

Non-local multiscale analyzes of composite laminates based on a damage-enhanced meanfield homogenization formulation

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Content

• Introduction

Mean-field homogenization Multi-scale modelling

• Mean-Field-Homogenization with non-local damage

- Incremental secant approach idea
- Non-local damage-enhanced incremental secant approach

• Finite-element implementation

- Direct resolution
- Staggered resolution

Applications

- Laminate with unloading
- Laminate with a hole

Conclusions



Introduction

• Multiscale methods

- Macro-scale
 - FE model
 - At one integration point $\overline{\epsilon}$ is know, $\overline{\sigma}$ is sought





• Multiscale methods

- Macro-scale
 - FE model
 - At one integration point $\overline{\epsilon}$ is know, $\overline{\sigma}$ is sought



- Micro-scale
 - Semi-analytical model
 - Predict composite meso-scale response
 - From components material models



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Multiscale methods

- Macro-scale _
 - FE model
 - At one integration point $\overline{\epsilon}$ is know, $\overline{\sigma}$ is sought
- Transition _
 - Downscaling: $\overline{\epsilon}$ is used as input of the MFH model •
 - Upscaling: σ is the output of the MFH model
- Micro-scale _
 - Semi-analytical model •
 - Predict composite meso-scale response •
 - From components material models •

Assumptions:

 $L_{\text{macro}} >> L_{\text{RVF}} >> L_{\text{micro}}$







Introduction

- Semi analytical Mean-Field Homogenization ۲
 - Based on the averaging of the fields _

$$\langle a \rangle = \frac{1}{V} \int_{V} a(X) \mathrm{d}V$$

- Meso-response _
 - From the volume ratios ($v_0 + v_1 = 1$)

$$\begin{cases} \overline{\boldsymbol{\sigma}} = \langle \boldsymbol{\sigma} \rangle = v_0 \langle \boldsymbol{\sigma} \rangle_{\omega_0} + v_{\mathrm{I}} \langle \boldsymbol{\sigma} \rangle_{\omega_{\mathrm{I}}} = v_0 \boldsymbol{\sigma}_0 + v_{\mathrm{I}} \boldsymbol{\sigma}_{\mathrm{I}} \\ \overline{\boldsymbol{\varepsilon}} = \langle \boldsymbol{\varepsilon} \rangle = v_0 \langle \boldsymbol{\varepsilon} \rangle_{\omega_0} + v_{\mathrm{I}} \langle \boldsymbol{\varepsilon} \rangle_{\omega_{\mathrm{I}}} = v_0 \boldsymbol{\varepsilon}_0 + v_{\mathrm{I}} \boldsymbol{\varepsilon}_{\mathrm{I}} \end{cases}$$



One more equation required •

$$\boldsymbol{\varepsilon}_{\mathrm{I}} = \boldsymbol{B}^{\varepsilon} : \boldsymbol{\varepsilon}_{\mathrm{0}}$$

Difficulty: find the adequate relations _

$$\begin{cases} \boldsymbol{\sigma}_{\mathrm{I}} = f(\boldsymbol{\varepsilon}_{\mathrm{I}}) \\ \boldsymbol{\sigma}_{0} = f(\boldsymbol{\varepsilon}_{0}) \\ \boldsymbol{\varepsilon}_{\mathrm{I}} = \boldsymbol{B}^{\varepsilon} : \boldsymbol{\varepsilon}_{0} \end{cases} \qquad \boldsymbol{B}^{\varepsilon} ?$$





Introduction

- Mean-Field Homogenization for different materials
 - Linear materials
 - Materials behaviours

$$\boldsymbol{\sigma}_{\mathrm{I}} = \overline{\boldsymbol{C}}_{\mathrm{I}} : \boldsymbol{\varepsilon}_{\mathrm{I}}$$
$$\boldsymbol{\sigma}_{0} = \overline{\boldsymbol{C}}_{0} : \boldsymbol{\varepsilon}_{0}$$

