

9 EVALUATION OF CRACK PROPAGATION IN AN OVERLAY SUBJECTED TO TRAFFIC AND THERMAL EFFECTS

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Abstract

The authors are actively involved in the development of finite element programs to study crack growth in the overlay of a structure including an anti-reflective cracking layer based on a bitumen impregnated geotextile. The program considers various positions of lorry axles and determines the total effect on the crack tip. This is combined with a temperature variation in the whole structure.

The two effects are combined in order to evaluate local damages. The F.E. program works on the following assumptions :

- the materials are elastic;
- secant moduli of the bitumen are taken into account;
- the crack has an effective width equal to the overlay maximum aggregate size;
- traffic and thermal damages can be added;
- the road structure is a semi-rigid type structure;
- the crack opening follows mode 1;
- the geotextile acts as a bitumen container and not as a reinforcement.

Keywords : reflective cracking, simulation, finite elements, model, traffic, thermal effects.

1 Introduction

The SAPLI 5 finite element programme has been developed at the M.S.M. department of Liege University for structural design. It works on mainly linear elastic mode.

As such it has been applied to simulate the crack growth in a structure constituted of (from top to bottom) (figure 1) :

- an asphalt overlay (6 cm);
- an interlayer made of a bitumen saturated geotextile (2 mm);
- a cement concrete slab (20 cm) including a vertical crack (5 mm width);
- a 15 cm thick granular subbase;
- the soil.

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6cm
20cm
15cm

For this part of the work was to traffic and the

The service number of cycle to propagate the

Fracture of concrete has programs as systems distributed over proposed by "Blunt-crack beam"

It has Clauwaert-Franck laboratory observations in concretes prepared zone instead of may be equal to

The SAPLI 5 traffic and thermal damages induced combined which service life.

2 Traffic and

2.1 Basic principles
The investigation elements by means The principal effects evaluated for effects taking account. Once the element, this calculation step as a simulation step until the the overlay.

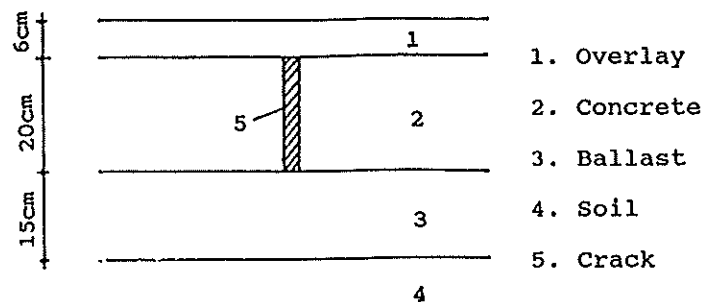


Figure 1

For this particular configuration, the objective of the work was to estimate the overlay service-life under traffic and thermal actions simultaneously.

The service life of the overlay is defined as the number of cycles of traffic and thermal loads necessary to propagate the crack across the overlay.

Fracture of a heterogeneous aggregate material such as concrete has been modelled in large finite element programs as systems of parallel cracks that are densely distributed over the finite element. This was originally proposed by Rashid (1982) and is known as the "Blunt-crack band Theory".

It has been presented by Haas (1989); Clauwaert-Francken (1989) and proved by field and laboratory observations (Rigo et al, 1989) that failures in concretes propagate with tortuosity creating a damaged zone instead of a unique crack. The cracked band width may be equal to the largest aggregates size.

The SAPLI 5 program has been used to evaluate the traffic and the thermal stresses separately and the damages induced by these stresses were evaluated and combined which permits the estimation of the overlay service life.

2 Traffic and thermal damages and their combination

2.1 Basic principle

The investigated structure has been divided into small elements by means of a 2D-finite element mesh (figure 3). The principal stresses and the resulting damages are evaluated for each element under traffic and/or thermal effects taking the stress history of each element into account. Once the damage is equal to unity for a given element, this one is removed from the mesh and the next calculation step is based on a mesh with an empty element as a simulation of the crack. This is realized step by step until the simulated crack reaches the top surface of the overlay.

