensor

$$\varepsilon_I = B^\varepsilon(I, \bar{C}_0, \bar{C}_I) : \varepsilon_0$$

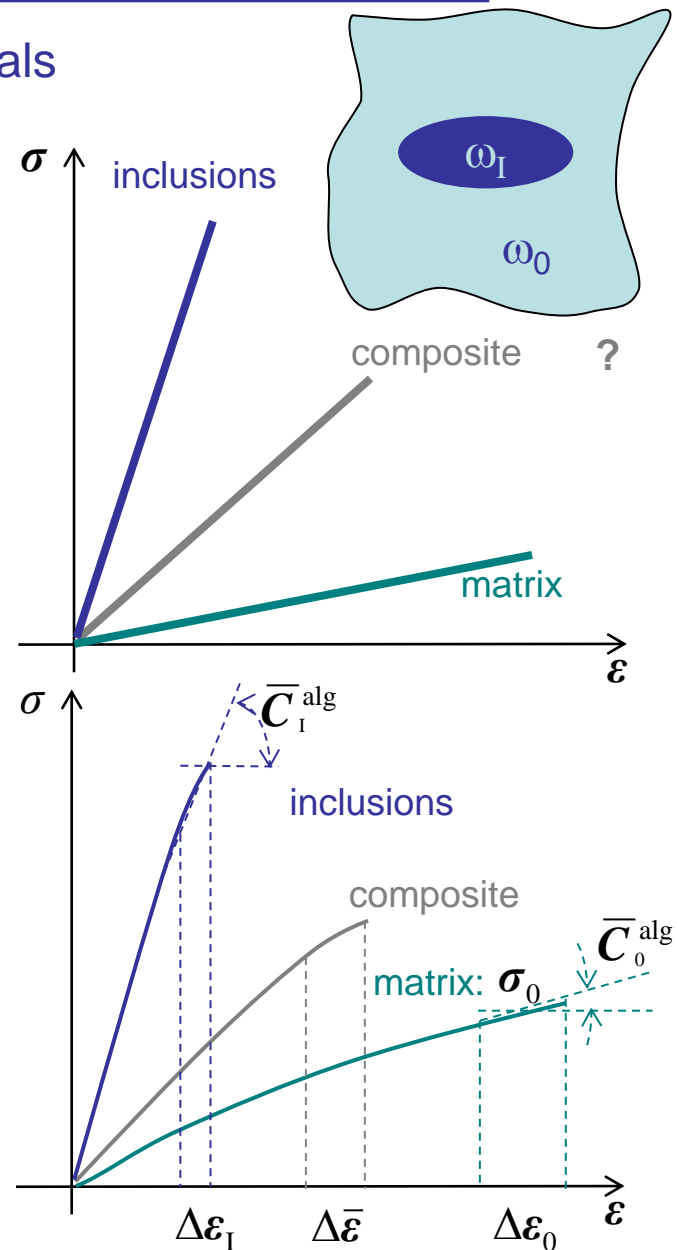
$$\text{with } B^\varepsilon = [I + S : \bar{C}_0^{-1} : (\bar{C}_I - \bar{C}_0)]^{-1}$$

– Non-linear materials

• Define a Linear Comparison Composite

• Common approach: incremental tangent

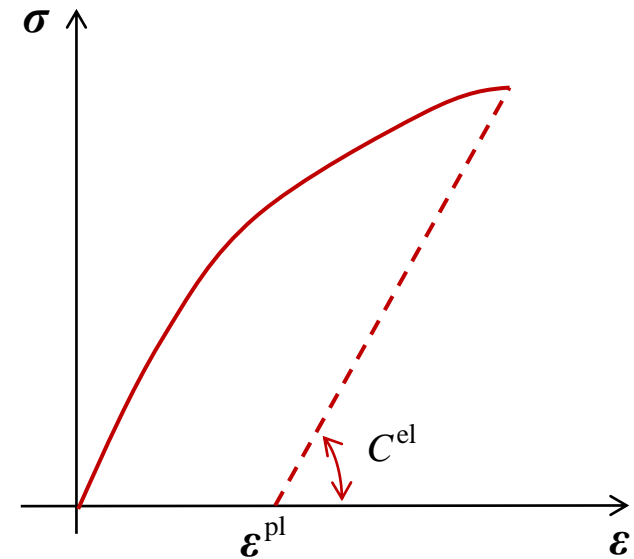
$$\Delta \varepsilon_I = B^\varepsilon(I, \bar{C}_0^{\text{alg}}, \bar{C}_I^{\text{alg}}) : \Delta \varepsilon_0$$



- Material models

- Elasto-plastic material

- Stress tensor $\boldsymbol{\sigma} = \mathbf{C}^{\text{el}} : (\boldsymbol{\varepsilon} - \boldsymbol{\varepsilon}^{\text{pl}})$
- Yield surface $f(\boldsymbol{\sigma}, p) = \boldsymbol{\sigma}^{\text{eq}} - \sigma^Y - R(p) \leq 0$
- Plastic flow $\Delta \boldsymbol{\varepsilon}^{\text{pl}} = \Delta p \mathbf{N} \quad \& \quad \mathbf{N} = \frac{\partial f}{\partial \boldsymbol{\sigma}}$
- Linearization $\delta \boldsymbol{\sigma} = \mathbf{C}^{\text{alg}} : \delta \boldsymbol{\varepsilon}$



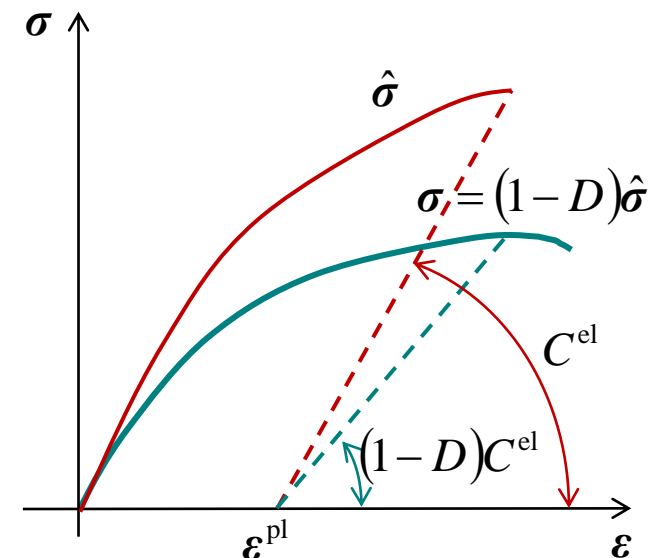
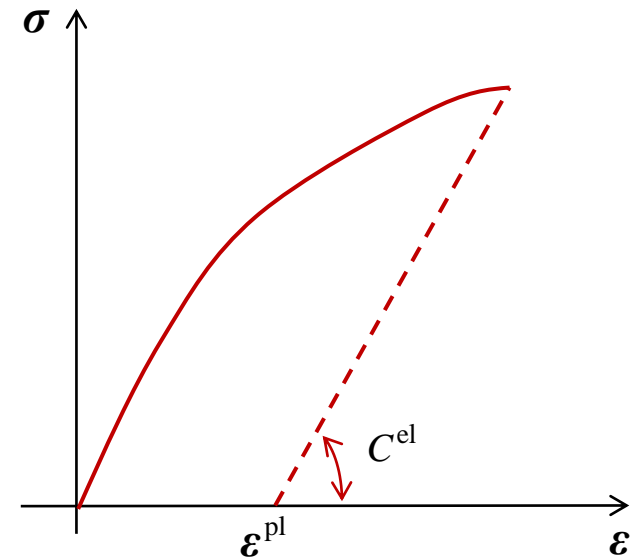
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- Local damage model

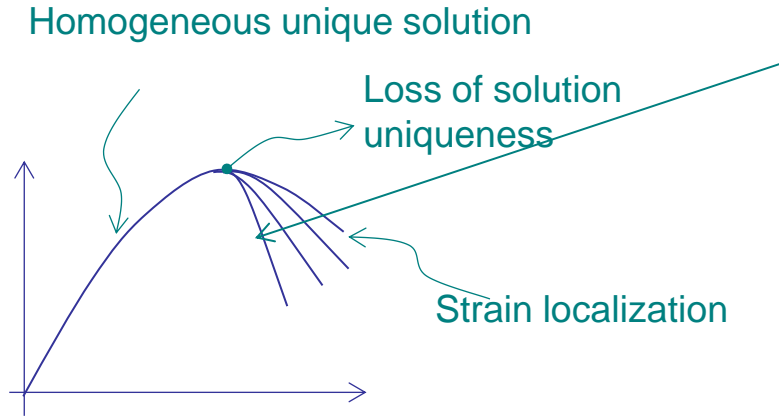
- Apparent-effective stress tensors $\boldsymbol{\sigma} = (1 - D) \hat{\boldsymbol{\sigma}}$
 - Plastic flow in the effective stress space
 - Damage evolution $\Delta D = F_D(\boldsymbol{\varepsilon}, \Delta p)$



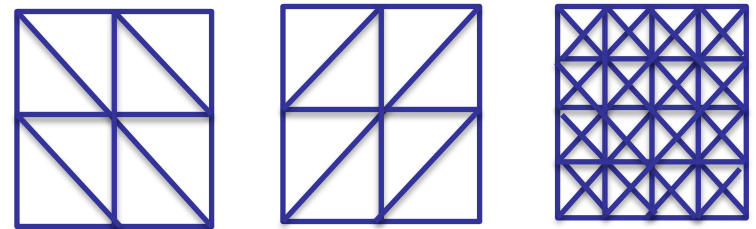
Non-local damage-enhanced mean-field-homogenization

- Finite element solutions for strain softening problems suffer from:

- Loss of solution uniqueness and strain localization
- Mesh dependence



The numerical results change with the size of mesh and direction of mesh



The numerical results change without convergence

- Implicit non-local approach [Peerlings et al 96, Geers et al 97, ...]

- A state variable is replaced by a non-local value reflecting the interaction between neighboring material points

$$\tilde{a}(\mathbf{x}) = \frac{1}{V_c} \int_{V_c} a(\mathbf{y}) w(\mathbf{y}; \mathbf{x}) dV$$

- Use Green functions as weight $w(\mathbf{y}; \mathbf{x})$

$$\Rightarrow \tilde{a} - c \nabla^2 \tilde{a} = a \Rightarrow \text{New degrees of freedom}$$

Non-local damage-enhanced mean-field-homogenization

Material models

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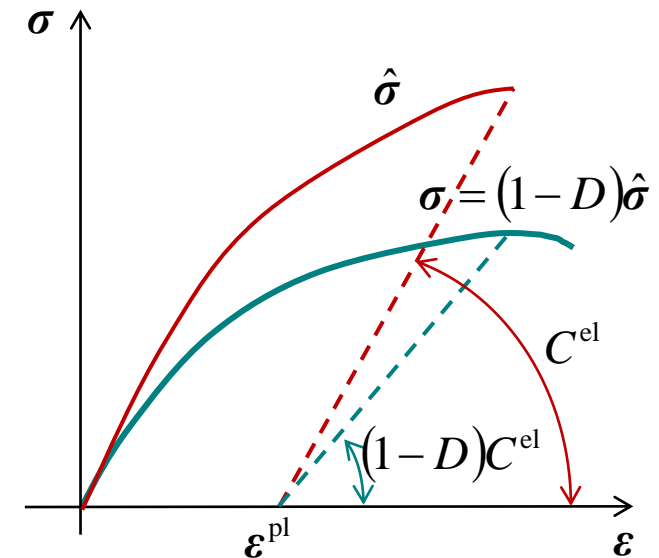
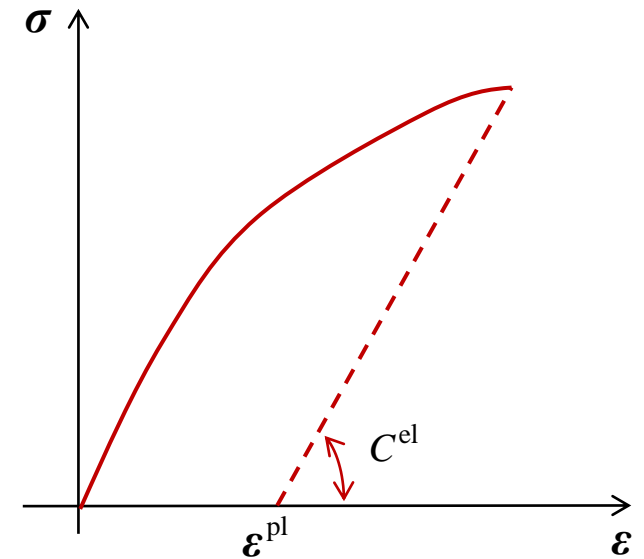
– Local damage model