- Mori-Tanaka assumption $\boldsymbol{\varepsilon}^{\infty} = \boldsymbol{\varepsilon}_0$
- Use Eshelby tensort $\boldsymbol{\varepsilon}_{\mathrm{I}} = \boldsymbol{B}^{\varepsilon} (\mathrm{I}, \overline{\boldsymbol{C}}_{0}, \overline{\boldsymbol{C}}_{\mathrm{I}}) : \boldsymbol{\varepsilon}_{0}$

with
$$\boldsymbol{B}^{\varepsilon} = [\boldsymbol{I} + \boldsymbol{S} : \overline{\boldsymbol{C}}_0^{-1} : (\overline{\boldsymbol{C}}_1 - \overline{\boldsymbol{C}}_0)]^{-1}$$

- Non-linear materials
 - Define a Linear Comparison Composite
 - Common approach: incremental tangent

$$\Delta \boldsymbol{\varepsilon}_{\mathrm{I}} = \boldsymbol{B}^{\varepsilon} \left(\mathrm{I}, \overline{\boldsymbol{C}}_{0}^{\mathrm{alg}}, \overline{\boldsymbol{C}}_{\mathrm{I}}^{\mathrm{alg}} \right) : \Delta \boldsymbol{\varepsilon}_{0}$$



- Material models
 - Elasto-plastic material
 - Stress tensor $\boldsymbol{\sigma} = \boldsymbol{C}^{\text{el}} : (\boldsymbol{\varepsilon} \boldsymbol{\varepsilon}^{\text{pl}})$
 - Yield surface $f(\boldsymbol{\sigma}, p) = \boldsymbol{\sigma}^{eq} \boldsymbol{\sigma}^{Y} \boldsymbol{R}(p) \leq 0$
 - Plastic flow $\Delta \varepsilon^{\rm pl} = \Delta p N$ & $N = \frac{\partial f}{\partial \sigma}$
 - Linearization $\delta \sigma = C^{\text{alg}} : \delta \varepsilon$





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 - Linearization $\delta \sigma = C^{\text{alg}} : \delta \varepsilon$
 - Local damage model
 - Apparent-effective stress tensors $\boldsymbol{\sigma} = (1 D)\hat{\boldsymbol{\sigma}}$
 - Plastic flow in the effective stress space
 - Damage evolution $\Delta D = F_D(\boldsymbol{\varepsilon}, \Delta p)$





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- Finite element solutions for strain softening problems suffer from:
 - The loss the uniqueness and strain localization
 - Mesh dependence



The numerical results change with the size of mesh and direction of mesh





The numerical results change without convergence

- Implicit non-local approach [Peerlings et al 96, Geers et al 97, ...]
 - A state variable is replaced by a non-local value reflecting the interaction between _ neighboring material points

$$\widetilde{a}(\mathbf{x}) = \frac{1}{V_{\rm C}} \int_{V_{\rm C}} a(\mathbf{y}) w(\mathbf{y}; \mathbf{x}) \mathrm{d}V$$

Use Green functions as weight w(y; x)

$$\implies$$
 $\widetilde{a} - c \nabla^2 \widetilde{a} = a \implies$ New degrees of freedom

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- Material models
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 - Plastic flow $\Delta \varepsilon^{\rm pl} = \Delta p N$ & $N = \frac{\partial f}{\partial \sigma}$
 - Linearization $\delta \sigma = C^{\text{alg}} : \delta \varepsilon$
 - Local damage model _
 - Apparent-effective stress tensors $\boldsymbol{\sigma} = (1 D) \hat{\boldsymbol{\sigma}}$ •
 - Plastic flow in the effective stress space
 - Damage evolution $\Delta D = F_D(\boldsymbol{\varepsilon}, \Delta p)$ •
 - Non-Local damage model
 - $\Delta D = F_{D}(\boldsymbol{\varepsilon}, \Delta \boldsymbol{\widetilde{p}})$ Damage evolution
 - Anisotropic governing equation $\tilde{p} \nabla \cdot (\boldsymbol{c}_{g} \cdot \nabla \tilde{p}) = p$ •
 - Linearization

$$\delta \boldsymbol{\sigma} = \left[(1 - D) \boldsymbol{C}^{\text{alg}} - \hat{\boldsymbol{\sigma}} \otimes \frac{\partial F_D}{\partial \boldsymbol{\varepsilon}} \right] : \delta \boldsymbol{\varepsilon} - \hat{\boldsymbol{\sigma}} \frac{\partial F_D}{\partial \tilde{\boldsymbol{p}}} \delta \tilde{\boldsymbol{p}}$$