2.2 Damage and crack propagation

The damage of an element is defined as the ratio between the defects surface (S_D) and the total surface (S).

$$D = \frac{S_D}{S} \quad (1)$$

If S' represents the resisting surface,

$$S' = S - S_D \quad (2)$$

The effective stress (σ') in an element is subjected to modifications in function of the evolution of the effective resisting surface

$$\sigma' = \sigma \cdot \frac{S}{S'} = \frac{\sigma}{1 - D} \quad (3)$$

If a material is subjected to damage, its elastic behavior is also affected and

$$E' = (1 - D) \cdot E \quad (4)$$

where

E = elastic modulus of the virgin material

E' = elastic modulus of the damaged material.

By assuming that the damage has a linear evolution versus the number of the applied stresses cycles,

$$D = \sum D_i = \sum \frac{n_i}{N_i} \quad (5)$$

where

D_i = the damage caused by a given stress level σ_i

N_i = the number of stress cycles at level σ_i corresponding to the failure ($D_i = 1$).

It is given by the well known Wöhler curves

n_i = the number of stress cycles at level σ_i already applied to the element.

Schmidt (1989) presented a model of the crack propagation into a bituminous surfacing. This principle is applicable to each element of the finite element mesh presented in figure 1 :

- for the initial mesh configuration (overlay without any crack) ($k = 1$). The stresses are evaluated for each element. For the one that is the most loaded it is possible to evaluate the service-life by referring to the Wöhler curve

$$n_1 = N_{1,1} = (N_{i,k})_{i=1, k=1} \quad (6)$$

where

n_1 = number of element before failure

$N_{1,1}$ = number of element in the initial configuration

- after failure of an element, the configuration of the element (element configuration) is given by equation (7)

$$D = \frac{n_2}{N_{2,2}} + \dots$$

where

n_2 = see equation (5)

n_2 = number of element to be removed

$N_{2,1}$ = the number of element in the stress configuration

$\frac{n_1}{N_{2,1}}$ = damage of failed element

$N_{2,2}$ = the number of element in the configuration

- for configuration before element failure

$$n_2 = N_{2,2} \cdot D$$

- for the configuration after failure

$$n_3 = N_{3,3} \cdot D$$

- the same procedure for the configuration of the overlay

- a more general case hereafter

where

n_1 = number of cycles applied to the virgin structure before removing the first element
 $N_{1,1}$ = number of cycles supported by the most loaded element ($N_{1,1}$) before failure with the first mesh configuration.

- after failure and removal of the first most loaded element, the stresses are evaluated with the new finite element configuration. For the newly most loaded element (element 2), the service-life is evaluated by equation (7)

$$D = \frac{n_2}{N_{2,2}} + \frac{n_1}{N_{2,1}} \quad (7)$$

where

n_1 = see equation (6)
 n_2 = number of cycles applied to the second element to be removed from the mesh
 $N_{2,1}$ = the maximum number of cycles to be applied for the stress on element 2 evaluated with configuration 1 of the finite element mesh

$\frac{n_1}{N_{2,1}}$ = damage supported by element 2 before element 1 failed (with configuration 1)

$N_{2,2}$ = the maximum number of cycles to be applied on element 2 for the stress evaluated with configuration 2

- for configuration 2, the number of cycles to apply before element 2 fails is

$$n_2 = N_{2,2} \cdot \left(1 - \frac{n_1}{N_{2,1}}\right) \quad (8)$$

- for the configuration 3, the number of cycles to be applied before element 3 fails is

$$n_3 = N_{3,3} \cdot \left(1 - \frac{n_1}{N_{3,1}} - \frac{n_2}{N_{3,2}}\right) \quad (9)$$

- the same procedure is repeated for various configurations until the crack reaches the top surface of the overlay;

- a more general version of equation (9) is given hereafter

$$n_i = N_{i,i} \cdot \left(1 - \sum_{j=1}^{j=i-1} \frac{n_j}{N_{i,j}}\right) \quad (10)$$

or

$$n_i = N_{i,i} \cdot (1 - \sum D_i) \quad (11)$$

The total service-life of the overlay is equal to

$$n_{TOT} = n_1 + n_2 + n_3 \dots \quad (12)$$

It must be pointed out that at each calculation level the deterioration of the material also affects the value of the elastic modulus and equation (4) is to be applied to each finite element at each state of the crack propagation.

2.3 Combination of traffic and thermal damage

The assumption is made that the traffic and thermal damages may be calculated separately and additioned (principle of the damage superposition).