- Apparent-effective stress tensors $\boldsymbol{\sigma} = (1 - D) \hat{\boldsymbol{\sigma}}$
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– Non-Local damage model

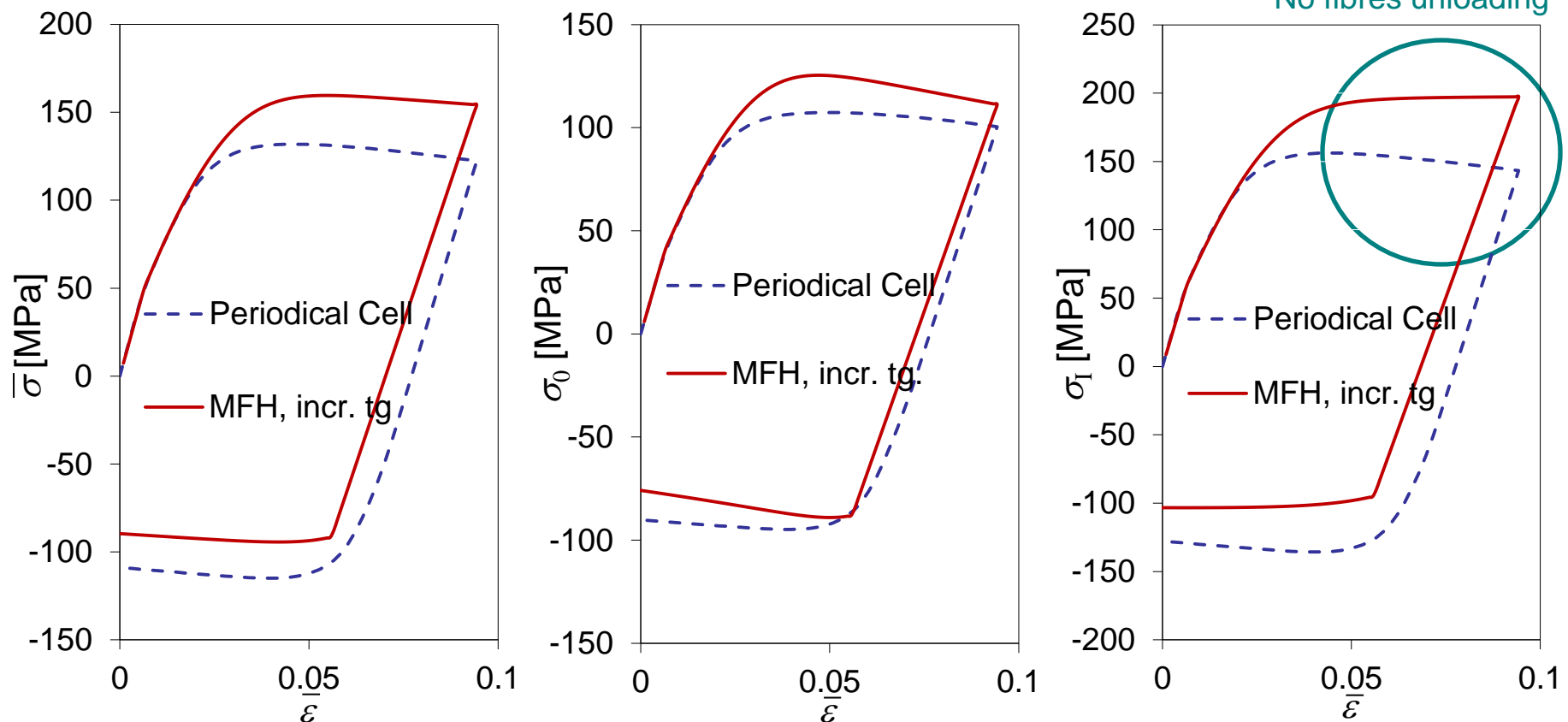
- Damage evolution $\Delta D = F_D(\boldsymbol{\varepsilon}, \Delta \tilde{p})$
- Anisotropic governing equation $\tilde{p} - \nabla \cdot (\mathbf{c}_g \cdot \nabla \tilde{p}) = p$
- Linearization

$$\delta \boldsymbol{\sigma} = \left[(1 - D) \mathbf{C}^{\text{alg}} - \hat{\boldsymbol{\sigma}} \otimes \frac{\partial F_D}{\partial \boldsymbol{\varepsilon}} \right] : \delta \boldsymbol{\varepsilon} - \hat{\boldsymbol{\sigma}} \frac{\partial F_D}{\partial \tilde{p}} \delta \tilde{p}$$



Non-local damage-enhanced mean-field-homogenization

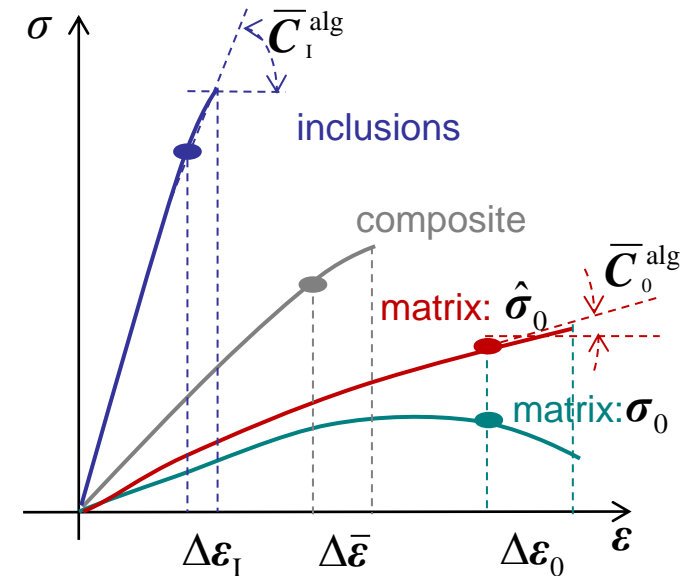
- Limitation of the incremental tangent method
 - Fictitious composite material
 - 50%-UD fibres
 - Elasto-plastic matrix with damage
 - Due to the incremental formalism, stress in fibres cannot decrease during loading



• Problem

- We want the fibres to get unloaded during the matrix damaging process
 - For the incremental-tangent approach

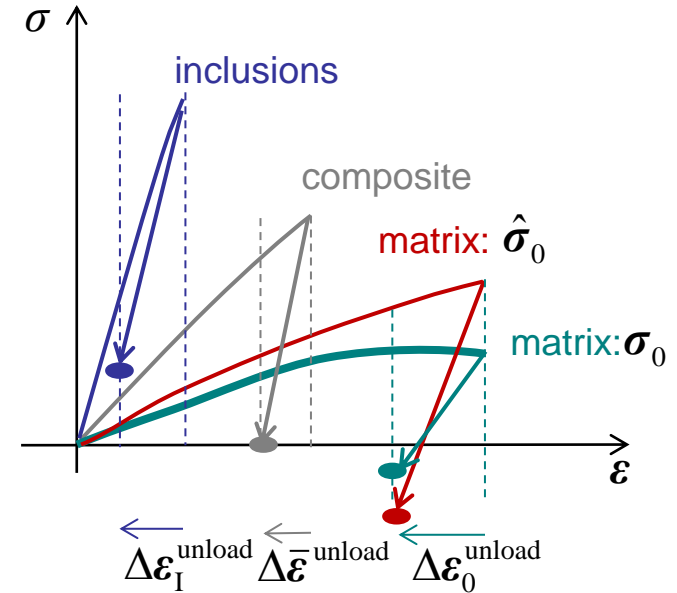
$$\Delta \boldsymbol{\varepsilon}_I = \mathbf{B}^\varepsilon \left(\mathbf{I}, (1-D) \bar{\mathbf{C}}_0^{\text{alg}}, \bar{\mathbf{C}}_I^{\text{alg}} \right) : \Delta \boldsymbol{\varepsilon}_0$$
 - To unload the fibres ($\boldsymbol{\varepsilon}_I < 0$) with such approach would require $\bar{\mathbf{C}}_I^{\text{alg}} < 0$
 - We cannot use the incremental tangent MFH
- We need to define the LCC from another stress state



Non-local damage-enhanced mean-field-homogenization

- Idea

- New incremental-secant approach
 - Perform a virtual elastic unloading from previous solution
 - Composite material unloaded to reach the stress-free state
 - Residual stress in components



• Idea

- New incremental-secant approach
 - Perform a virtual elastic unloading from previous solution
 - Composite material unloaded to reach the stress-free state
 - Residual stress in components

- Apply MFH from unloaded state
 - New strain increments (>0)

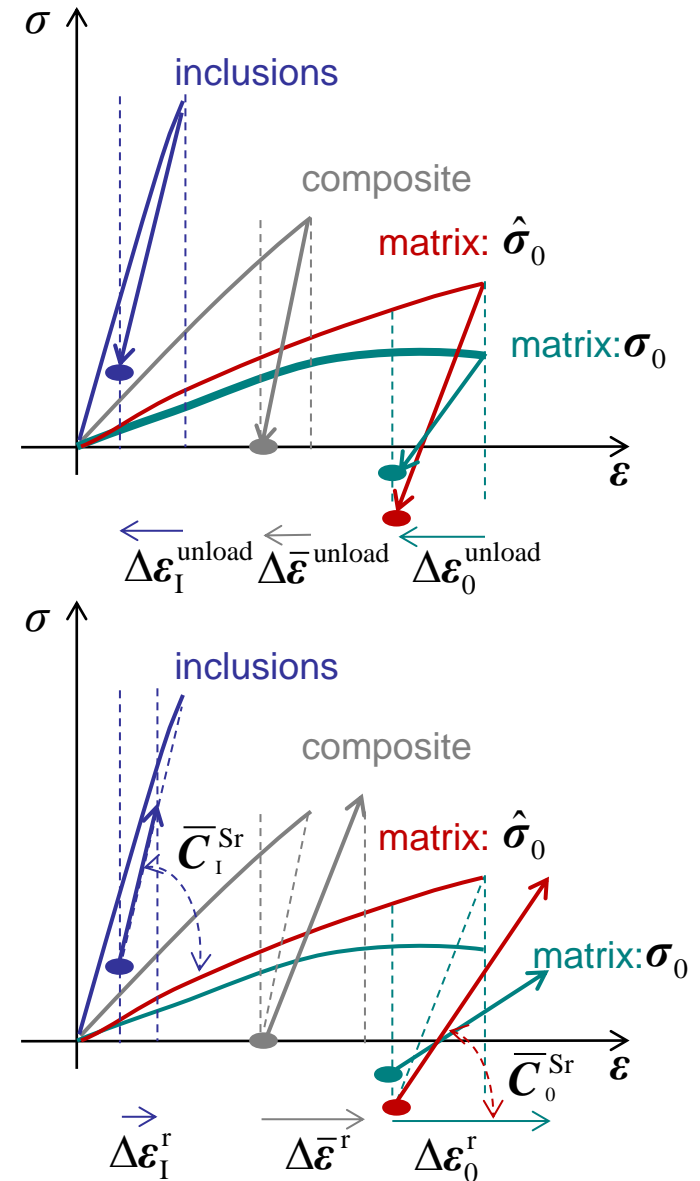
$$\Delta \boldsymbol{\varepsilon}_{I/0}^r = \Delta \boldsymbol{\varepsilon}_{I/0} + \Delta \boldsymbol{\varepsilon}_{I/0}^{\text{unload}}$$