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• Problem

- We want the fibres to get unloaded during the matrix damaging process
 - For the incremental-tangent approach

$$\Delta \boldsymbol{\varepsilon}_{\mathrm{I}} = \boldsymbol{B}^{\varepsilon} \Big(\mathrm{I}, (1-D) \overline{\boldsymbol{C}}_{0}^{\mathrm{alg}}, \overline{\boldsymbol{C}}_{\mathrm{I}}^{\mathrm{alg}} \Big) : \Delta \boldsymbol{\varepsilon}_{0}$$

- To unload the fibres ($\boldsymbol{\varepsilon}_{\mathrm{I}} < 0$) with such approach would require $\overline{\boldsymbol{C}}_{\mathrm{I}}^{\mathrm{alg}} < 0$
- We cannot use the incremental tangent MFH
- We need to define the LCC from another stress state



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• Idea

- New incremental-secant approach
 - Perform a virtual elastic unloading from previous solution
 - Composite material unloaded to reach the stress-free state
 - Residual stress in components





Idea ۲

- New incremental-secant approach _
 - Perform a virtual elastic unloading from • previous solution
 - Composite material unloaded to reach the stress-free state
 - Residual stress in components
 - Apply MFH from unloaded state •
 - New strain increments (>0)

$$\Delta \boldsymbol{\varepsilon}_{\mathrm{I}/\mathrm{0}}^{\mathrm{r}} = \Delta \boldsymbol{\varepsilon}_{\mathrm{I}/\mathrm{0}} + \Delta \boldsymbol{\varepsilon}_{\mathrm{I}/\mathrm{0}}^{\mathrm{unload}}$$

Use of secant operators

$$\Delta \boldsymbol{\varepsilon}_{\mathrm{I}}^{\mathrm{r}} = \boldsymbol{B}^{\varepsilon} \left(\mathrm{I}, (1-D) \overline{\boldsymbol{C}}_{0}^{\mathrm{Sr}}, \overline{\boldsymbol{C}}_{\mathrm{I}}^{\mathrm{Sr}} \right) : \Delta \boldsymbol{\varepsilon}_{0}^{\mathrm{r}}$$

Possibility of have unloading

$$\begin{cases} \Delta \boldsymbol{\varepsilon}_{\mathrm{I}}^{\mathrm{r}} > 0 \\ \Delta \boldsymbol{\varepsilon}_{\mathrm{I}} < 0 \end{cases}$$



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- New incremental-secant approach
 - Equations summary
 - Inputs
 - Internal variable at last increment
 - Residual tensor after virtual unloading
 - $\Delta \overline{\boldsymbol{\varepsilon}}, \Delta \widetilde{\boldsymbol{p}}$ from FE resolution
 - Solve iteratively the system

$$\begin{cases} \Delta \overline{\boldsymbol{\varepsilon}}^{(r)} = v_0 \Delta \boldsymbol{\varepsilon}_0^{(r)} + v_I \Delta \boldsymbol{\varepsilon}_I^{(r)} \\ \Delta \boldsymbol{\varepsilon}_I^r = \Delta \boldsymbol{\varepsilon}_I + \Delta \boldsymbol{\varepsilon}_I^{\text{unload}} \\ \Delta \boldsymbol{\varepsilon}_0^r = \Delta \boldsymbol{\varepsilon}_0 + \Delta \boldsymbol{\varepsilon}_0^{\text{unload}} \\ \Delta \boldsymbol{\varepsilon}_I^r = \boldsymbol{B}^{\varepsilon} \left(\mathbf{I}, (1 - D) \overline{\boldsymbol{C}}_0^{\text{Sr}}, \overline{\boldsymbol{C}}_I^{\text{Sr}} \right) : \Delta \boldsymbol{\varepsilon}_0^r \end{cases}$$