It must also be considered that thermal stresses are also cyclic loads (one cycle/day) so that the approach developed in section 2.2. herebefore is also applicable.

Considering that the traffic loads are applied with a frequency of f cycles per day, the thermal loads are applied at a frequency of 1 cycle per day :

$$\text{traffic frequency} = f \cdot \text{thermal frequency} \quad (13)$$

The combination of the traffic and thermal damages may thus be done as follow :

- for each element, the traffic (σ_{tr}) and thermal (σ_{th}) stresses are calculated separately;
- the Wöhler curves for the overlay permit the evaluation of
 - N_{tr} for σ_{tr} and
 - N_{th} for σ_{th}
- the number of cycles necessary to obtain the failure of the most loaded element of the configuration is calculated by

$$D_i = D_{i-1} + \frac{n}{N_{tr}} + \frac{n/f}{N_{th}} \quad (14)$$

where

D_i = damage of the most loaded element for a given configuration (= 1 at failure)

D_{i-1} = damage of the same element induced by the stresses supported during the previous configurations

$$\frac{n}{N_{tr}} = \text{traffic}$$

$$\frac{n}{f \cdot N_{th}} = \text{therm.}$$

Equation 13 :

$$n = (1 - D_{i-1})$$

It must be separately for

$$E_{tr,i} = E_{tr,o}$$

$$E_{th,i} = E_{th,o}$$

where

$E_{tr,i}$ = elastic configu

$E_{tr,o}$ = elastic traffic

$E_{th,i}$ and $E_{th,o}$

3 Practical an

3.1 Materials

The proposal me structure. The were the next :

- overlay

E (Mpa)
5400
3100
2100
1330
390
210

The frequency case of traff totally corre to be valid f

10)

$$\frac{n}{N_{tr}} = \text{traffic damage}$$

11)

$$\frac{n}{f \cdot N_{th}} = \text{thermal damage}$$

Equation 13 might also be presented as follow :

12)

$$n = (1 - D_{i-1}) \cdot \frac{N_{tr} \cdot N_{th} \cdot f}{N_{tr} + N_{th} \cdot f} \quad (15)$$

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It must be reminded that equation (4) is applicable separately for traffic and thermal loadings :

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$$E_{tr,i} = E_{tr,o} \cdot (1 - D_i) \quad (16)$$

$$E_{th,i} = E_{th,o} \cdot (1 - D_i)$$

where

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$E_{tr,i}$ = elastic modulus for the traffic loads for the configuration i

$E_{tr,o}$ = elastic modulus for the virgin material for the traffic loads

$E_{th,i}$ and $E_{th,o}$ = same but for thermal loads.

(13)

3 Practical analysis of the reflection crack growth

s may

3.1 Materials characteristics

th)

The proposal methodology has been applied to a semi-rigid structure. The basic characteristics of the materials were the next :

- overlay

ire of

E (Mpa)	Frequency (Hz)	Temperature (°C)
5400	10	20
3100	0,1	10
2100	0,1	12,5
1330	0,1	15
390	0,1	25
210	0,1	30

(14)

The frequency of 10 Hz is supposed to be valid for the case of traffic loading while 0,1 Hz (which is not totally correct but no data are available) is supposed to be valid for thermal loading.

- the interface

Characteristics	Traffic	Thermal
G	2 Mpa	1 Mpa
E_V	1300 Mpa	1000 Mpa
E_H	4 MPa	2 Mpa
ν	0	0

where

G = the interface elastic shear modulus

E_V = the vertical (perpendicular to the interface plane) elastic modulus

E_H = the elastic modulus parallel to the interface plane

ν = POISSON'S ratio.

E_V is equal to the bitumen bulk modulus in order to take into account the container effect of the geotextile. The effect also induces $\nu = 0$.

- The concrete

E = 15.000 Mpa

$\nu = 0,3$

$\rho = 2300 \text{ kg/m}^3$

$\alpha = 12 \cdot 10^{-6}$

The crack in this layer is supposed to be empty.

- The granular sub-grade

E = 200 Mpa

$\nu = 0,5$

$\rho = 2000 \text{ kg/m}^3$

$\alpha = 12 \cdot 10^{-6}$

which corresponds to a rigidity modulus $K = 100 \text{ Mpa/m}$.