- Use of secant operators

$$\Delta \boldsymbol{\varepsilon}_I^r = \mathbf{B}^\varepsilon \left(\mathbf{I}, (1-D) \bar{\mathbf{C}}_0^{\text{Sr}}, \bar{\mathbf{C}}_I^{\text{Sr}} \right) : \Delta \boldsymbol{\varepsilon}_0^r$$

- Possibility of have unloading

$$\begin{cases} \Delta \boldsymbol{\varepsilon}_I^r > 0 \\ \Delta \boldsymbol{\varepsilon}_I < 0 \end{cases}$$

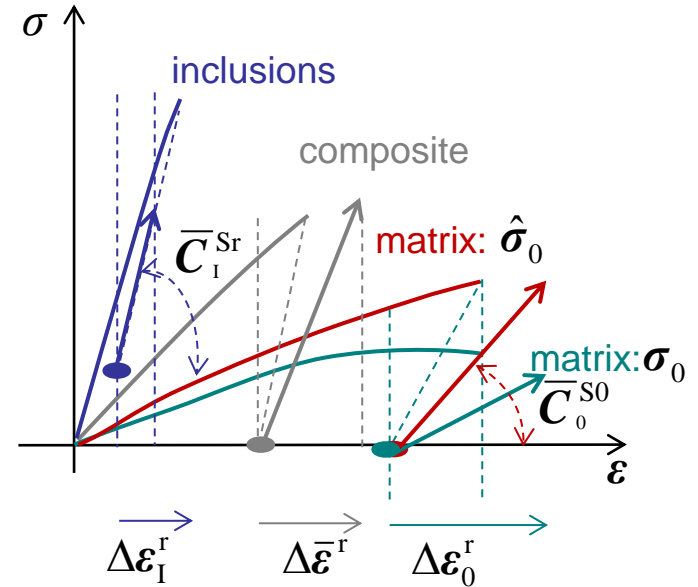


- Zero incremental-secant approach
 - For soft matrix response
 - Remove residual stress in matrix
 - Avoid adding spurious internal energy
 - Solve iteratively the system

$$\left\{ \begin{array}{l} \Delta \bar{\boldsymbol{\varepsilon}}^{(r)} = v_0 \Delta \boldsymbol{\varepsilon}_0^{(r)} + v_I \Delta \boldsymbol{\varepsilon}_I^{(r)} \\ \Delta \boldsymbol{\varepsilon}_I^r = \Delta \boldsymbol{\varepsilon}_I + \Delta \boldsymbol{\varepsilon}_I^{\text{unload}} \\ \Delta \boldsymbol{\varepsilon}_0^r = \Delta \boldsymbol{\varepsilon}_0 + \Delta \boldsymbol{\varepsilon}_0^{\text{unload}} \\ \Delta \boldsymbol{\varepsilon}_I^r = \mathbf{B}^\varepsilon \left(\mathbf{I}, (1-D) \bar{\mathbf{C}}_0^{S0}, \bar{\mathbf{C}}_I^{Sr} \right) : \Delta \boldsymbol{\varepsilon}_0^r \end{array} \right.$$

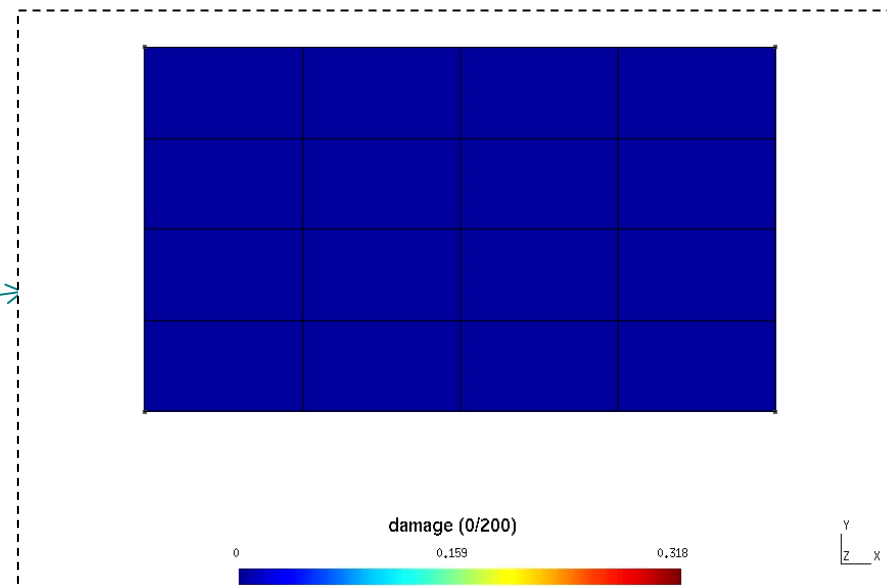
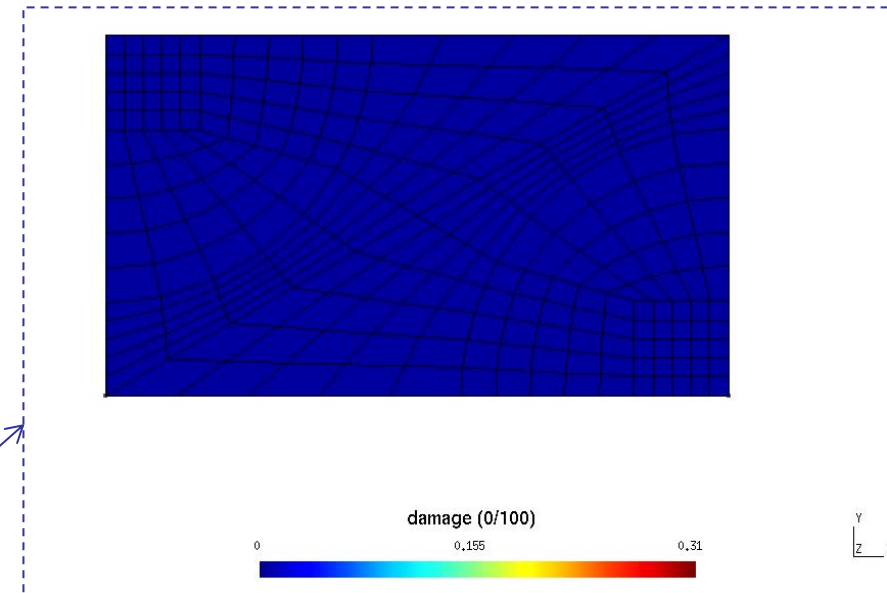
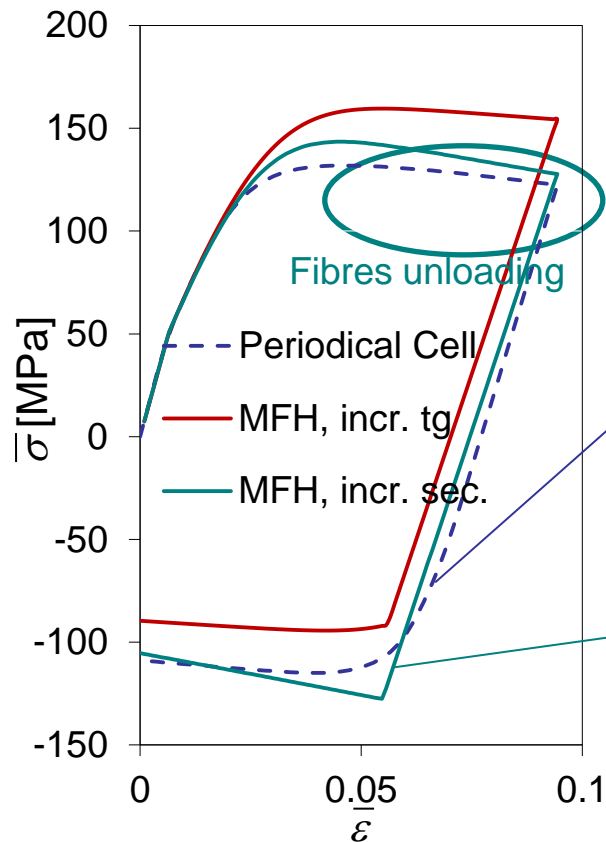
- With the stress tensors

$$\left\{ \begin{array}{l} \bar{\boldsymbol{\sigma}} = v_0 \boldsymbol{\sigma}_0 + v_I \boldsymbol{\sigma}_I \\ \boldsymbol{\sigma}_I = \boldsymbol{\sigma}_I^{\text{res}} + \bar{\mathbf{C}}_I^{Sr} : \Delta \boldsymbol{\varepsilon}_I^r \\ \boldsymbol{\sigma}_0 = (1-D) \bar{\mathbf{C}}_0^{S0} : \Delta \boldsymbol{\varepsilon}_0^r \end{array} \right.$$

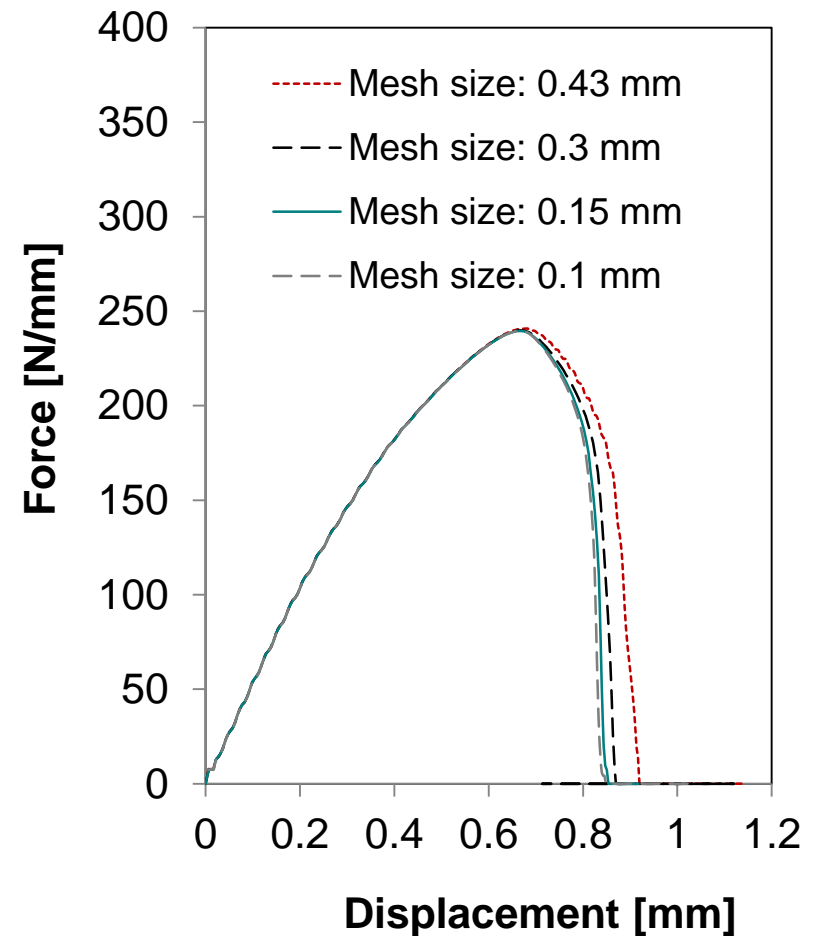
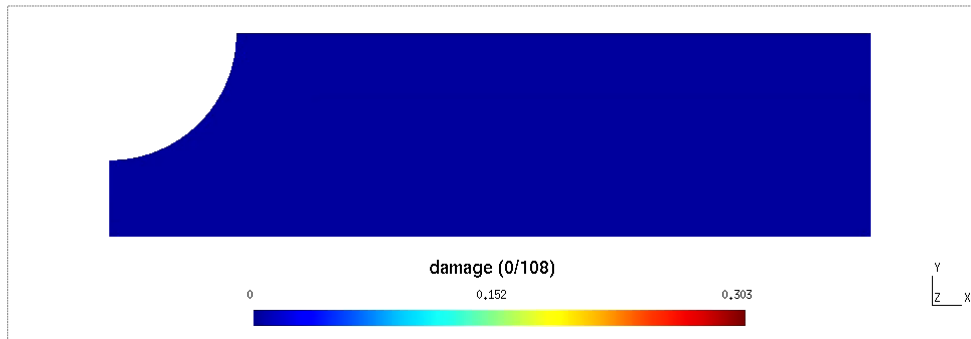