• With the stress tensors

$$\begin{cases} \overline{\boldsymbol{\sigma}} = v_0 \boldsymbol{\sigma}_0 + v_{\mathrm{I}} \boldsymbol{\sigma}_{\mathrm{I}} \\ \boldsymbol{\sigma}_{\mathrm{I}} = \boldsymbol{\sigma}_{\mathrm{I}}^{\mathrm{res}} + \overline{\boldsymbol{C}}_{\mathrm{I}}^{\mathrm{Sr}} : \Delta \boldsymbol{\varepsilon}_{\mathrm{I}}^{\mathrm{r}} \\ \boldsymbol{\sigma}_0 = (1 - D) \hat{\boldsymbol{\sigma}}_0^{\mathrm{res}} + (1 - D) \overline{\boldsymbol{C}}_0^{\mathrm{Sr}} : \Delta \boldsymbol{\varepsilon}_0^{\mathrm{r}} \end{cases}$$



- New incremental-secant approach (2)
 - Alternative
 - For soft matrix response
 - Remove residual stress in matrix
 - Avoid adding spurious internal energy
 - Solve iteratively the system

$$\begin{cases} \Delta \overline{\boldsymbol{\varepsilon}}^{(\mathrm{r})} = v_0 \Delta \boldsymbol{\varepsilon}_0^{(\mathrm{r})} + v_{\mathrm{I}} \Delta \boldsymbol{\varepsilon}_{\mathrm{I}}^{(\mathrm{r})} \\ \Delta \boldsymbol{\varepsilon}_{\mathrm{I}}^{\mathrm{r}} = \Delta \boldsymbol{\varepsilon}_{\mathrm{I}} + \Delta \boldsymbol{\varepsilon}_{\mathrm{I}}^{\mathrm{unload}} \\ \Delta \boldsymbol{\varepsilon}_0^{\mathrm{r}} = \Delta \boldsymbol{\varepsilon}_0 + \Delta \boldsymbol{\varepsilon}_0^{\mathrm{unload}} \\ \Delta \boldsymbol{\varepsilon}_{\mathrm{I}}^{\mathrm{r}} = \boldsymbol{B}^{\varepsilon} \left(\mathbf{I}, (1-D) \overline{\boldsymbol{C}}_0^{\mathrm{S0}}, \overline{\boldsymbol{C}}_{\mathrm{I}}^{\mathrm{Sr}} \right) : \Delta \boldsymbol{\varepsilon}_0^{\mathrm{r}} \end{cases}$$

• With the stress tensors

$$\begin{cases} \overline{\boldsymbol{\sigma}} = v_0 \boldsymbol{\sigma}_0 + v_{\mathrm{I}} \boldsymbol{\sigma}_{\mathrm{I}} \\ \boldsymbol{\sigma}_{\mathrm{I}} = \boldsymbol{\sigma}_{\mathrm{I}}^{\mathrm{res}} + \overline{\boldsymbol{C}}_{\mathrm{I}}^{\mathrm{Sr}} : \Delta \boldsymbol{\varepsilon}_{\mathrm{I}}^{\mathrm{r}} \\ \boldsymbol{\sigma}_0 = (1 - D) \overline{\boldsymbol{C}}_0^{\mathrm{S0}} : \Delta \boldsymbol{\varepsilon}_0^{\mathrm{r}} \end{cases}$$







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- Weak formulation
 - Strong form $\begin{cases}
 \nabla \cdot \overline{\sigma}^{T} + f = 0 & \text{for the homogenized composite material} \\
 \widetilde{p} - \nabla \cdot (c_{g} \cdot \nabla \widetilde{p}) = p & \text{for the matrix phase}
 \end{cases}$
 - Boundary conditions