3.2 The loads

The structure is supposed to present one crack each 5 meters.

The traffic loads correspond to the local standards (13 ton/axle) and are modelled by a pressure of 6,62 bars on a 9 cm width band. This pressure and band width give in the 2D-model approximately the same principal stresses in the overlay in the cross section on top of the crack as those calculated by a 3D-model with 13 T applied with a pressure of 7 bars on a surface of 30 x 30 cm. The loads are applied at a frequency of 10 Hz. Delcuve (1990) and Hendrick (1992) have shown that the most critical position of this load correspond to the position showed on figure 2.

The overlay temperature is distributed as follow : 30°C at the top surface and 10°C at the interface. The concrete slab is subjected to a $\Delta T = 10^\circ\text{C}$ which corresponds to a total movement of a 0,6 mm on 5 m.

It is supposed
section per day

3.3 Traffic load
The load is applied

Figure 2 :

Only half of the loads are obtained by anti-symmetric loading. On the other side, at the center of the horizontal movement, relative vertical movement is zero.

Figure 3 shows the overlay under load.

It must be clear that the crack propagation and appears to be

3.4 Thermal movement
The movement of the concrete slab stresses in the concrete and nearly vertical

It is supposed that 1000 vehicles are passing on this section per day.

3.3 Traffic loading

The load is applied as shown on figure 2.

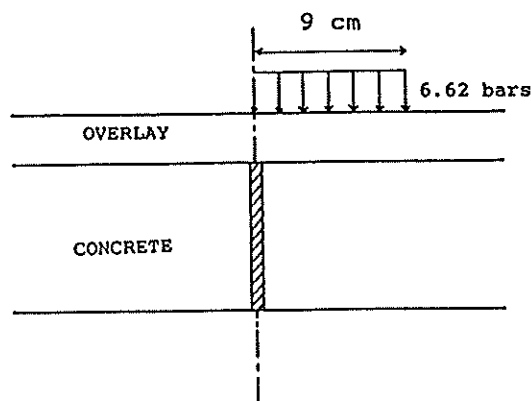


Figure 2 : traffic loads applied to the structure

Only half of the structure is studied and the results are obtained by the superposition of a symmetric and a anti-symmetric loading configuration.

On the other hand, for the traffic loading the nodes at the central cross-section of the overlay have no horizontal movement (for symmetry reasons) and the relative vertical movements are blocked.

Figure 3 shows the propagation of the crack in the overlay under traffic loading only.

It must be observed that the load presence influences the crack propagation. This one is deviated transversely and appears to propagate at $\pm 45^\circ$.

3.4 Thermal loads

The movement of the concrete base induces very important stresses in the overlay and the crack propagation is fast and nearly vertical (figure 4).