Non-local damage-enhanced mean-field-homogenization

- New results for damage
 - Fictitious composite material
 - 50%-UD fibres
 - Elasto-plastic matrix with damage

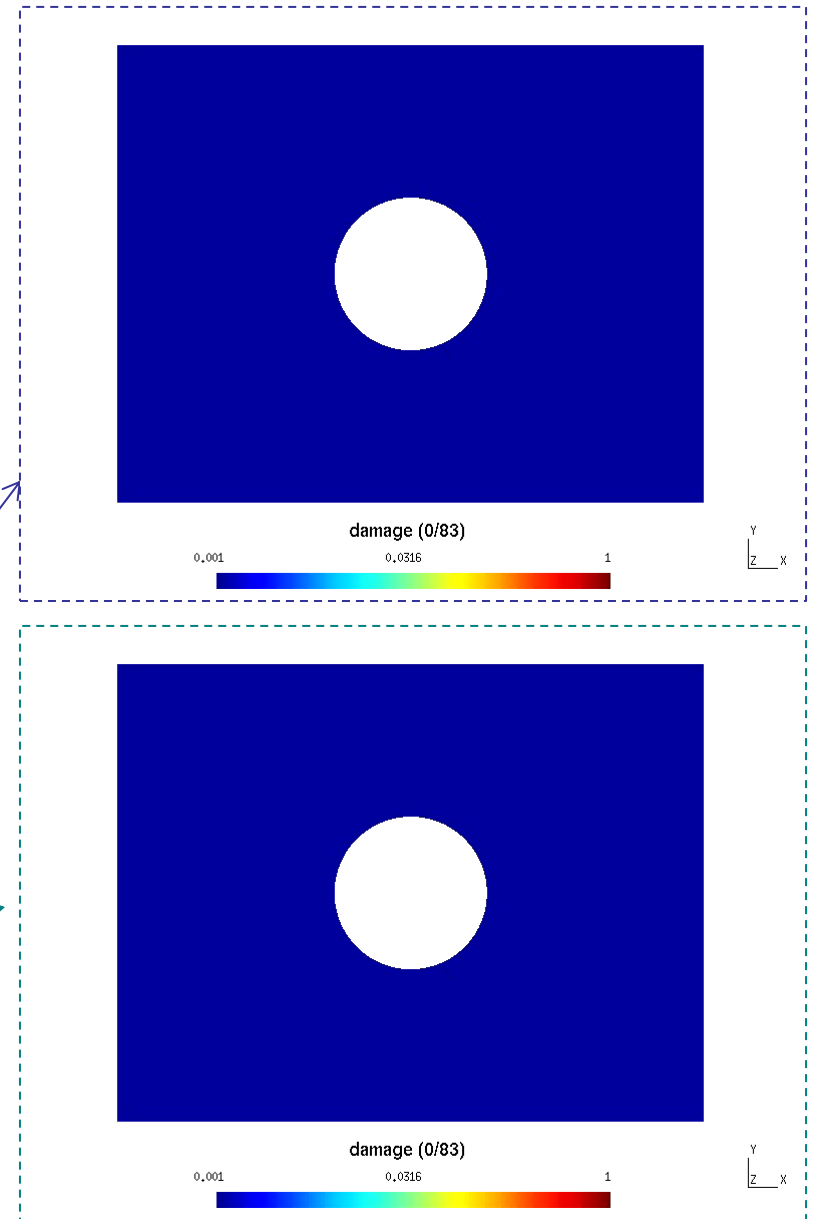
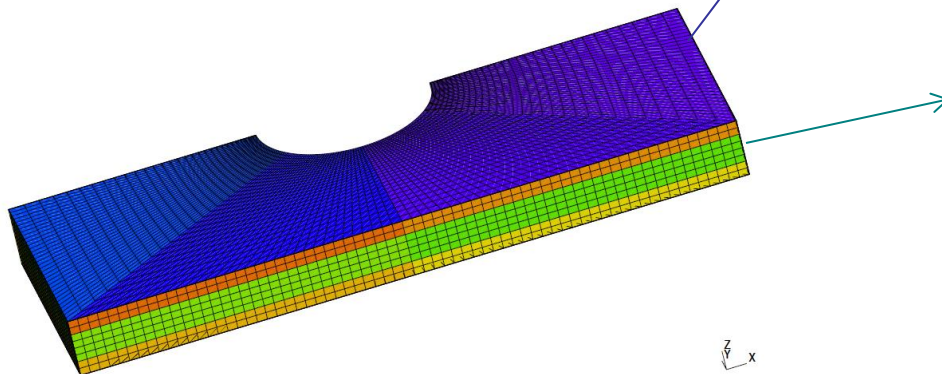
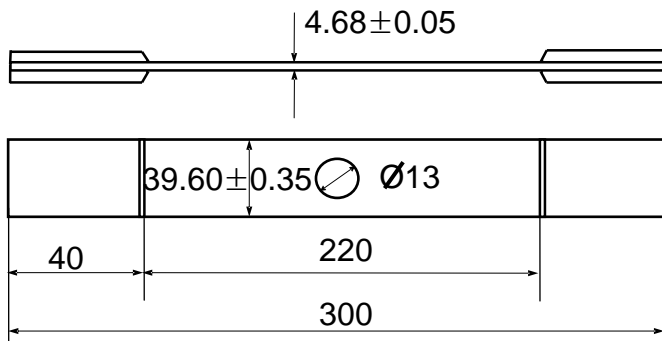


- Mesh-size effect
 - Fictitious composite material
 - 30%-UD fibres
 - Elasto-plastic matrix with damage
 - Notched ply



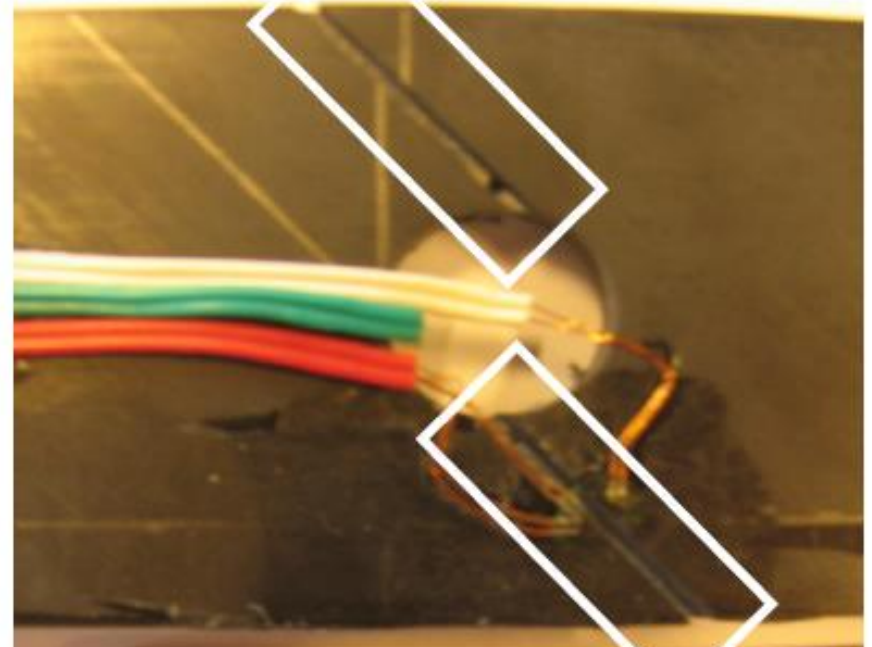
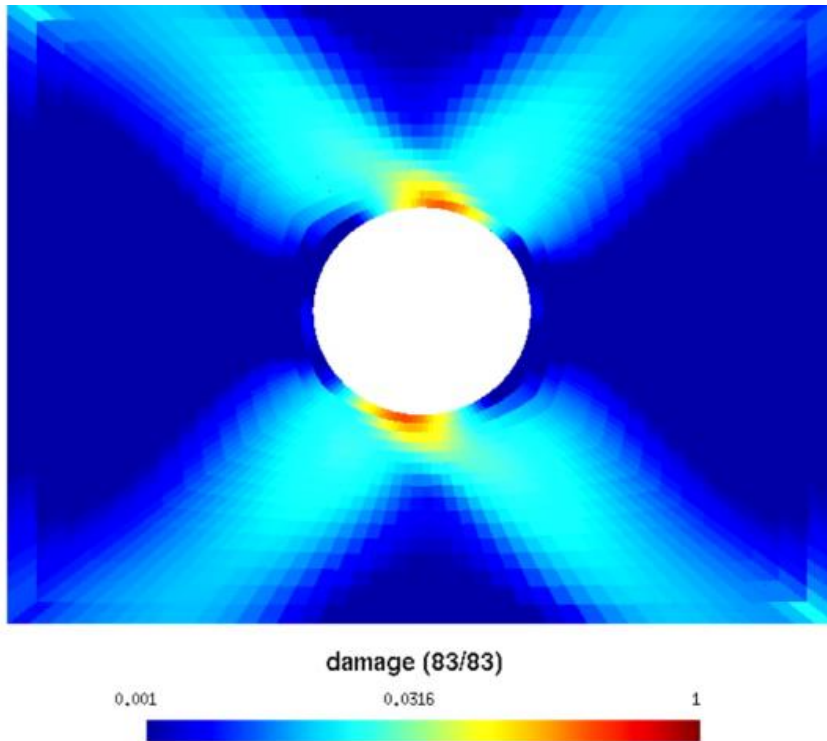
Non-local damage-enhanced mean-field-homogenization

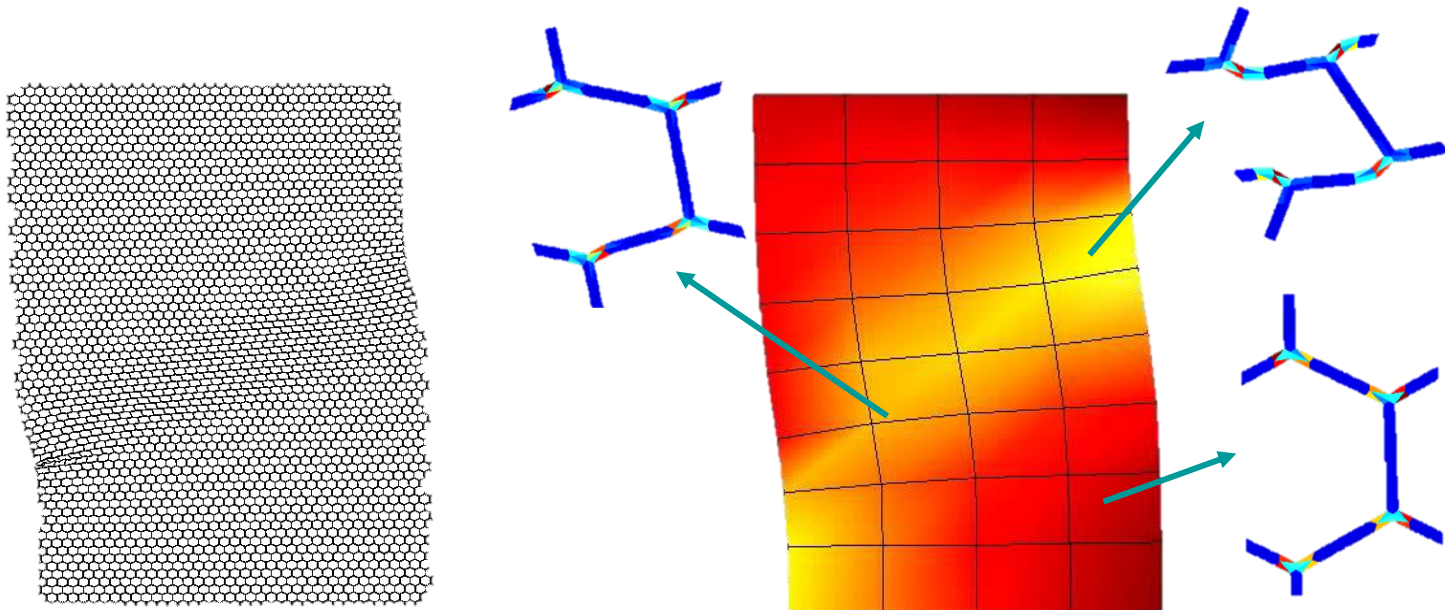
- Laminate plate with hole
 - Carbon-fibres reinforced epoxy
 - 60%-UD fibres
 - Elasto-plastic matrix with damage
 - $[-45_2/45_2]_S$ stacking sequence



