

## 4<sup>th</sup> SYMPOSIUM OF VKI PHD RESEARCH Development of advanced models for transition to turbulence in hypersonic flows Prediction of transition under uncertainties

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#### **Introduction & Motivation**



## **Introduction & Motivation**

#### <u>Transition prediction – The State of the Art</u>

- Experiments : empirical criteria and correlation (Shuttle, Van Driest)
  - Good : successfully used (Apollo, Shuttle);
  - Bad : expensive, limited in time and no real operating conditions (Re, Ma);
- **<u>CFD</u>** : Transition models (Menter, Goldberg,  $R-\gamma$ )
  - Good : fast , design;
  - Bad : simplified physics, very sensitive to free stream conditions (Re, Ma, Tu);



## **Introduction & Motivation**

#### Three pillars for predictive engineering simulations



#### <u> Transition prediction – What we propose</u>

- Introduce *Uncertainty Quantification* (UQ) in deterministic simulations for transition prediction to :
  - Take into account the *physical variability* of the system to simulate
  - Transition is a *stochastic process*
  - Improve and verify transition tools currently used in design
  - A stochastic model for transition prediction does not yet exist

## Outline

#### 1. Introduction to deterministic and probabilistic tools

- Linear Stability Theory and transition prediction
- Uncertainty Quantification and numerical simulations
- 2. Formulation of the method
  - Assumptions for the deterministic simulations
  - Description of the method
- 3. The VKI-H3 test case
  - The forward problem
  - The inverse problem

#### 4. Conclusions & Future works

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#### Linear Stability Theory and transition prediction

- Small disturbances: baseline + disturbances;
- Linearization of Navier-Stokes equations + parallel flow approximation;

$$u = \bar{u} + u' \qquad \qquad \frac{\partial u'}{\partial x} + \frac{\partial v'}{\partial y} = 0$$
  
baseline disturbances  
 $\bar{v} = 0$   $\bar{u} = \bar{u}(y)$   
parallel flow  

$$\frac{\partial u'}{\partial t} + \bar{u}\frac{\partial u'}{\partial x} + v'\frac{\partial \bar{u}}{\partial y} = -\frac{1}{\rho}\frac{\partial p'}{\partial x} + \nu \left[\frac{\partial^2 u'}{\partial x^2} + \frac{\partial^2 u'}{\partial y^2}\right]$$

• Wave like disturbances;

$$q'(x,y,t) = \tilde{q}(y) e^{i(\alpha x - \omega t)}$$
 Propagation in time (real)

• Space amplification theory **Propagation in space (complex)** 

Degrez G., "Two dimensional boundary layer", 2012

#### Linear Stability Theory and transition prediction

• Orr-Sommerfeld equations;

$$i\alpha(\bar{u}-c)(\tilde{v}''-\alpha^2\tilde{v}) - i\alpha\tilde{v}\bar{u}'' = \frac{1}{R}\left[\tilde{v}^{iv} - 2\alpha^2\tilde{v}'' + \alpha^4\tilde{v}\right]$$

-1

- B.C. : disturbances vanish at the wall and in the far field;  $\tilde{v}(0) = \tilde{v}'(0) = 0$   $\lim_{y \to \infty} \tilde{v}(y) = \lim_{y \to \infty} \tilde{v}'(y) = 0$
- Eigenvalue problem;



Amplification rates contour lines for Blasius velocity profile plotted in the R-  $\alpha$  plane

Neutral Stability curve : c<sub>i</sub> = 0 boundary between damped and amplified disturbances

Degrez G., *"Two dimensional boundary layer"*, Course Notes, 2012



#### Linear Stability Theory and transition prediction

• *e<sup>N</sup>* transition prediction method;



$$N = \log \frac{A}{A_o} = \int_{x_0}^x -\alpha_i dx$$

N factors computed on the HIFire I reentry vehicle Mach number = 5.28, H = 21km



- **Transition** : *N*-factor = *N*<sub>exp</sub>
- N = N(wind tunnel, free stream parameters)
- *N*-factor = **4-5** (WT) , **13-14** (Flight);

#### **Uncertainty quantification and numerical simulations**

- Goal : study how physical variability of systems affects Quantity of Interest
- **UQ** : End-to-end study of the reliability of scientific predictions;



Mean velocity divergence field

laccarino G. *et a*l., "Numerical methods for uncertainty propagation in high speed flows," V European Conference on Computational Fluid Dynamics, 2011

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#### Assumptions for the deterministic simulations

• Wave-like disturbances



- 2D waves ( $\beta$  vanishes );
- Spatial amplification theory :  $\omega$  real,  $\alpha$  complex, wave propagation speed  $c = \omega/\alpha_r$ , amplification rate in space  $-\alpha_i$ ;
- Transition prediction with the  $e^{N}$  method : VKI-H3 *N*-factor = 5 (Mach 6 WT);

#### **Description of the method**





7 degree half-cone at Mach 6, 2012.