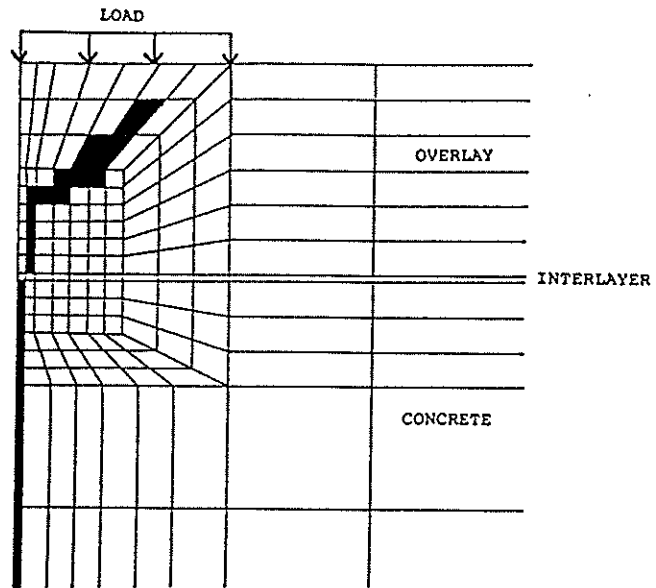


Figure 3 : Crack propagation in the overlay under repeated traffic loads

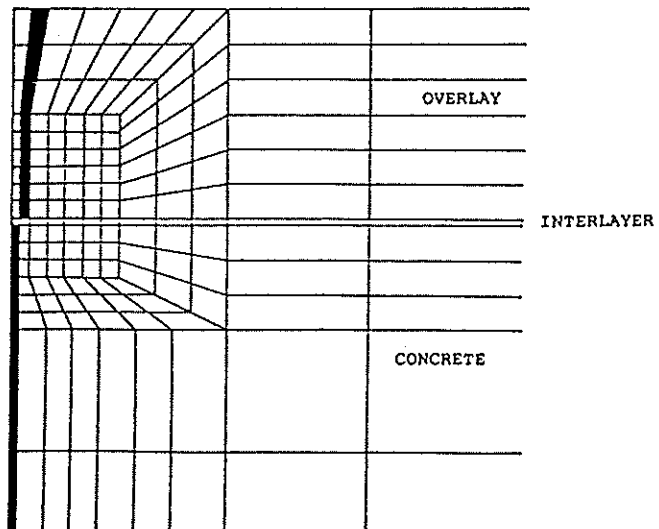


Figure 4 : Crack propagation in the overlay under thermal load

3.5 Combined
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(figure 5).

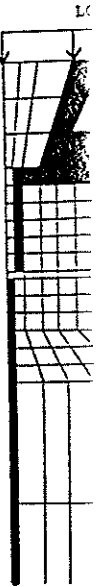


Figure 5

3.6 Crack pro
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expressed in
traffic cycle

3.5 Combined action

When traffic and thermal effects are combined the crack orientation is intermediate between the two above described situations : the crack propagates more vertically than in the case of traffic loads alone (figure 5).

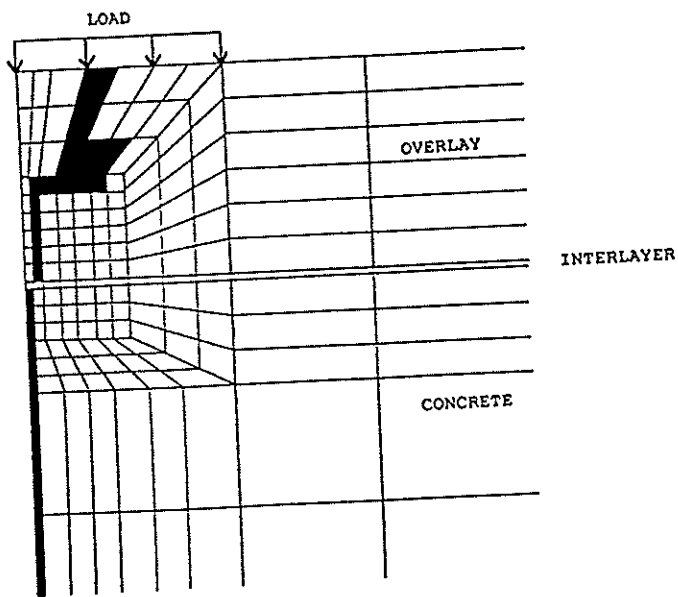


Figure 5 : Crack propagation in the overlay under combined effects (traffic and thermal)

3.6 Crack propagation

For the data presented in section 3.1. it was possible to present in figure 6 the evolution of the crack height as a function of the number of cycles. The results are expressed in days knowing that 1 thermal cycle and 1000 traffic cycles are applied per day.

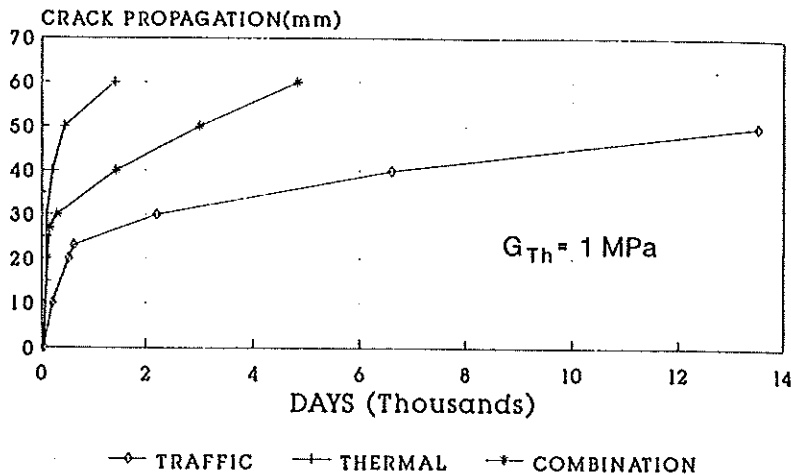


Figure 6 : Crack propagation due to traffic and/or thermal actions as a function of the days of application (1000 vehicles/1 day, 1 thermal cycle/day) (see parameters in section 3.1.)