Non-local damage-enhanced mean-field-homogenization

- Laminate plate with hole (2)
 - Carbon-fibres reinforced epoxy
 - 60%-UD fibres
 - Elasto-plastic matrix with damage
 - $[-45_2/45_2]_S$ stacking sequence





Computational homogenization for cellular materials

V.-D. Nguyen (ULg), L. Noels (ULg)

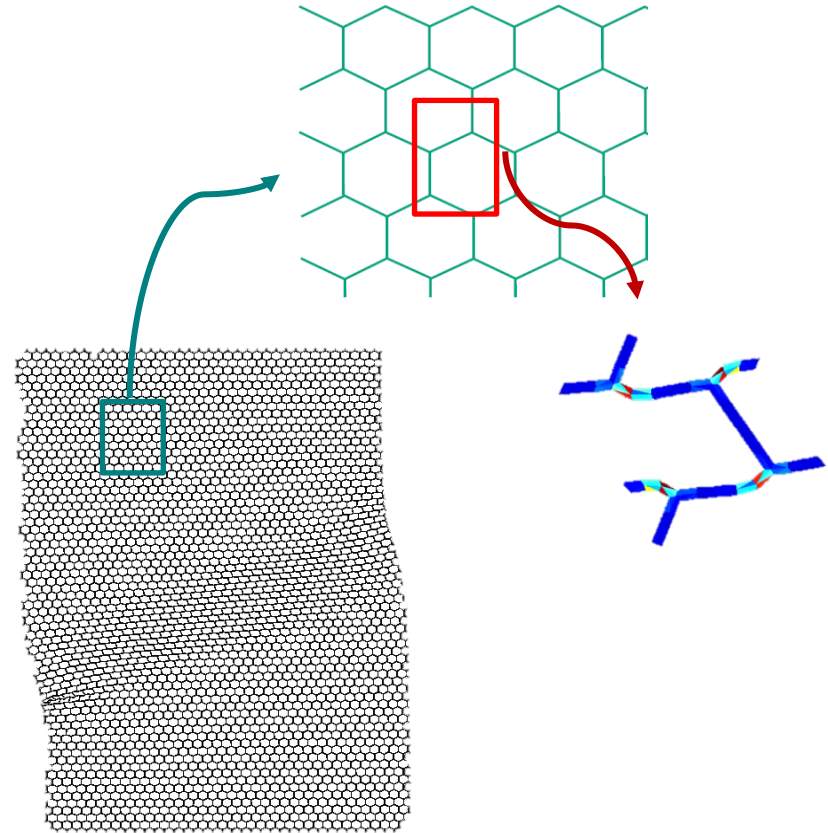
Les recherches ont été financées grâce à la subvention “Actions de recherche concertées ARC 09/14-02 BRIDGING- From imaging to geometrical modelling of complex micro structured materials: Bridging computational engineering and material science

- Challenges

- Micro-structure

- Not perfect with non periodic mesh

➡ How to constrain the periodic boundary conditions?



- Challenges

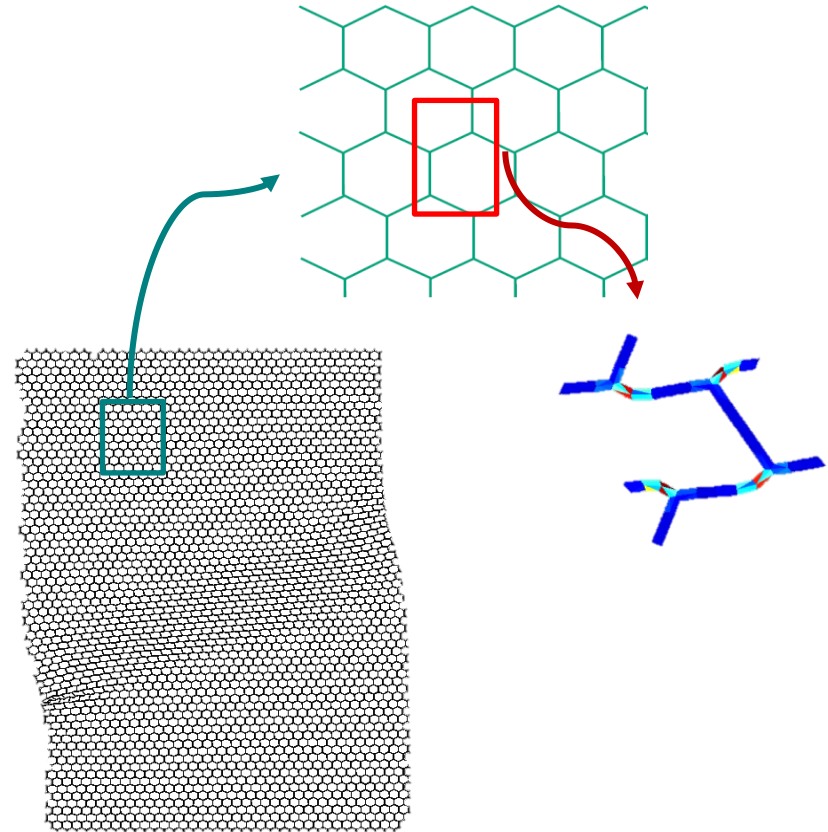
- Micro-structure

- Not perfect with non periodic mesh

➡ How to constrain the periodic boundary conditions?

- Thin components
 - Experiences micro-buckling

➡ How to capture the instability?



- Challenges

- Micro-structure

- Not perfect with non periodic mesh

➡ How to constrain the periodic boundary conditions?

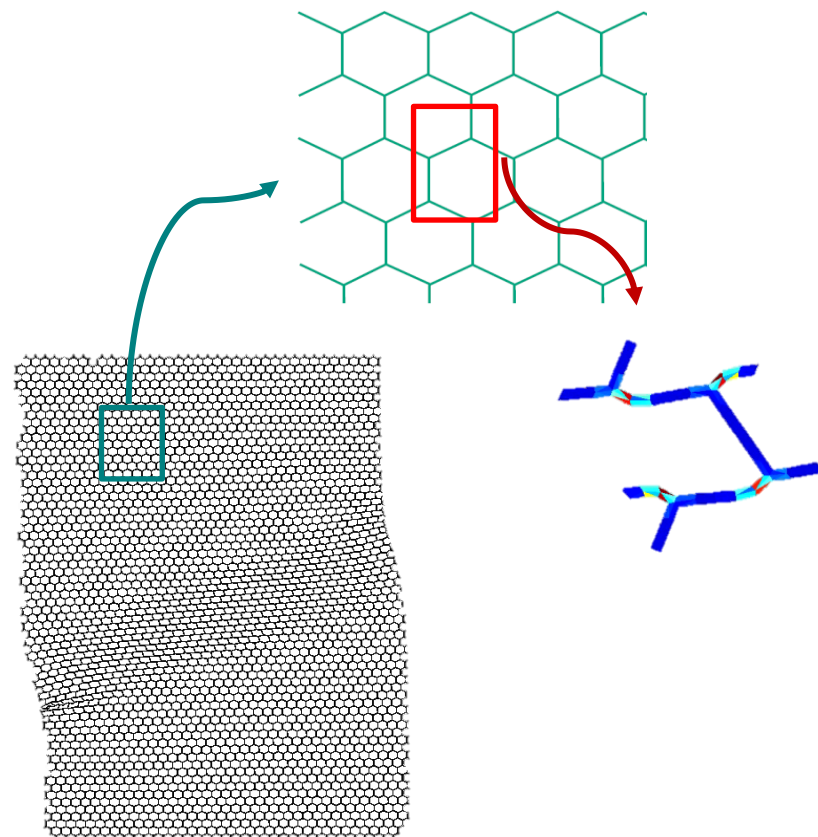
- Thin components
 - Experiences micro-buckling

➡ How to capture the instability?

- Transition

- Homogenized tangent not always elliptic
 - Localization bands

➡ How can we recover the solution unicity at the macro-scale?



- Challenges

- Micro-structure

- Not perfect with non periodic mesh

➡ How to constrain the periodic boundary conditions?

- Thin components
 - Experiences micro-buckling

➡ How to capture the instability?

- Transition

- Homogenized tangent not always elliptic
 - Localization bands

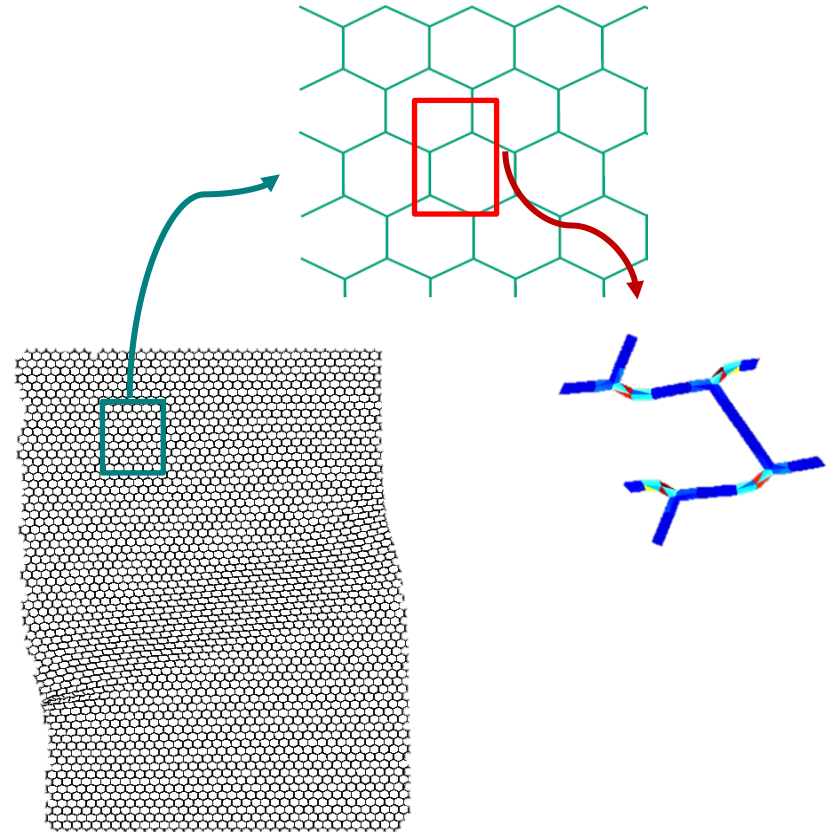
➡ How can we recover the solution unicity at the macro-scale?

- Macro-scale

- Localization bands

➡ How to remain computationally efficient

➡ How to capture the instability?