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#### **Description of the method**

Definition of the uncertainties 2.



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N



#### **Description of the method**

4. Probability of transition



$$p_T = 1 - \int_0^{N_{crit}} p df(\bar{N}) d\bar{N} = \gamma$$

- $p_{\tau}$  , probability of transition
- $N_{crit}$ , critical *N*-factor at the transition onset



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#### Study of natural transition on a 7° half-cone model

- Transition detected by surface measurements of the heat flux;
- Different Reynolds number conditions;



#### The forward problem

- **Goal** : computation of the probability of transition caused by assumed freestream perturbation spectrum (Frequency distribution);
- **Assumption** : transition caused by perturbations in the BL upstream of the transition location;
- **Transfer function** : Linear Stability analysis to compute N=N(F);

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#### The forward problem

• **UQ approach** : free stream perturbations as *pdf* of the Frequency with *normal distributions* ;

Test case	$\mu_f$ [kHz]	$\sigma_F$ [kHz]	
Low Reynolds	330	10	Range [kHz]
Medium Reynolds	410	20	$200 \div 800$
High Reynolds	480	25	

Computation of the *probability of transition* and comparison with experiments;





#### The inverse problem

• **VKI-H3 Low-Re**: MCMC to obtain the posterior *pdf* of the mean and the variance of the frequency distribution (Geweke's test for convergence)



#### The inverse problem

• Intermittency distribution : VKI-Low Reynolds



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#### **Conclusions**

- **Goal** : combination of deterministic and probabilistic tools for transition prediction in high speed flow;
- **Method** : forward problem (intermittency distribution for given conditions) and inverse problem (frequency distribution for given measurements);
- Added value
  - Forward problem –intermittency distributions resembling experimental data with fast and reliable computations (LST + e<sup>N</sup> method);
  - *Inverse problem* inferring perturbation spectrum for given conditions;

#### **Future works**

- **RANS model for transition prediction**: using the forward model to build a look-up table to obtain intermittency distributions at different conditions (Stanford SU2 code);
- New stochastic transition prediction method;
- **Comparison with experimental data** : assessment of the assumptions for the inverse problem (frequency distributions) and comparison with experimental data;



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#### The inverse problem

- D input parameters  $s = (s_1, s_2, ..., s_D)$  through the computational model f(s) to give K outputs  $m = (g_1(f(s_1, r)), g_2(f(s_2, r)), ..., g_k(f(s_k, r)))$  with r auxiliary parameters  $s = (r_1, r_2, ..., r_N)$ ;
- **The forward problem** : solving *m* with given *s* and *r*;
- **The inverse problem** : inferring *s* given the measurements of *m* for given *r*;
- **Parameters** : input  $s = (s_1, s_2) = (\mu_F, \sigma_F)$ , auxiliary r (conditions for the test cases), output  $m = (\gamma_1, \gamma_2 ..., \gamma_k)$  at  $x_1, x_2 ..., x_k$ ;
- **The strategy** : given set of noisy measurements  $m = m + \eta = (\gamma_1, \gamma_2 ..., \gamma_k)$  to seek for the input parameters  $s = (\mu_F, \sigma_F)$  using the computational model f(s);
- The Bayesian inversion :

$$p(s|m) = \frac{p(m|s) \times p(s)}{p(m)} \propto p(m|s) \times p(s)$$

- p(s/m) = posterior pdf (probability of the input given the measurements)
- p(m|s) = likelihood *pdf* (probability of the measurements given the input)
- *p(s)* = prior *pdf* (information on the input parameters)

## The MCMC algorithm





Statistical inverse analysis and stochastic modelling of transition – part 2

#### The Bayesian inversion





#### The inverse problem

• *MCMC algorithm* : Markov Chain Monte Carlo to obtain the posterior *pdf* 