4 Crack propagation as a function of the interface properties

It appeared to be interesting to follow the crack propagation as a function of the G value of the interlayer.

For this purpose the elastic shear modulus was adapted and the next results were obtained.

Figure 7 gives the evolution of the overlay service-life versus the thermal shear modulus of the interlayer. For this exercise, the traffic shear modulus was kept constant and equal to 2 Mpa. All the other data reported in section 3.1. remained constant.

It can be observed that the crack propagation seems to be rapid for the early age of the structure due to an important thermal contribution and then, depending on the $G_{thermal}$ value the crack propagation is more or less slowed down.

The overlay service-life is dramatically decreased when the $G_{thermal}$ value goes up to 1 Mpa.

Crack propagation

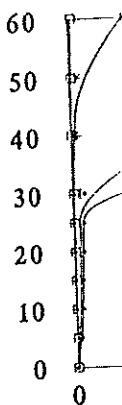


Figure 7 : Th
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5 Conclusions

This paper is to understanding of both traffic and

The presented evaluate the effects. It is The latter are Emphasis is to order to decrease real materials

Crack propagation (mm)

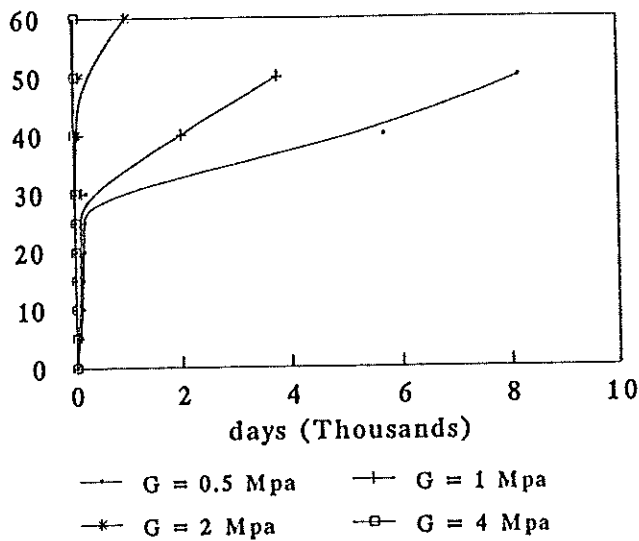


Figure 7 : The overlay crack propagation during its service-life (traffic and thermal effects) versus the thermal shear modulus of the interlayer ($G_{\text{traffic}} = 2 \text{ Mpa}$, see the data in section 3.1.)

5 Conclusions

This paper is to be considered as a contribution to the understanding of the crack propagation in overlays due to both traffic and thermal effects.

The presented method might be used in order to evaluate the crack propagation under the combined effects. It is based on a number of assumptions and data. The latter are missing dramatically in certain fields. Emphasis is to be placed on correct data collection in order to decrease the gap between calculation tools and real materials and structures behavior.

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10 DESIGN OF AS FABRIC SYSTEMS FOR REFLECTIVE CRACKING

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ABSTRACT

For an asphalt/fabric repaving reflective cracking is developed. Thermal loading is considered. Methods and the overlay thickness damage accumulation. Semi-analytical factors.

Consistent with findings from laboratory work a reduction of required fabric. Possible reduction of damage. Controlled reduction of shear crack propagation. Keywords: Design, Reflective Cracking, Mechanics, Traffic, Thermal, S

1 Introduction

Asphalt overlays are applied for the repair of work of cracked pavements. Overlay thickness is prevention of reflection cracking through the overlay before design has been applied successfully in the past. The main objective of this paper is the design of overlay thickness for the prevention of reflective cracking. Traffic - and thermal loading crack propagation problem is treated. Whereas semi-analytical solutions are used for crack propagation. Computational re

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