- Recover solution unicity: second-order FE²

- Macro-scale

- High-order Strain-Gradient formulation

$$\bar{\mathbf{P}}(\bar{\mathbf{X}}) \cdot \nabla_0 - \bar{\mathbf{Q}}(\bar{\mathbf{X}}): (\nabla_0 \otimes \nabla_0) = 0$$

- Partitioned mesh (//)

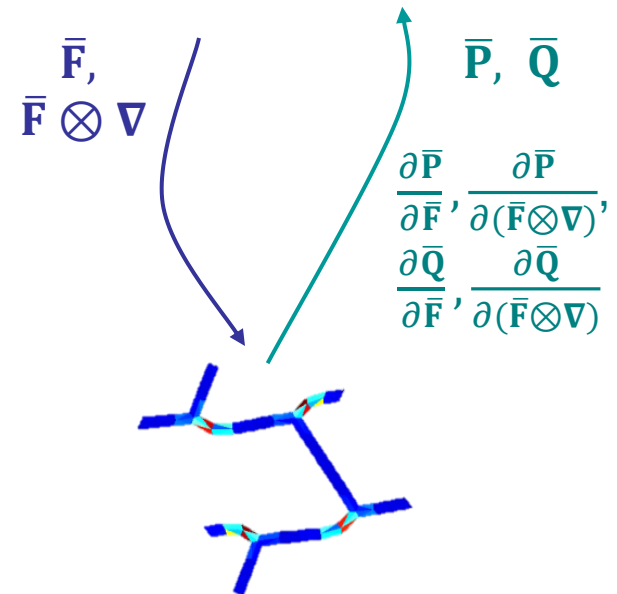
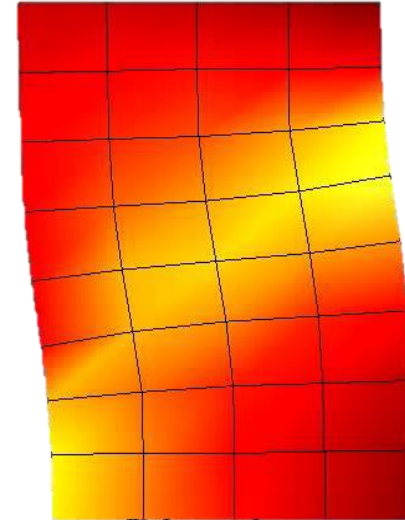
- Transition

- Gauss points on different processors
 - Each Gauss point is associated to one mesh and one solver

- Micro-scale

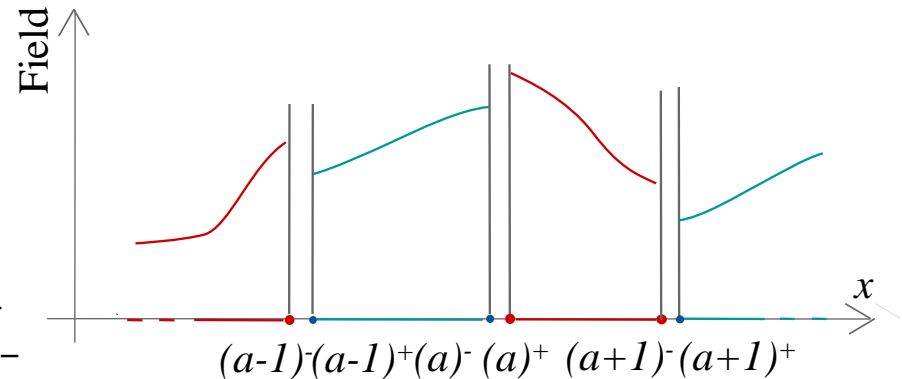
- Usual continuum

$$\mathbf{P}(\mathbf{X}) \cdot \nabla_0 = 0$$



Computational homogenization for foamed materials

- Discontinuous Galerkin (DG) implementation of the second order continuum
 - Finite-element discretization
 - Same **discontinuous** polynomial approximations for the
 - Test** functions φ_h and
 - Trial** functions $\delta\varphi$
 - Definition of operators on the interface trace:
 - Jump operator:**
$$[[\cdot]] = \frac{\cdot^+ - \cdot^-}{\cdot^+ + \cdot^-}$$
 - Mean operator:**
$$\langle \cdot \rangle = \frac{\cdot^+ + \cdot^-}{2}$$
 - Continuity is weakly enforced, such that the method
 - Is consistent
 - Is stable
 - Has the optimal convergence rate
 - Can be used to weakly enforce higher discontinuities



- Second-order FE2 method

- Macro-scale second order continuum

$$\bar{\mathbf{P}}(\bar{\mathbf{X}}) \cdot \nabla_0 - \bar{\mathbf{Q}}(\bar{\mathbf{X}}) : (\nabla_0 \otimes \nabla_0) = 0$$

- Requires \mathcal{C}^1 shape functions on the mesh
- The \mathcal{C}^1 can be weakly enforced using the DG method

$$a(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) = a^{\text{bulk}}(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) + a^{\text{PI}}(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) + a^{\text{QI}}(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) = b(\delta \bar{\mathbf{u}})$$

- Second-order FE2 method

- Macro-scale second order continuum

$$\bar{\mathbf{P}}(\bar{\mathbf{X}}) \cdot \nabla_0 - \bar{\mathbf{Q}}(\bar{\mathbf{X}}) : (\nabla_0 \otimes \nabla_0) = 0$$

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- Usual volume terms

$$a^{\text{bulk}}(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) = \int_{\bar{V}} [\bar{\mathbf{P}}(\bar{\mathbf{u}}) : (\delta \bar{\mathbf{u}} \otimes \nabla_0) + \bar{\mathbf{Q}}(\bar{\mathbf{X}}) : (\delta \bar{\mathbf{u}} \otimes \nabla_0 \otimes \nabla_0)] dV$$

- Second-order FE2 method

- Macro-scale second order continuum

$$\bar{\mathbf{P}}(\bar{\mathbf{X}}) \cdot \nabla_0 - \bar{\mathbf{Q}}(\bar{\mathbf{X}}) : (\nabla_0 \otimes \nabla_0) = 0$$

- Requires \mathcal{C}^1 shape functions on the mesh
- The \mathcal{C}^1 can be weakly enforced using the DG method

$$a(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) = a^{\text{bulk}}(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) + a^{\text{PI}}(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) + a^{\text{QI}}(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) = b(\delta \bar{\mathbf{u}})$$

- Weak enforcement of the \mathcal{C}^0

- Continuity
- Consistency
- Stability

between the finite elements

$$a^{\text{PI}}(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) = \int_{\partial_I \bar{V}} \left[\llbracket \delta \bar{\mathbf{u}} \rrbracket \cdot \langle \bar{\mathbf{P}} - \bar{\mathbf{Q}} \cdot \nabla_0 \rangle \cdot \bar{\mathbf{N}} + \llbracket \bar{\mathbf{u}} \rrbracket \cdot \langle \bar{\mathbf{P}}(\delta \bar{\mathbf{u}}) - \bar{\mathbf{Q}}(\delta \bar{\mathbf{u}}) \cdot \nabla_0 \rangle \cdot \bar{\mathbf{N}} + \llbracket \bar{\mathbf{u}} \rrbracket \otimes \bar{\mathbf{N}} : \left\langle \frac{\beta_P}{h_s} \mathbf{C}^0 \right\rangle : \llbracket \delta \bar{\mathbf{u}} \rrbracket \otimes \bar{\mathbf{N}} \right] dV$$

- Allows efficient parallelization as elements are disjoint

- Second-order FE2 method

- Macro-scale second order continuum

$$\bar{\mathbf{P}}(\bar{\mathbf{X}}) \cdot \nabla_0 - \bar{\mathbf{Q}}(\bar{\mathbf{X}}) : (\nabla_0 \otimes \nabla_0) = 0$$

- Requires \mathcal{C}^1 shape functions on the mesh
- The \mathcal{C}^1 can be weakly enforced using the DG method

$$a(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) = a^{\text{bulk}}(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) + a^{\text{PI}}(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) + a^{\text{QI}}(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) = b(\delta \bar{\mathbf{u}})$$

- Weak enforcement of the \mathcal{C}^1

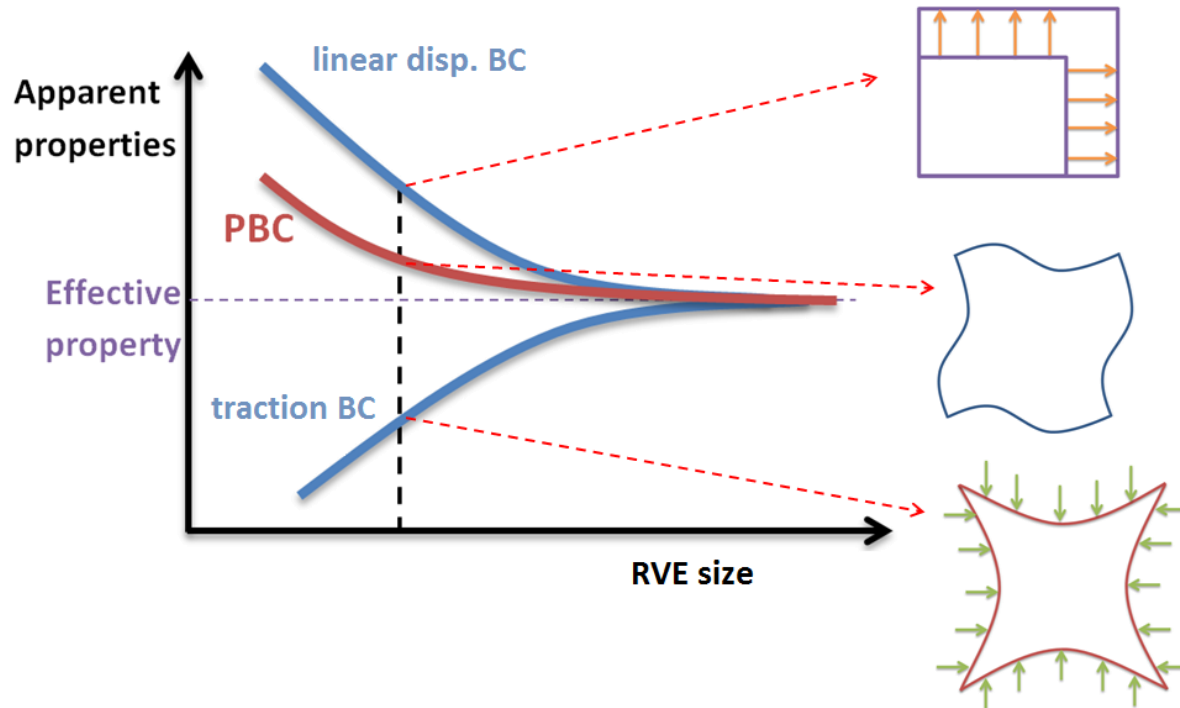
- Continuity
- Consistency
- Stability

between the finite elements

$$a^{\text{QI}}(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) = \int_{\partial_I \bar{V}} \left[\llbracket \delta \bar{\mathbf{u}} \otimes \nabla_0 \rrbracket \cdot \langle \bar{\mathbf{Q}} \rangle \cdot \bar{\mathbf{N}} + \llbracket \bar{\mathbf{u}} \otimes \nabla_0 \rrbracket \cdot \langle \bar{\mathbf{Q}}(\delta \bar{\mathbf{u}}) \rangle \cdot \bar{\mathbf{N}} + \llbracket \bar{\mathbf{u}} \otimes \nabla_0 \rrbracket \otimes \bar{\mathbf{N}} : \left\langle \frac{\beta_P}{h_s} \mathbf{J}^0 \right\rangle : \llbracket \delta \bar{\mathbf{u}} \otimes \nabla_0 \rrbracket \otimes \bar{\mathbf{N}} \right] dV$$

- Allows efficient parallelization as elements are disjoint

- Micro-scale periodic boundary conditions
 - Convergence in terms of RVE size



- Periodic boundary condition is the optimum choice for periodic structures
- Periodic boundary condition remains the optimum choice for non-periodic structures

- Micro-scale periodic boundary conditions (2)