 $\begin{cases} \boldsymbol{\sigma} \cdot \boldsymbol{n} = \boldsymbol{T} \\ \boldsymbol{n} \cdot \left(\boldsymbol{c}_{g} \cdot \nabla \widetilde{p} \right) = 0 \end{cases}$

Finite-element discretization

$$\begin{cases} \widetilde{p} = N_{\widetilde{p}}^{a} \widetilde{p}^{a} \\ u = N_{u}^{a} u^{a} \end{cases}$$
$$\implies \begin{bmatrix} K_{uu} & K_{u\widetilde{p}} \\ K_{\widetilde{p}u} & K_{\widetilde{p}\widetilde{p}} \end{bmatrix} \begin{bmatrix} du \\ d\widetilde{p} \end{bmatrix} = \begin{bmatrix} F_{ext} - F_{int} \\ F_{p} - F_{\widetilde{p}} \end{bmatrix}$$





- Resolution strategies
 - Fully coupled resolution

$$\begin{bmatrix} \boldsymbol{K}_{uu} & \boldsymbol{K}_{u\tilde{p}} \\ \boldsymbol{K}_{\tilde{p}u} & \boldsymbol{K}_{\tilde{p}\tilde{p}} \end{bmatrix} \begin{bmatrix} d\boldsymbol{u} \\ d\tilde{\boldsymbol{p}} \end{bmatrix} = \begin{bmatrix} \boldsymbol{F}_{ext} - \boldsymbol{F}_{int} \\ \boldsymbol{F}_{p} - \boldsymbol{F}_{\tilde{p}} \end{bmatrix}$$

- Staggered dynamic resolution
 - Explicit resolution of the displacement dofs

$$\ddot{\boldsymbol{u}}^{n+1} = \frac{1}{1-\alpha_M} \boldsymbol{M} \left[\boldsymbol{F}_{ext}^n - \boldsymbol{F}_{int}^n \right] - \frac{\alpha_M}{1-\alpha_M} \boldsymbol{u}^n$$
$$\dot{\boldsymbol{u}}^{n+1} = \dot{\boldsymbol{u}}^n + \Delta t \left[1 - \gamma_M \right] \ddot{\boldsymbol{u}}^n + \Delta t \gamma_M \ddot{\boldsymbol{u}}^{n+1}$$
$$\boldsymbol{u}^{n+1} = \boldsymbol{u}^n + \Delta t \dot{\boldsymbol{u}}^n + \Delta t^2 \left[\frac{1}{2} - \beta_M \right] \ddot{\boldsymbol{u}}^{n+1} + \Delta t^2 \beta_M \ddot{\boldsymbol{u}}^{n+1}$$

• Resolution of the non-local equation once every *N* steps

$$\boldsymbol{K}_{\widetilde{p}\widetilde{p}}d\widetilde{\boldsymbol{p}}=\boldsymbol{F}_{p}-\boldsymbol{F}_{\widetilde{p}}$$



Finite-element implementation

• Mesh-size effect

- Fictitious composite
 - 30%-UD fibres
 - Elasto-plastic matrix with damage
- Notched ply





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Laminate: calibration

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- Carbon-fibres reinforced epoxy _
 - 60%-UD fibres •
- [-45₂/45₂]_S staking sequence





- Laminate plate with hole ۲
 - Carbon-fibres reinforced epoxy _
 - 60%-UD fibres •
 - Elasto-plastic matrix with damage
 - $[-45_2/45_2]_S$ staking sequence



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- Laminate plate with hole (2)
 - Carbon-fibres reinforced epoxy
 - 60%-UD fibres
 - Elasto-plastic matrix with damage
 - [-45₂/45₂]_S staking sequence















- Laminate plate with hole (3)
 - Carbon-fibres reinforced epoxy
 - 60%-UD fibres
 - Elasto-plastic matrix with damage
 - $[-45_2/45_2]_S$ staking sequence





Conclusions

- New damage-enhanced incremental secant MFH approach
 - Efficient computationally
 - Allows fibres unloading during matrix softening

• Non-local damage-enhanced MFH

- Good description of the meso-scale response
- Can be used to study coupons problems
- Perspective
 - From damage to crack