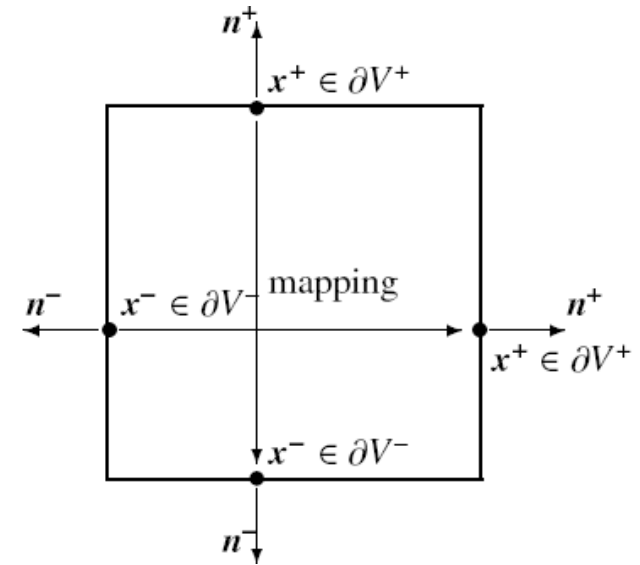
- Defined from the fluctuation field

$$\mathbf{w} = \mathbf{u} - (\bar{\mathbf{F}} - \mathbf{I}) \cdot \mathbf{X} + \frac{1}{2} (\bar{\mathbf{F}} \otimes \nabla_0) : (\mathbf{X} \otimes \mathbf{X})$$

- Stated on opposite RVE sizes

$$\begin{cases} \mathbf{w}(\mathbf{X}^+) = \mathbf{w}(\mathbf{X}^-) \\ \int_{\partial V^-} \mathbf{w}(\mathbf{X}) d\partial V = \mathbf{0} \end{cases}$$

- Can be achieved by constraining opposite nodes

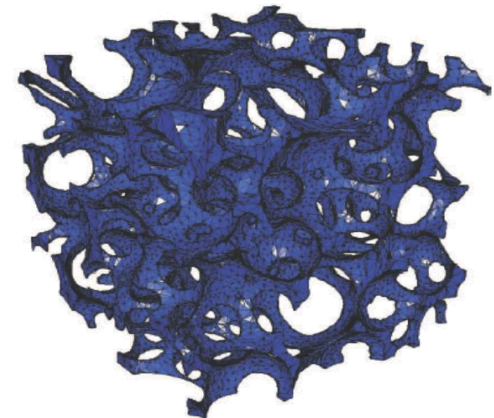


- Foamed materials

- Usually random meshes
- Important voids on the boundaries

- Honeycomb structures

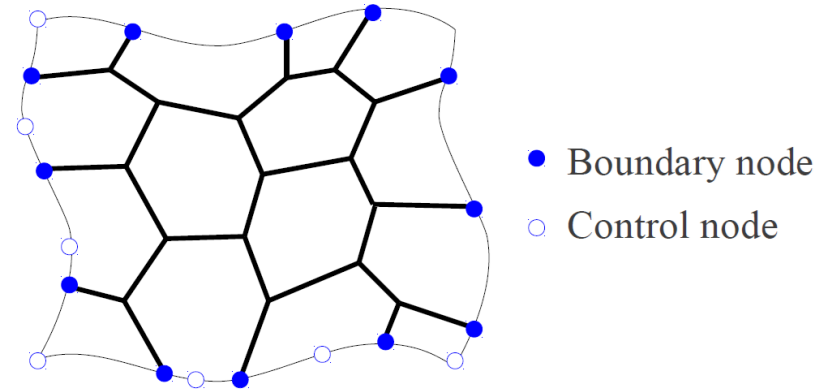
- Not periodic due to the imperfections



- Micro-scale periodic boundary conditions (2)

- New interpolant method

$$\left\{ \begin{array}{l} \mathbf{w}(\mathbf{X}^-) = \sum_k \mathbf{N}(\mathbf{X}) \mathbf{w}^k \\ \mathbf{w}(\mathbf{X}^+) = \sum_k \mathbf{N}(\mathbf{X}) \mathbf{w}^k \\ \int_{\partial V^-} \left(\sum_k \mathbf{N}(\mathbf{X}) \mathbf{w}^k \right) d\partial V = \mathbf{0} \end{array} \right.$$



- Use of Lagrange, cubic spline .. interpolations
- Fits for
 - Arbitrary meshes
 - Important voids on the RVE sides
- Results in new constraints in terms of the boundary and control nodes displacements

$$\tilde{\mathcal{C}} \tilde{\mathbf{u}}_b - \mathbf{g}(\bar{\mathbf{F}}, \bar{\mathbf{F}} \otimes \nabla_0) = 0$$

- Capturing instabilities

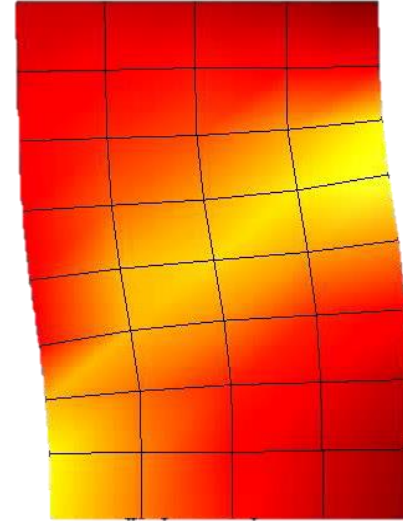
- Macro-scale: localization bands

- Path following method on the applied loading

$$a(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) = \bar{\mu} b(\delta \bar{\mathbf{u}})$$

- Arc-length constraint on the load increment

$$\bar{h}(\Delta \bar{\mathbf{u}}, \Delta \bar{\mu}) = \frac{\Delta \bar{\mathbf{u}} \cdot \Delta \bar{\mathbf{u}}}{\bar{X}_0^2} + \Delta \bar{\mu}^2 - \Delta L^2 = 0$$



- Capturing instabilities

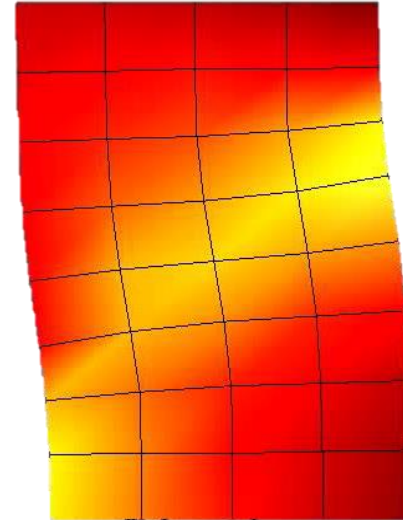
- Macro-scale: localization bands

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$$a(\bar{\mathbf{u}}, \delta \bar{\mathbf{u}}) = \bar{\mu} b(\delta \bar{\mathbf{u}})$$

- Arc-length constraint on the load increment

$$\bar{h}(\Delta \bar{\mathbf{u}}, \Delta \bar{\mu}) = \frac{\Delta \bar{\mathbf{u}} \cdot \Delta \bar{\mathbf{u}}}{\bar{X}_0^2} + \Delta \bar{\mu}^2 - \Delta L^2 = 0$$



- Micro-scale

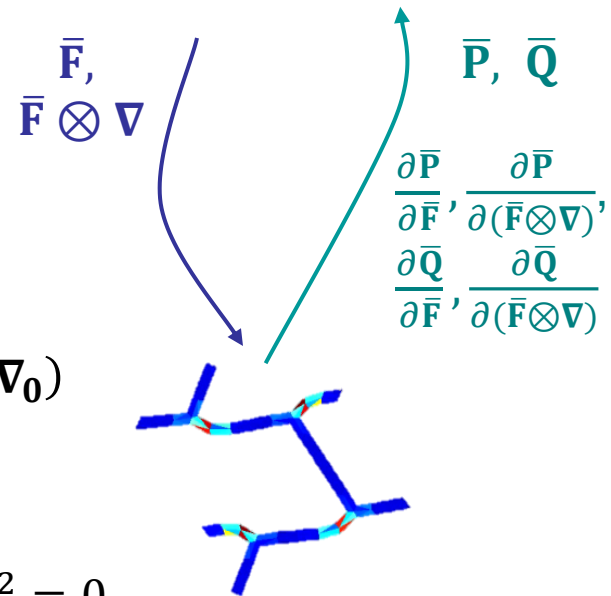
- Path following method on the applied boundary conditions

$$\tilde{\mathcal{C}} \tilde{\mathbf{u}}_b - \mathbf{g}(\bar{\mathbf{F}}, \bar{\mathbf{F}} \otimes \nabla_0) = 0$$

$$\begin{cases} \bar{\mathbf{F}} = \bar{\mathbf{F}}_0 + \mu \Delta \bar{\mathbf{F}} \\ \bar{\mathbf{F}} \otimes \nabla_0 = (\bar{\mathbf{F}} \otimes \nabla_0)_0 + \mu \Delta(\bar{\mathbf{F}} \otimes \nabla_0) \end{cases}$$

- Arc-length constraint on the load increment

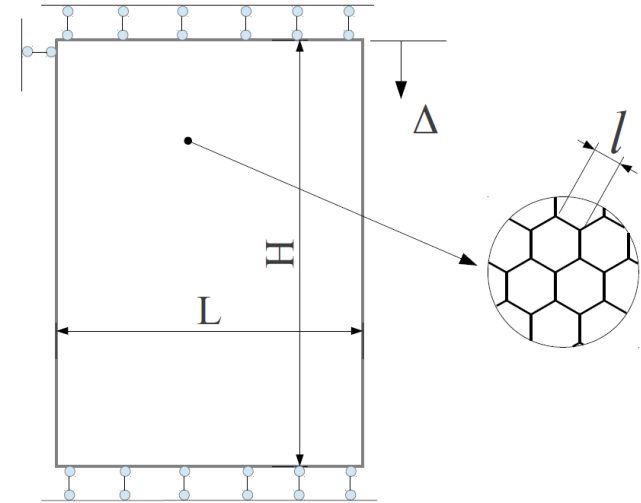
$$h(\Delta \mathbf{u}, \Delta \mu) = \frac{\Delta \mathbf{u} \cdot \Delta \mathbf{u}}{X_0^2} + \Delta \mu^2 - \Delta l^2 = 0$$



Computational homogenization for foamed materials

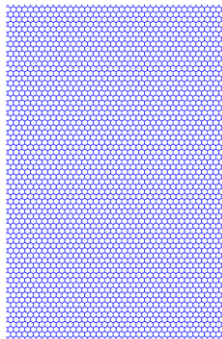
- Compression of an hexagonal honeycomb

- Elasto-plastic material

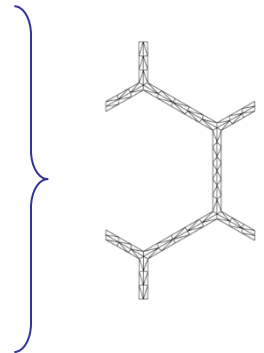
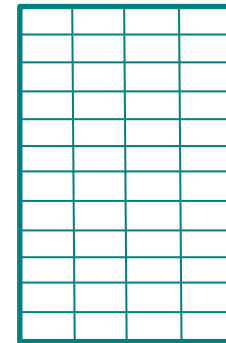
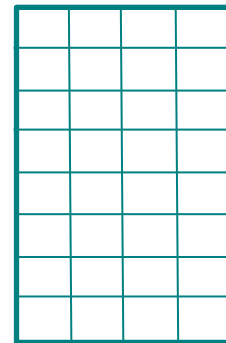
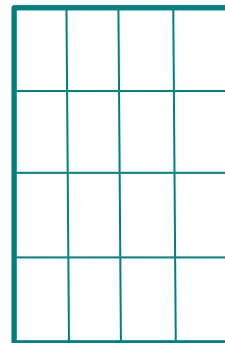


- Comparison of different solutions

Full direct simulation



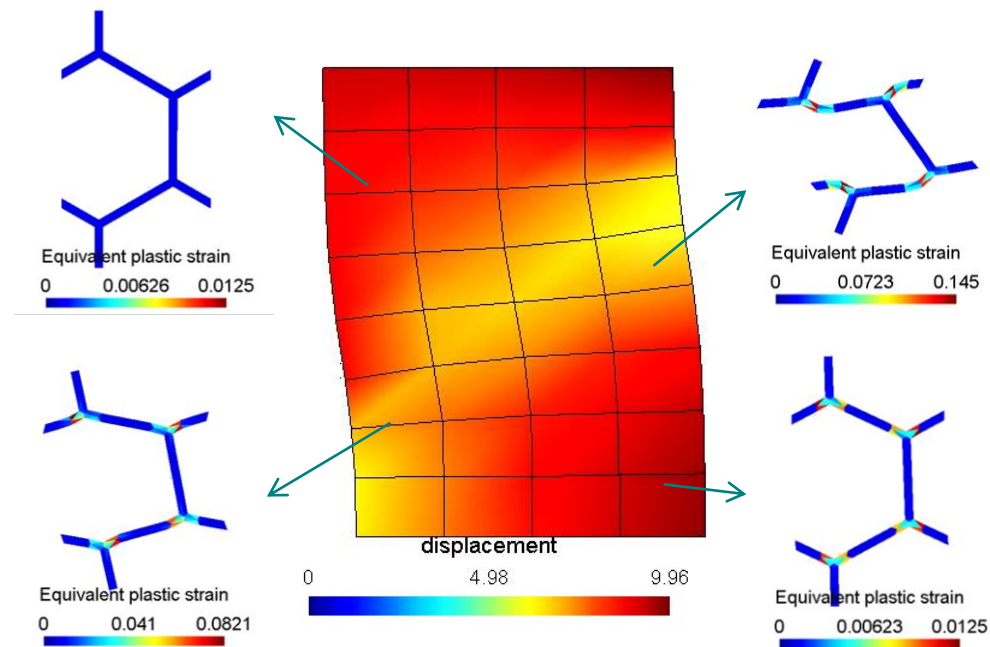
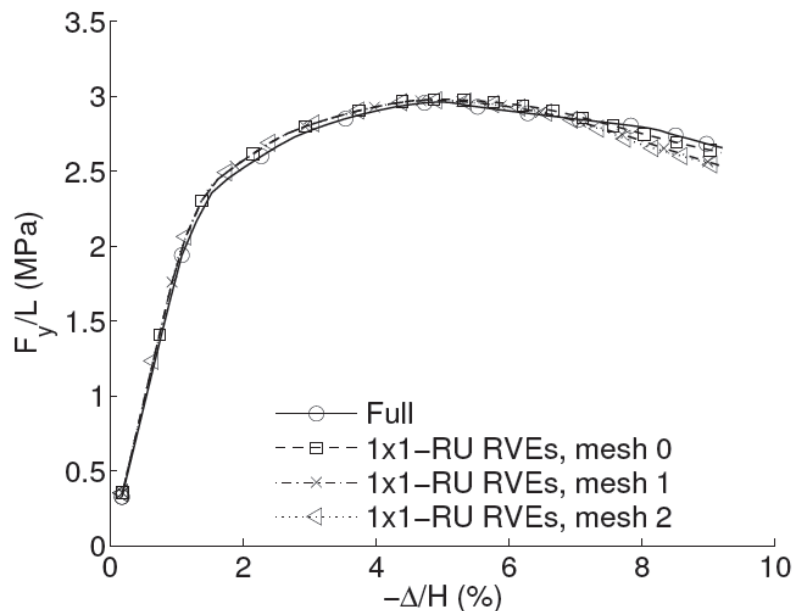
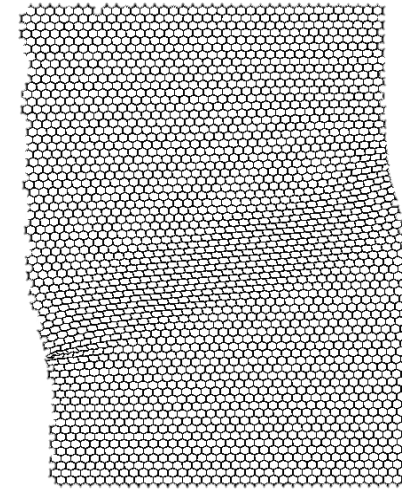
Multiscale with different macro-meshes



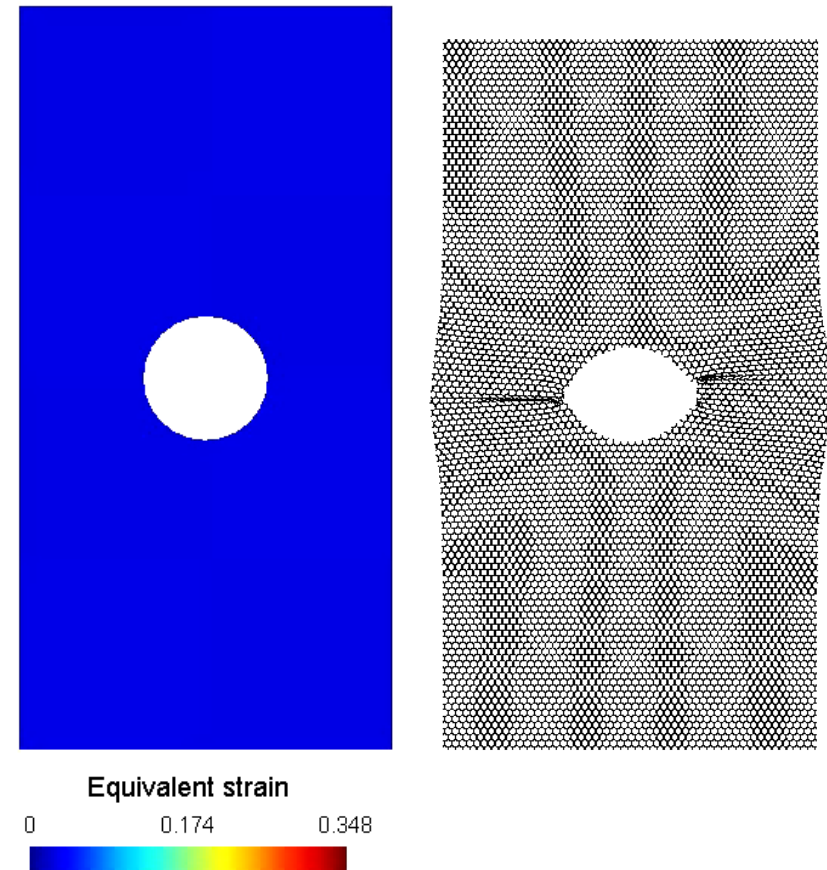
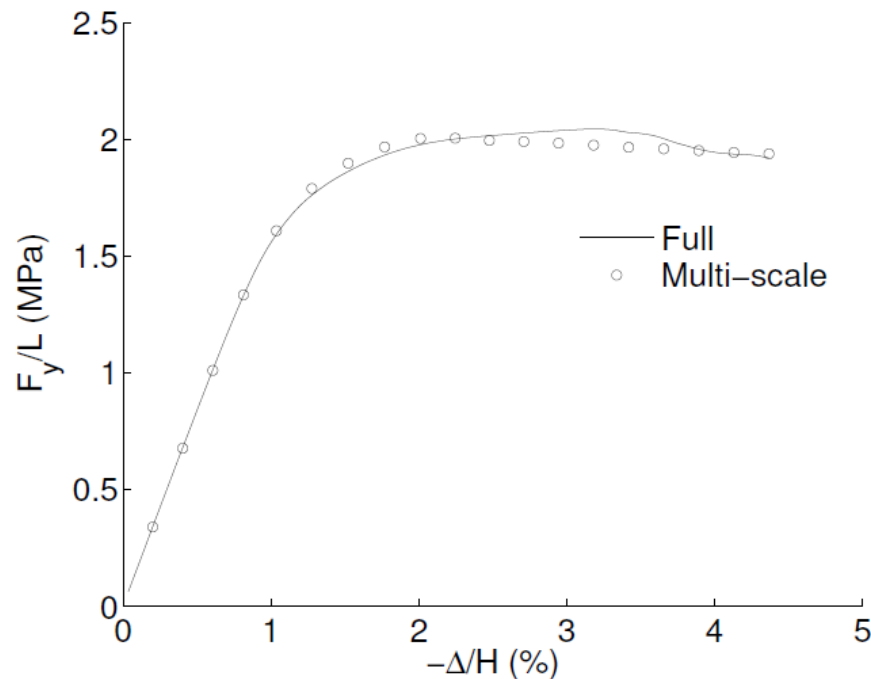
Computational homogenization for foamed materials

- Compression of an hexagonal honeycomb (2)

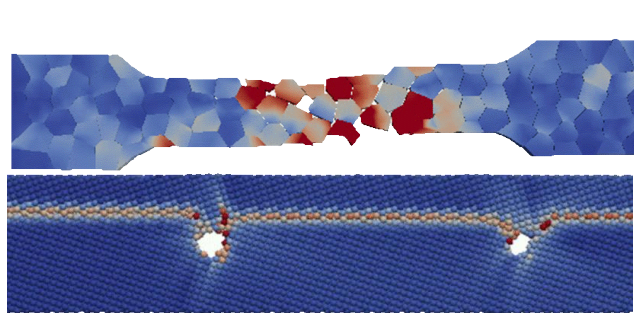
- Captures of the softening onset
- Captures the softening response
- No macro-mesh size effect



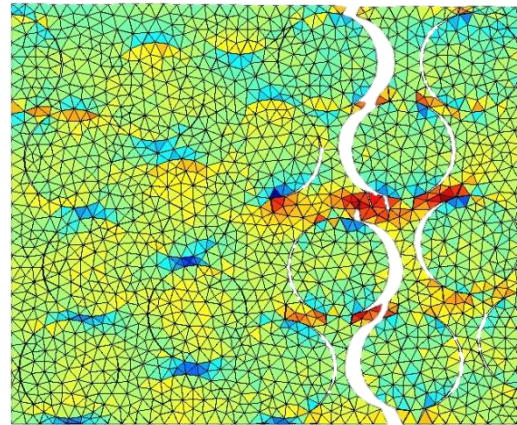
- Compression of an hexagonal honeycomb plate with a centered hole
 - Results given by full and multi-scale models are comparable



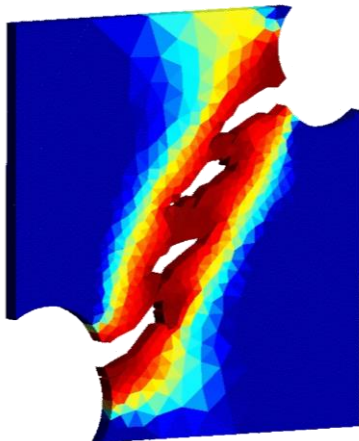
- Non-local damage-enhanced mean-field-homogenization
 - MFH with damage model for the matrix material
 - Non-local implicit formulation
 - Can capture the strain softening
 - More in
 - 10.1016/j.ijsolstr.2013.07.022
 - 10.1016/j.ijplas.2013.06.006
 - 10.1016/j.cma.2012.04.011
 - 10.1007/978-1-4614-4553-1_13
- Computational homogenization for foamed materials
 - Second-order FE² method
 - Micro-buckling propagation
 - General way of enforcing PBC
 - More in
 - 10.1016/j.cma.2013.03.024
 - 10.1016/j.commatsci.2011.10.017
 - 10.1016/j.ijsolstr.2014.02.029
- Open-source software
 - Implemented in GMSH
 - <http://geuz.org/gmsh/>



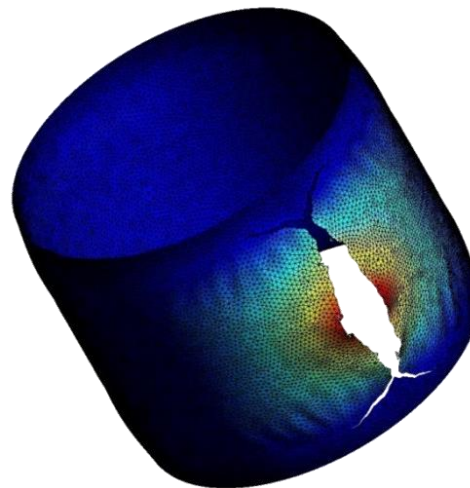
QC method for grain-boundary sliding



DG-based fracture framework



Damage to crack transition



DG-based fracture framework

Ludovic Noels, G. Becker,
L. Homsy, V. Lucas, S. Mulay,
V.-D. Nguyen, V. Péron-Lühns,
V.-H. Truong, F. Wan, L. Wu

SVE size effect on meso-scale properties

